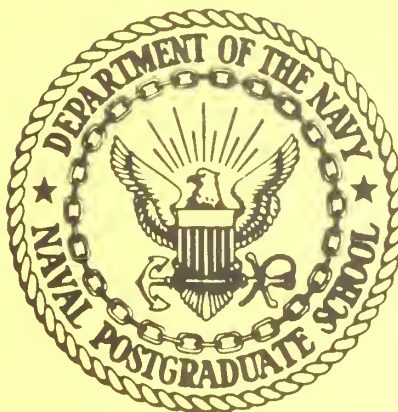


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NAVAL POSTGRADUATE SCHOOL

Monterey, California



A SHORT TABLE OF
LANCHESTER-CLIFFORD-SCHLAFLI FUNCTIONS

by
James G. Taylor
and
Gerald G. Brown

October 1977

NAVAL POSTGRADUATE SCHOOL
Monterey, California

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This report contains a reduced set of tables of Lanchester-Clifford-Schläfli (LCS) functions. A companion report contains a more extensive (and currently the most extensive available) set of tables of the LCS functions. These functions may be used to analyze Lanchester-type combat between two homogeneous forces modelled by power attrition-rate coefficients with "no effect." Theoretical background for the LCS functions is given, as well as a narrative description of the physical circumstances under which the associated		

20. Cont.

Lanchester-type combat model may be expected to be applicable. Numerical examples are given to illustrate the use of the LCS functions for analyzing "aimed-fire" combat modelled by the power attrition-rate coefficients with "no offset." Our results and these tabulations allow one to study this particular variable-coefficient combat model almost as easily and thoroughly as Lanchester's classic constant-coefficient model.

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1. Introduction

Lanchester-type* differential-equation combat models are an important tool for analyzing many important problems of military operations research. In such a combat model, a so-called attrition-rate coefficient represents the fire effectiveness of a particular weapon-system type against a particular target type, i.e. the weapon-system type's effective firepower against such a target. Time-dependent attrition-rate coefficients are used to model temporal variations in firepower on the battlefield. Thus, we see that time-dependent attrition-rate coefficients are important (and, in fact, essential [4-6]) for the quantitative analysis of hypothetical combat.

Militarily realistic computer-based Lanchester-type models of quite complex military systems have been developed for almost the entire spectrum of combat operations, from combat between battalion-sized units to theater-level operations. Nevertheless, a simple combat model may yield a clearer understanding of significant interrelationships that are difficult to perceive in a more complex model, and such insights can subsequently provide valuable guidance for more detailed computerized investigations. In this report we consider such a simplified variable-coefficient Lanchester-type model of combat between two homogeneous forces.

For this variable-coefficient Lanchester-type model of combat between two homogeneous forces, different functional forms for the attrition-rate coefficients lead to different mathematical functions being involved in representing and computing the force-level trajectories. In a previous paper [5] we have discussed the plausibility of the hypothesis that except for the special case of a constant ratio of attrition-rate coefficients,

*So-called after pioneering work of F. W. Lanchester [3].

the solutions to such differential equations cannot be represented in term of "elementary" functions of analysis. Thus, new transcendental functions arise in the study of combat modelled with time-dependent attrition-rate coefficients. In particular, we have previously introduced [5-6] so-called Lanchester-Clifford-Schläfli (LCS) functions for analyzing combat modelled with power attrition-rate coefficients with "no offset" (see Section 3 below).

In the Appendix to this report is contained a reduced set of tables for the LCS functions: it contains tables of five-decimal-place values of the hyperbolic-like LCS functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ (see Section 4 below) for 11 fractional values of α (see Section 6 below). A companion report [8] contains the most extensive set of tables currently available. The main body of this report provides the theoretical and modelling background for the use of these tables. In particular, we examine a model of a constant-speed attack on a static defensive position and show how associated range-dependent kill rates give rise to time-dependent attrition-rate coefficients with "no offset." Numerical computations are presented to illustrate the use of the LCS functions for analyzing such "aimed-fire" combat. As a consequence of the availability of these tables, one can now study this variable-coefficient combat model almost as easily and thoroughly as Lanchester's classic constant-coefficient model.

2. Variable-Coefficient Lanchester-Type Equations of Modern Warfare.

We consider combat between two homogeneous forces modelled by the following variable-coefficient Lanchester-type [3] (see [4,5]) equations of modern warfare

$$\left\{ \begin{array}{ll} \frac{dx}{dt} = -a(t)y & \text{with } x(0) = x_0 , \\ \frac{dy}{dt} = -b(t)x & \text{with } y(0) = y_0 , \end{array} \right. \quad (2.1)$$

where $t = 0$ denotes the time at which the battle begins, $x(t)$ and $y(t)$ denote the numbers of X and Y at time t , and $a(t)$ and $b(t)$ denote time-dependent Lanchester attrition-rate coefficients, which represent the effectiveness of each side's fire. These coefficients depend on variables such as force separation, tactical posture of targets, rate of target acquisition, firing rate, etc. (see [4-7] for further details). Variable attrition-rate coefficients are used to model temporal variations in firepower on the battlefield. In any analysis of combat, moreover, we should use the above equations (2.1) only for x and $y > 0$ and, for example, set $dx/dt = 0$ when $x = 0$, since negative force levels have no physical meaning.

Mathematically, we assume that the attrition-rate coefficients $a(t)$ and $b(t)$ are defined, positive, and continuous for $t_0 < t < +\infty$ with $t_0 \leq 0$. We also assume that $a(t)$ and $b(t) \in L(t_0, T)$ for any finite $T \geq t_0$. We further take $a(t)$ and $b(t)$ to be given in the form

$$a(t) = k_a g(t) , \quad \text{and} \quad b(t) = k_b h(t) , \quad (2.2)$$

where k_a and k_b are positive constants chosen so that $a(t)/b(t) = k_a/k_b$ when $g(t) \equiv h(t)$. We introduce the combat-intensity parameter λ_I and the relative-fire-effectiveness parameter λ_R defined by

$$\lambda_I = \sqrt{k_a k_b}, \quad \text{and} \quad \lambda_R = k_a/k_b. \quad (2.3)$$

From our assumptions about $a(t)$ and $b(t)$, it follows that, for example, $a(t) \notin L(t_0, T)$ implies $\int_{t_0}^T a(t)dt = +\infty$.

The X force level as a function of time may be represented as [5,6]

$$x(t) = x_0 \{C_Y(0)C_X(t) - S_Y(0)S_X(t)\} - y_0 \sqrt{\lambda_R} \{C_X(0)S_X(t) - S_X(0)C_X(t)\}, \quad (2.4)$$

where the hyperbolic-like general Lanchester functions (GLF) $C_X(t)$ and $S_X(t)$ are linearly-independent solutions to the X force-level equation

$$\frac{d^2 x}{dt^2} - \left\{ \frac{1}{a(t)} \frac{da}{dt} \right\} \frac{dx}{dt} - a(t)b(t)x = 0, \quad (2.5)$$

with initial conditions

$$C_X(t_0) = 1, \quad S_X(t_0) = 0, \quad (2.6)$$

$$\{1/a(t_0)\} dC_X/dt(t_0) = 0, \quad \{1/a(t_0)\} dS_X/dt(t_0) = 1/\sqrt{\lambda_R}.$$

Here t_0 denotes the largest finite time at which $a(t)$ or $b(t)$ ceases to be defined, positive, or continuous. The Y force level as a function of time is given by a similar expression, with $C_Y(t)$ and $S_Y(t)$ being analogously defined for the corresponding Y force-level equation.

It is sometimes convenient to introduce the new independent variable τ defined by

$$\tau = \int_{t_0}^t \sqrt{a(s)b(s)} \, ds . \quad (2.7)$$

It is readily seen that the transformation $\tau = \tau(t)$ is well defined and invertible. Let us denote $\tau(0)$ as τ_0 . We observe that $t_0 \leq 0$ implies that $\tau_0 \geq 0$. If we denote the "average intensity of combat" as $\overline{\sqrt{a(t)b(t)}}$, then

$$\overline{\sqrt{a(t)b(t)}} \, t = \left\{ (1/t) \int_0^t \sqrt{a(s)b(s)} \, ds \right\} t = \tau - \tau_0 . \quad (2.8)$$

The substitution (2.7) transforms (2.5) into

$$\frac{d^2 x}{d\tau^2} - \left(\frac{1}{2} \right) \left\{ \frac{d}{d\tau} \ln R(\tau) \right\} \frac{dx}{d\tau} - x = 0 , \quad (2.9)$$

with initial conditions

$$x(\tau_0) = x_0 , \quad \text{and} \quad \{1/\sqrt{R(\tau_0)}\} \, dx/d\tau(\tau_0) = -y_0 ,$$

where $R(\tau) = a(t)/b(t)$.

3. Combat Modelled with Power Attrition-Rate Coefficients.

The above equations (2.1) basically apply to "aimed-fire" combat when target-acquisition times do not depend on the numbers of targets available (see [5,6] for further details). A large class of tactical situations of interest can be modelled with the following general power attrition-rate coefficients [5-7]

$$a(t) = k_a (t + C)^\mu, \quad \text{and} \quad b(t) = k_b (t + C + A)^\nu, \quad (3.1)$$

where A and $C \geq 0$. We will call A the offset parameter, since it allows us to model (with μ and $\nu \geq 0$) battles between opposing weapon systems with different maximum effective ranges (see [5,6]). We will call C the starting parameter, since it allows us to model (again, with μ and $\nu \geq 0$) battles that begin within the maximum effective ranges of the two opposing systems. We observe that for the general power attrition-rate coefficients (3.1) we have $t_0 = -C$, and μ and ν must be > -1 in order that $a(t)$ and $b(t) \in L(t_0, T)$.

The above nomenclature is motivated and possible applications of our work are indicated by considering S. Bonder's model of the constant-speed attack on a static defensive position (see [4-7] for further details)

$$\frac{dx}{dt} = -\alpha(r)y, \quad \text{and} \quad \frac{dy}{dt} = -\beta(r)x, \quad (3.2)$$

where r denotes the range between opposing forces, and $\alpha(r)$ and $\beta(r)$ denote range-dependent attrition-rate coefficients. Range is related to time by

$$r(t) = R_0 - vt, \quad (3.3)$$

where R_0 denotes the opening range of battle and $v > 0$ denotes the constant attack speed. For example, let us consider the constant-speed attack of a homogeneous Y force against the static defensive position of a homogeneous X force. Figure 1 diagrammatically portrays this situation.

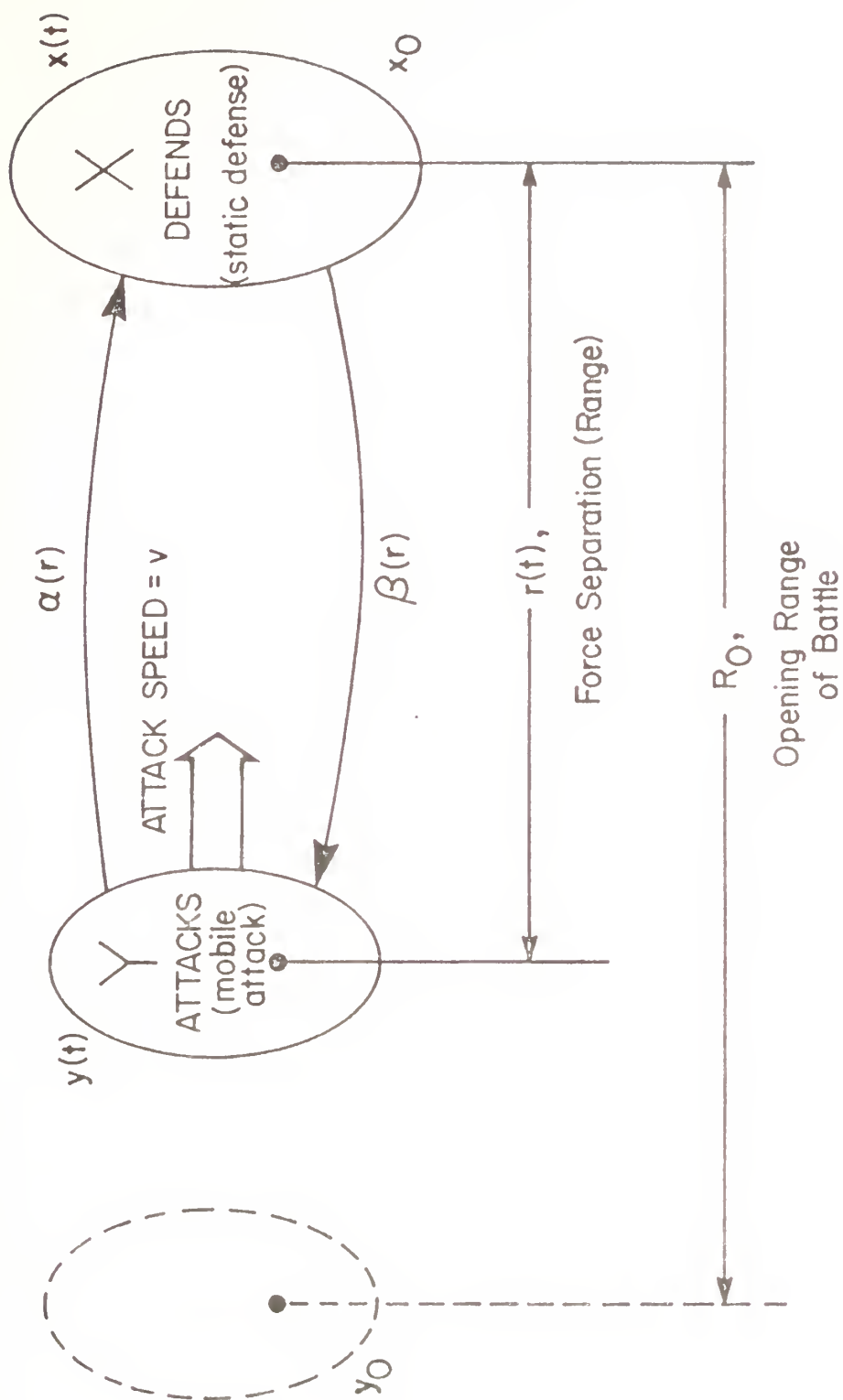


Figure 1. Diagram of Bonder's constant-speed attack model.

Force separation, $r(t)$, is given by $r(t) = R_0 - vt$.

The basic idea is that force separation, i.e. range between the opposing forces, changes over time, and the fire effectiveness of, for example, a single Y firer, denoted as $\alpha(r)$, depends on this force separation.

In many cases of tactical interest, we may model the fire effectiveness of, for example, the Y weapon system (as a function of range) with

$$\alpha(r) = \begin{cases} \alpha_0 \left(1 - \frac{r}{R_\alpha}\right)^\mu & \text{for } 0 \leq r \leq R_\alpha, \\ 0 & \text{for } R_\alpha \leq r, \end{cases} \quad (3.4)$$

where R_α denotes the maximum effective range of the Y weapon system and $\mu \geq 0$. Here μ is used to model the range dependency of Y's attrition-rate coefficient (see Figure 2). We model $\beta(r)$ similarly, with corresponding quantities R_β and ν being analogous to R_α and μ above.

If we use (3.3) to eliminate range r from (3.4), we obtain

$$\begin{cases} \frac{dx}{dt} = -a(t)y, \\ \frac{dy}{dt} = -b(t)x, \end{cases} \quad (3.5)$$

where the time-dependent attrition-rate coefficients $a(t)$ and $b(t)$ are given by (3.1). It follows that the offset and starting parameters are given by

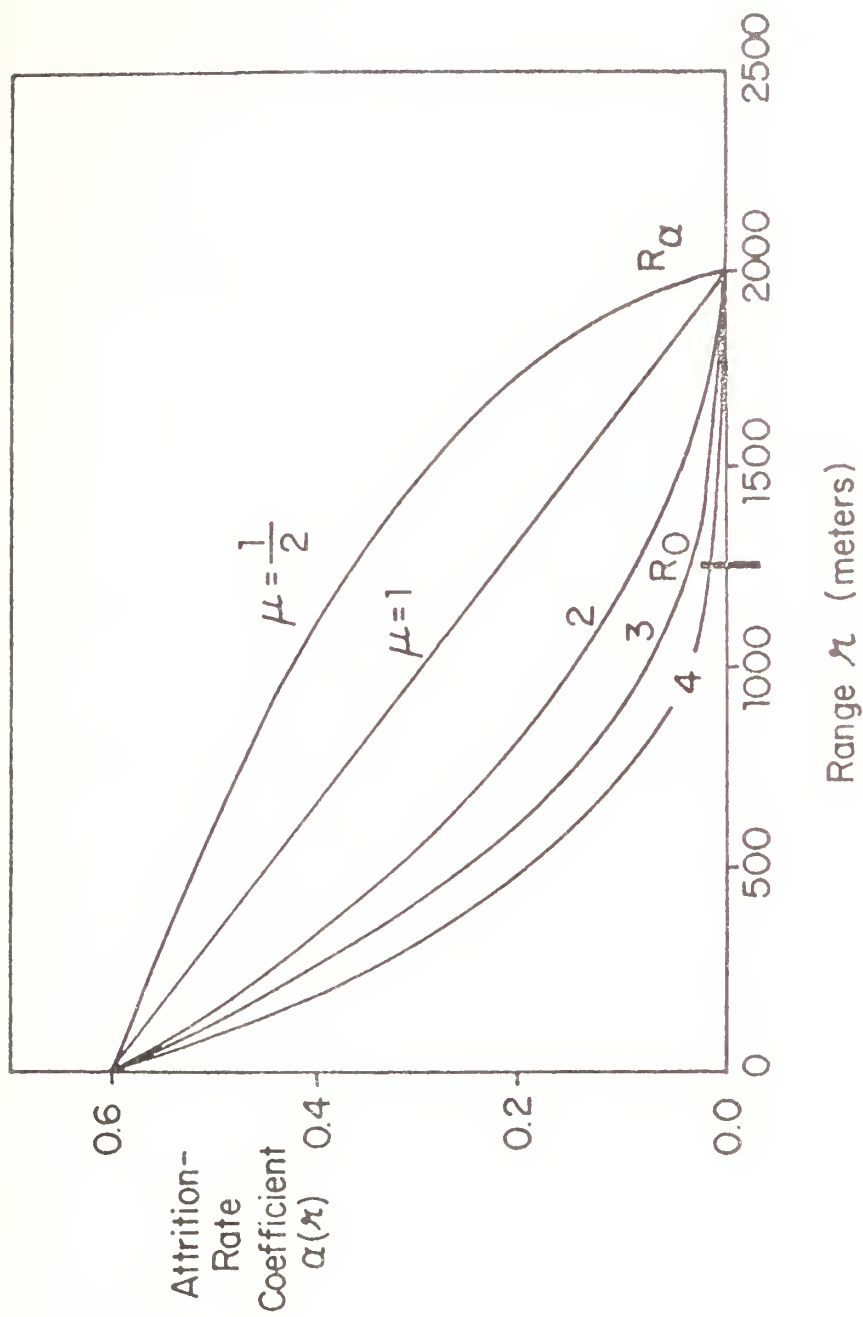


Figure 2. Dependence of Y's attrition-rate coefficient $\alpha(r)$ on the exponent μ with the maximum effective range of the weapon system and kill rate at zero range held constant. [NOTES: 1. The maximum effective range of the system is denoted as $R_\alpha = 2000$ meters. 2. $\alpha(0) = \alpha_0 = 0.6X$ casualties/(unit time \times number of Y firers) denotes the weapon-system kill rate for Y at zero force separation (range). 3. The opening range of battle is denoted as $R_0 = 1250$ meters and (as shown) $R_0 < R_\alpha$.]

$$A = \left(\frac{R_\beta - R_\alpha}{v} \right), \quad \text{and} \quad C = \left(\frac{R_\alpha - R_0}{v} \right), \quad (3.6)$$

and that

$$k_a = \alpha_0 \left(\frac{v}{R_\alpha} \right)^\mu, \quad \text{and} \quad k_b = \beta_0 \left(\frac{v}{R_\beta} \right)^\nu. \quad (3.7)$$

We observe that A and $C \geq 0$ if and only if $R_\beta \geq R_\alpha \geq R_0$. By considering (3.6) and Figure 3, the reader should have no trouble in understanding our terminology for A and C .

When the offset parameter is equal to zero (i.e. $A = 0$), then the coefficients (3.1) reduce to

$$a(t) = k_a (t+C)^\mu, \quad \text{and} \quad b(t) = k_b (t+C)^\nu. \quad (3.8)$$

We will refer to (3.8) as power attrition-rate coefficients with "no offset."

As we have seen above in Bonder's constant-speed attack model, these coefficients model, for example, combat between weapon systems with the same maximum effective range so that there is no "offset" in the "reaching out" of the weapon systems against each other in combat (again, see Figure 3). For these coefficients (3.8), the transformed X force-level equation (2.9) becomes

$$\frac{d^2 x}{d\tau^2} + \left(\frac{2q-1}{\tau} \right) \frac{dx}{d\tau} - x = 0, \quad (3.9)$$

with initial conditions

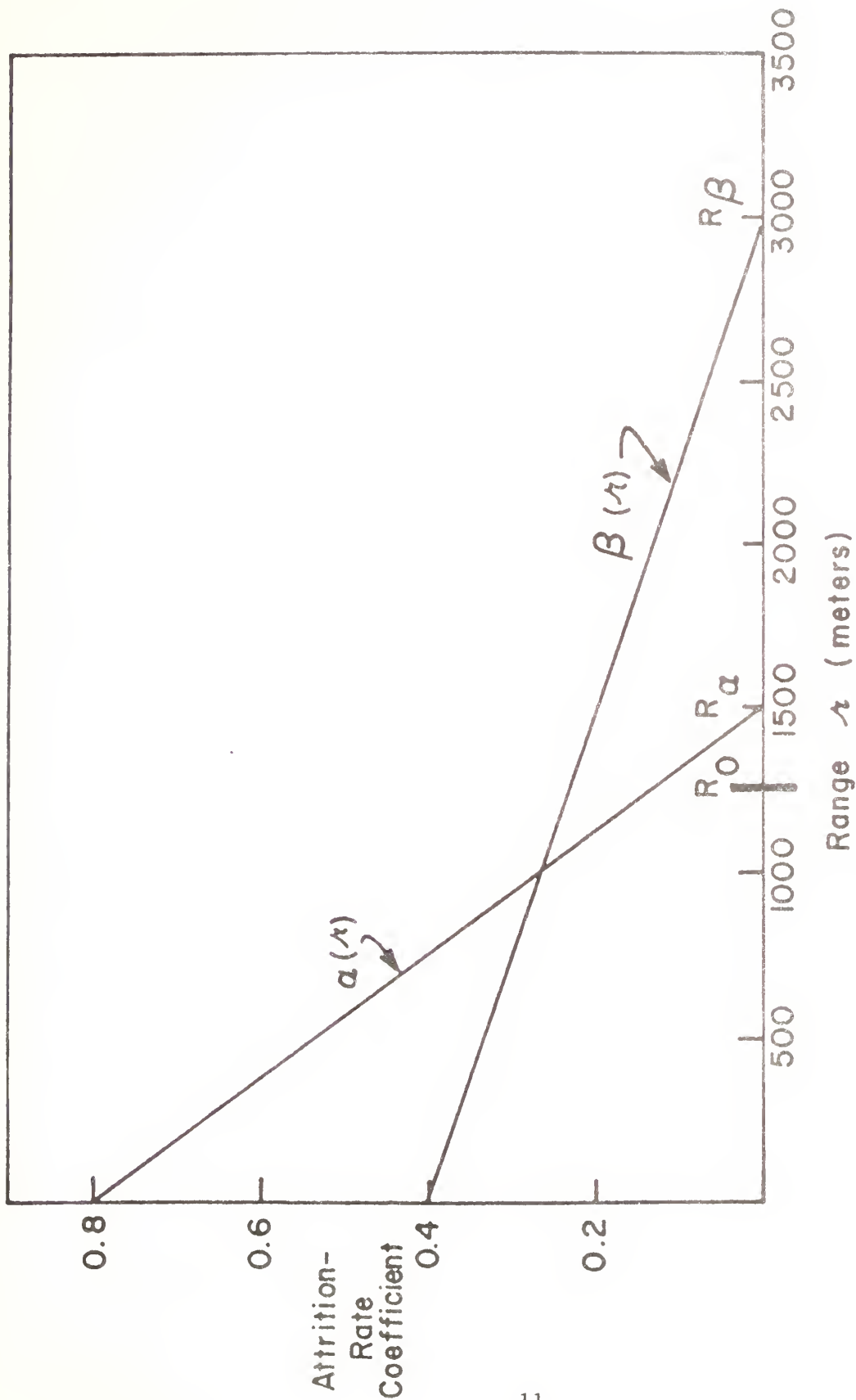


Figure 3. Explanation of the offset parameter A and the starting parameter C for power attrition-rate coefficients modelling constant-speed attack. [NOTES: 1. The maximum effective ranges of the X and Y weapon systems are denoted as R_α and R_β , respectively. 2. The opening range of battle is denoted as R_0 and (as shown) $R_0 < \min(R_\alpha, R_\beta)$. 3. The offset parameter is given by $A = (R_\beta - R_\alpha)/v$. 4. The starting parameter is given by $C = (R_\alpha - R_0)/v$.]

$$x(\tau_0) = x_0, \quad \text{and} \quad \left\{ \left(\frac{\tau}{2} \right)^{2q-1} \frac{dx}{d\tau} \right\}_{\tau=\tau_0} = -y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{2q-1}.$$

Here

$$q = \left(\frac{\nu + 1}{\mu + \nu + 2} \right), \quad (3.10)$$

$$\tau = \left(\frac{2\lambda_I}{\mu + \nu + 2} \right) (t + C)^{(\mu + \nu + 2)/2}, \quad (3.11)$$

and

$$\tau_0 = \left(\frac{2\lambda_I}{\mu + \nu + 2} \right) C^{(\mu + \nu + 2)/2}. \quad (3.12)$$

Let us observe that $0 < q < 1$ when μ and $\nu > -1$. Furthermore, $q > 1/2$ if and only if $dR/dt < 0$, i.e. $R(t)$ is a strictly decreasing function of time.

4. Lanchester-Clifford-Schläfli (LCS) Functions.

Consider the function $F_\alpha(x)$ defined by the power series

$$F_\alpha(x) = \Gamma(\alpha) \sum_{k=0}^{\infty} \frac{(x/2)^{2k}}{\{k! \Gamma(k + \alpha)\}}. \quad (4.1)$$

For $\alpha \neq 0, -1, -2, \dots$ the radius of convergence for $F_\alpha(x)$ is infinite by the ratio test for convergence of power series [2]. Hence, $F_\alpha(z)$ is an entire function of the complex variable $z = x + iy$, with an essential

singularity at the point at infinity. Now consider the function $H_\alpha(x)$ defined by the infinite series

$$H_\alpha(x) = \Gamma(\alpha) \sum_{k=0}^{\infty} \frac{(x/2)^{2(k+\alpha)}}{\{k! \Gamma(k+\alpha+1)\}} . \quad (4.2)$$

Observing that

$$H_\alpha(x) = (1/\alpha)(x/2)^{2\alpha} F_{\alpha+1}(x) , \quad (4.3)$$

we see that for $\alpha > 0$ the infinite series (4.2) is uniformly convergent on compact subsets of the complex plane. From (4.3) one can readily deduce the recursive relation

$$F_\alpha(x) = F_{\alpha+1}(x) + \left\{ \frac{(x/2)^2}{\alpha(\alpha+1)} \right\} F_{\alpha+2}(x) . \quad (4.4)$$

We will call the functions $F_\alpha(x)$ and $H_\alpha(x)$ Lanchester-Clifford-Schl fli (LCS) functions (see Note 10 on pp. 66-67 of [5]). Other properties are readily deduced and are given in Table I.

The function $F_\alpha(x)$ satisfies the linear second-order ordinary differential equation

$$\frac{d^2 F_\alpha}{dx^2} + \left(\frac{2\alpha-1}{x} \right) \frac{dF_\alpha}{dx} - F_\alpha = 0 , \quad (4.5)$$

with initial conditions

Table I. Properties of the LCS Functions $F_{\alpha}(x)$ and $H_{\alpha}(x)$.

1. $dF_{\alpha}/dx = (x/2)^{1-2\alpha}H_{\alpha}(x)$
2. $dH_{\alpha}/dx = (x/2)^{2\alpha-1}F_{\alpha}(x)$
3. $F_{\alpha}(x)F_{1-\alpha}(x) - H_{\alpha}(x)H_{1-\alpha}(x) = 1 \quad \forall x$
where α is not an integer (including zero)
4. $F_{\alpha}(x=0) = 1$
5. $H_{\alpha}(x=0) = 0 \quad \text{for } \alpha > 0$
6. $dF_{\alpha}/dx(x=0) = 0$
7. $\{(x/2)^{1-2\alpha}dH_{\alpha}/dx\}_{x=0} = 1$
8. $F_{1/2}(x) = \cosh x$
9. $H_{1/2}(x) = \sinh x$

$$F_{\alpha}(0) = 1, \quad \text{and} \quad \frac{dF_{\alpha}}{dx}(0) = 0,$$

while $H_{\alpha}(x)$ satisfies

$$\frac{d^2 H_{\alpha}}{dx^2} - \left(\frac{2\alpha-1}{x}\right) \frac{dH_{\alpha}}{dx} - H_{\alpha} = 0, \quad (4.6)$$

with initial conditions

$$H_{\alpha}(0) = 0, \quad \text{and} \quad \left\{ \left(\frac{x}{2}\right)^{1-2\alpha} \frac{dH_{\alpha}}{dx} \right\}_{x=0} = 1.$$

Thus, $\{F_{\alpha}, H_{1-\alpha}\}$ is a fundamental system of solutions to

$$\frac{d^2 F}{dx^2} + \left(\frac{2\alpha-1}{x}\right) \frac{dF}{dx} - F = 0, \quad (4.7)$$

with Wronskian $W(F_{\alpha}, H_{1-\alpha}) = (x/2)^{1-2\alpha}$. It follows that the GLF for the X and Y force-level equations for combat modelled with the attrition-rate coefficients (3.8) are given by

$$C_X(t) = F_q(\tau(t)), \quad S_X(t) = \left(\frac{\lambda_I}{\mu + \nu + 2}\right)^{2q-1} H_p(\tau(t)), \quad (4.8)$$

$$C_Y(t) = F_p(\tau(t)), \quad S_Y(t) = \left(\frac{\lambda_I}{\mu + \nu + 2}\right)^{1-2q} H_q(\tau(t)), \quad (4.9)$$

where $p = 1-q$. If we define

$$T_{\alpha}(x) = H_{1-\alpha}(x)/F_{\alpha}(x) , \quad (4.10)$$

then

$$T_X(t) = \frac{S_X(t)}{C_X(t)} = \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{2q-1} \frac{H_p(\tau(t))}{F_q(\tau(t))} , \quad (4.11)$$

or

$$T_X(t) = \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{2q-1} T_q(\tau(t)) , \quad (4.12)$$

where $T_X(t)$ denotes a hyperbolic-like GLF, which corresponds to the hyperbolic tangent. Observing that for $\mu, \nu > -1$, $\lim_{t \rightarrow +\infty} \tau(t) = +\infty$, we see that $T_{\alpha}(x)$ is a strictly increasing function of x on the interval $[0, +\infty)$ and

$$0 \leq T_{\alpha}(x) < \frac{\Gamma(1-\alpha)}{\Gamma(\alpha)} \quad \text{for } 0 \leq x < +\infty , \quad (4.13)$$

with

$$\lim_{x \rightarrow +\infty} T_{\alpha}(x) = \frac{\Gamma(1-\alpha)}{\Gamma(\alpha)} , \quad (4.14)$$

since by the results of Taylor and Comstock [7] the parity-condition parameter $Q^* = Q^*(\mu, \nu, C = 0)$ is given by

$$\lim_{t \rightarrow +\infty} T_X(t) = \frac{1}{Q^*(\mu, \nu, 0)} = \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{2q-1} \frac{\Gamma(p)}{\Gamma(q)} . \quad (4.15)$$

We recall that Taylor and Comstock [7] have introduced the so-called parity-condition parameter Q^* as the value (or range of such values) for the initial condition Q to the initial-value problem

$$\left\{ \begin{array}{ll} \frac{dE_X^-}{dt} = -\frac{1}{\sqrt{\lambda_R}} a(t) E_Y^- & \text{with } E_X^-(t_0) = 1, \\ \frac{dE_Y^-}{dt} = -\sqrt{\lambda_R} b(t) E_X^- & \text{with } E_Y^-(t_0) = Q, \end{array} \right. \quad (4.16)$$

such that $E_X^-(t; Q^*)$ and $E_Y^-(t; Q^*) > 0$ for all $t \geq t_0$. In other words, Q^* is the value of Q in (4.16) above such that neither E_X^- nor E_Y^- ever become zero. In this case, both $E_X^-(t; Q^*)$ and $E_Y^-(t; Q^*)$ are positive, strictly decreasing functions, similar to decreasing exponentials. Thus, we may call Q^* "the Y equivalent of an X force of unit strength," since the forces are "at parity," with neither force being annihilated in finite time. Taylor and Comstock have shown that for either $a(t) \notin L(0, +\infty)$ or $b(t) \notin L(0, +\infty)$, then Q^* is unique and given by

$$\lim_{t \rightarrow +\infty} \frac{S_X(t)}{C_X(t)} = \frac{1}{Q^*}. \quad (4.17)$$

The significance of the parity-condition parameter Q^* is that it allows us to predict force annihilation as the following theorem shows.

THEOREM 1 (Taylor and Comstock [7]): Assume that either $a(t) \notin L(0, +\infty)$ or $b(t) \notin L(0, +\infty)$. Then the X force will be annihilated in finite time if and only if

$$\frac{x_0}{y_0} < \sqrt{\lambda_R} \left\{ \frac{C_X(0) - Q^* S_X(0)}{Q^* C_Y(0) - S_Y(0)} \right\}. \quad (4.18)$$

5. Use of LCS Functions for Analyzing Combat.

The Lanchester-Clifford-Schläfli (LCS) functions $F_{\alpha}(x)$ and $H_{\alpha}(x)$ are useful for analyzing "aimed-fire" combat (see Section 3 above) modelled with the power attrition-rate coefficients with "no offset" (3.8), which we rewrite here as

$$a(t) = k_a (t + C)^{\mu}, \quad \text{and} \quad b(t) = k_b (t + C)^{\nu}. \quad (5.1)$$

In other words, the LCS functions arise in solving the differential combat model (2.1) with attrition-rate coefficients (5.1). In order that both $a(t)$ and $b(t) \in L(t_0, T)$, we must have μ and $\nu > -1$. Military situations modelled by these equations have been discussed in Section 3 above, e.g. combat between two weapon systems with the same maximum effective range. For such combat, the LCS functions may be used to

- (1) compute force-level declines,
 - (2) predict force annihilation,
- and
- (3) predict the time of force annihilation.

Let us now see how the LCS functions may be used to obtain the above information about force-level declines and force-annihilation prediction. According to (2.4), (4.8), and (4.9) above, the X force level is given by

$$x(t) = x_0 \{F_p(\tau_0) F_q(\tau(t)) - H_q(\tau_0) H_p(\tau(t))\} \\ - y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{2q-1} \{F_q(\tau_0) H_p(\tau(t)) - H_p(\tau_0) F_q(\tau(t))\}, \quad (5.2)$$

where q is given by (3.10), $p = 1-q$, and $\tau(t)$ is given by (3.11), which we rewrite as

$$\tau(t) = \left(\frac{2\lambda_I}{\mu + \nu + 2} \right) (t + C)^{(\mu+\nu+2)/2}, \quad (5.3)$$

The time to annihilate the X force* is determined by $x(t_a^X) = 0$, and it follows that

$$T_q(\tau(t_a^X)) = \frac{x_0 F_p(\tau_0) + y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} H_p(\tau_0)}{x_0 H_q(\tau_0) + y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} F_q(\tau_0)}, \quad (5.4)$$

where from (4.10)

$$T_q(\tau(t)) = H_p(\tau(t)) / F_q(\tau(t)), \quad (5.5)$$

and we recall that $p + q = 1$. It follows that the time to annihilate X , t_a^X , is given by

* If we multiply the first equation of (2.1) by y , the second by x , add, and integrate, we obtain

$$x(t) y(t) = x_0 y_0 - \int_0^t \{a(s) y^2(s) + b(s) x^2(s)\} ds,$$

which shows that $x(t)$ and $y(t)$ can have at most one finite zero. Hence, if $x(t_a^X) = 0$, then we know that $y(t) > 0$ for all $t \geq 0$.

$$t_a^X = \tau^{-1} \left\{ T_q^{-1} \left[\frac{x_0 F_p(\tau_0) + y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} H_p(\tau_0)}{x_0 H_q(\tau_0) + y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} F_q(\tau_0)} \right] \right\}. \quad (5.6)$$

Taylor and Comstock [7] have shown that $T_q(\tau)$ is strictly increasing and satisfies (see also (4.12) above)

$$0 \leq T_q(\tau) < \Gamma(p)/\Gamma(q), \quad (5.7)$$

where $p = 1-q$. It follows that in order for X to be annihilated in finite time, the right-hand side of (5.4) must be less than $\Gamma(p)/\Gamma(q)$. Let us observe that for $t_0 = -C = 0$, (5.4) simplifies to

$$T_q(\tau(t_a^X)) = \frac{x_0}{y_0 \sqrt{\lambda_R}} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{p-q}. \quad (5.8)$$

Thus, we have proved the following theorem concerning force-annihilation prediction.

THEOREM 2: Consider combat between two homogeneous forces modelled by (2.1) with power attrition-rate coefficients (5.1). Assume that μ and $\nu > -1$ and that the above equations hold for all time. Then the X force will be annihilated in finite time if and only if

$$\begin{aligned}
& \Gamma(q) \left\{ x_0 F_p(\tau_0) + y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} H_p(\tau_0) \right\} \\
& < \Gamma(p) \left\{ x_0 H_q(\tau_0) + y_0 \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} F_q(\tau_0) \right\}, \quad (5.9)
\end{aligned}$$

where $q = (\nu + 1)/(\mu + \nu + 2)$ and $p = 1 - q$. For $t_0 = 0$ (i.e. $C = 0$ so that $\tau_0 = 0$), X will be annihilated in finite time if and only if

$$\frac{x_0}{y_0} < \frac{\Gamma(p)}{\Gamma(q)} \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p}. \quad (5.10)$$

6. Tabulation of LCS Functions.

This report contains a reduced set of tables of the Lanchester-Clifford-Schläfli functions. The Appendix contains tables of five-decimal-place values of the hyperbolic-like LCS functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and $T_\alpha(x)$ for various values of the argument x , namely $x = 0.00$ (0.01) 2.00 (0.1) 10.0, and $\alpha = 1/2, 1/3, 2/3, 1/4, 3/4, 1/5, 2/5, 3/5, 4/5, 3/7$, and $4/7$. These values of the index α correspond to $u, \nu = 0, 1, 2$, and 3 in (3.8) and allow one to analyze, for example, a basic spectrum of range capabilities for weapon systems in the constant-speed-attack model of Section 3. These tables have been calculated by the recursive means given in Section 8 of [5]. A more extensive tabulation (namely, for $\alpha = 1/2, 1/3, 2/3, 1/4, 3/4, 1/5, 2/5, 3/5, 4/5, 2/7, 3/7, 4/7, 5/7, 4/9, 5/9, 3/11, 5/11, 6/11, 8/11, 5/13, 8/13, 5/17, 12/17, 5/21$, and $16/21$ corresponding to $\mu, \nu = 0, 1/4, 1/2, 1, 1\frac{1}{2}, 2, 3$)

is to be found in a companion report [8]. This companion report contains the most extensive set of tables of the Lanchester-Clifford-Schläfli functions currently available.

A representative tabulation of the hyperbolic-like LCS functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and $T_\alpha(x)$ is given in, for example, Tables 8A and 8B of the Appendix for $\alpha = 3/5$. The values of the argument x are the same as those used for the tabulation of the hyperbolic functions by Abramowitz and Stegun [1]. We observe from Table 8B and (4.13) that the limiting value of $T_\alpha(x)$ as $x \rightarrow +\infty$ (here $\alpha = 3/5$) is quickly reached, with three-decimal-place accuracy already attained for $x = 4.5$. Moreover, the use of these tables (specifically, Tables 8A and 8B of the Appendix) for combat analysis is illustrated in the next section.

7. Numerical Examples

In this section we examine a couple of numerical examples to show some of the insights that may be gained into the dynamics of combat between two homogeneous forces from our results (see also [6]). These examples illustrate the use of the LCS functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and $T_\alpha(x)$ for analyzing "aimed-fire" combat modelled with the power attrition-rate coefficients with "no offset" (5.1). As in [4-7], we consider S. Bonder's model (3.2) for the constant-speed attack against a static defensive position. We will focus on the use of the LCS functions for predicting force annihilation, since the computing of force-level trajectories with Lanchester functions is adequately handled elsewhere (see [4-5]),

Let us accordingly consider the constant-speed attack of a homogeneous Y force against the static defensive position of a homogeneous X force (see Section 3 above for further modelling details, especially Figure 1). For our numerical computations, we assume that the fire effectiveness of the Y weapon system varies linearly with range, i.e.

$$\alpha(r) = \begin{cases} \alpha_0 \left(1 - \frac{r}{R_\alpha}\right) & \text{for } 0 \leq r \leq R_\alpha, \\ 0 & \text{for } R_\alpha \leq r, \end{cases} \quad (7.1)$$

and that the fire effectiveness of the X weapon system varies quadratically with range, i.e.

$$\beta(r) = \begin{cases} \beta_0 \left(1 - \frac{r}{R_\beta}\right)^2 & \text{for } 0 \leq r \leq R_\beta, \\ 0 & \text{for } R_\beta \leq r, \end{cases} \quad (7.2)$$

with $R_\alpha = R_\beta$, i.e. both weapon systems have the same maximum effective range. In other words, $\mu = 1$ in (3.4) and $\nu = 2$ for $\beta(r)$. We consider a battle modelled by the input data given in Table II. In terms of time as the independent variable, the attrition-rate coefficients (7.1) and (7.2) become via (3.3)

$$a(t) = k_a(t + C) \quad \text{and} \quad b(t) = k_b(t + C)^2, \quad (7.3)$$

Table II. Input Data for Numerical Examples

$$\mu = 1, \quad \nu = 2$$

$$\alpha_0 = 0.06 \text{ X casualties/minute/Y firer}$$

$$\beta_0 = 0.6 \text{ Y casualties/minute/X firer}$$

$$R_\alpha = R_\beta = 2000 \text{ meters}$$

$$v = 5 \text{ miles/hour}$$

where $R_\alpha = R_\beta$,

$$C = \frac{R_\alpha - R_0}{v}, \quad k_a = \frac{\alpha_0 v}{R_\alpha}, \quad \text{and} \quad k_b = \beta_0 \left(\frac{v}{R_\beta} \right)^2. \quad (7.4)$$

From the input data given in Table II, we compute the parameter values shown in Table III, since the transformed X force-level equation is given by (3.9) with $q = (v + 1)/(\mu + v + 2)$, $p = 1 - q$, $\mu = 1$, and $v = 2$. Thus, the X force level may be computed with $F_\alpha(\tau)$ and $H_{1-\alpha}(\tau)$ with $\alpha = q = 3/5$. Force-annihilation prediction involves the limiting value of $T_\alpha(\tau) = H_{1-\alpha}(\tau)/F_\alpha(\tau)$ as $\tau \rightarrow +\infty$. From Table 8B of the Appendix and Table III, we note the predicted agreement between $\Gamma(1-\alpha)/\Gamma(\alpha)$ and the limiting value of $T_\alpha(x)$ as $x \rightarrow +\infty$ [recall (4.13)] for $\alpha = q = 3/5$. We now consider two cases: (I) $R_0 = 2000$ meters, and (II) $R_0 = 1250$ meters.

When $R_0 = 2000$ meters (see Figure 3 of [4]), we have $C = 0$ and $\tau_0 = 0$. The maximum time that the battle can last is $t_{\max} = R_0/v = 14.91$ minutes, since at this time the attackers reach their final objective, i.e. the defender's position (again, see Figure 1). We now consider the qualitative behavior of the $\mu = 1$, $v = 2$ force-level trajectory shown in Figure 3 of [4]. Theorem 2 tells us that the X force can be annihilated in finite time if and only if

$$\frac{x_0}{y_0} < \frac{\Gamma(p)}{\Gamma(q)} \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + v + 2} \right)^{q-p}, \quad (7.3)$$

where $q = 3/5$ and $p = 1 - q$. Using the numerical values in Table III, we compute from (7.3) that the X force can be annihilated in finite time if and only if

Table III. Parameter Values for Numerical Examples

$$k_a = 4.0233 \times 10^{-3} \text{ X casualties/minute}^\mu/\text{Y firer}$$

$$k_b = 2.6979 \times 10^{-3} \text{ Y casualties/minute}^\nu/\text{X firer}$$

$$p = 2/5, \quad q = 3/5$$

$$\Gamma(p)/\Gamma(q) = 1.48951$$

$$A = 0$$

$$\frac{x_0}{y_0} < 0.420 . \quad (7.4)$$

When the X force can be annihilated, its annihilation time is given by (5.8), which we rewrite here as

$$T_q(\tau(t_a^X)) = \frac{x_0}{y_0 \sqrt{\lambda_R}} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{p-q} , \quad (7.5)$$

where

$$\tau(t) = \left(\frac{2\lambda_I}{\mu + \nu + 2} \right) t^{(\mu+\nu+2)/2} . \quad (7.6)$$

Thus, for the numerical values given in Table III, the time of annihilation of the X force is given by

$$T_q(\tau(t_a^X)) = 3.544 \frac{x_0}{y_0} , \quad (7.7)$$

with $q = 3/5$. We will now illustrate further computations for $x_0 = 10$ and $y_0 = 30$. From (7.4) we see that the X force can be annihilated in finite time (but we must verify that $t_a^X \leq t_{\max}$). In this case (7.7) becomes

$$T_q(\tau(t_a^X)) = 1.18122 . \quad (7.8)$$

We must now determine $\tau(t_a^X)$ such that $\tau(t_a^X) = T_q^{-1}(1.18122)$ by using interpolation methods and Tables 8A and 8B. From Table 8A, we find

$$\begin{aligned} T_q(\tau) &= 1.18172 & \text{for } &= 1.01 \\ T_q(\tau) &= 1.17630 & \text{for } &= 1.00 \end{aligned}$$

so that using linear interpolation, we obtain

$$\tau(t_a^X) = 1.009, \quad (7.9)$$

whence use of (7.6) yields

$$t_a^X = 14.24 \text{ minutes}, \quad (7.10)$$

which is less than $t_{\max} = 14.91$ minutes so that the defending X force is indeed annihilated before the attacking Y force reaches its final objective. Since $r(t) = R_0 - vt$, we find that force separation at the instant of annihilation of the X force is

$$r_a^X = 89.8 \text{ meters}. \quad (7.11)$$

Further results may be computed in a similar fashion and are given in Table IV.

When $R_0 = 1250$ meters (see Figure 3 of [5]), we have $C = 5.5923$ minutes, $\tau_0 = 0.0975$, and $t_{\max} = R_0/v = 9.32$ minutes. In this case Theorem 2 tells us that the X force can be annihilated in finite time if and only if

Table IV. Annihilation of the X Force as a Function
of the Initial Force Ratio for $R_0 = 2000$ meters

$\frac{(x_0/y_0)}{}$	$\frac{t_a^X}{}$ (minutes)	$\frac{r_a^X}{}$ (meters)
0.333	14.24	89.8
0.250	11.61	443.2
0.200	10.19	633.2

$$\frac{x_0}{y_0} < \sqrt{\lambda_R} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{q-p} \frac{\Gamma(p)}{\Gamma(q)} \frac{\left\{ F_q(\tau_0) - \frac{\Gamma(q)}{\Gamma(p)} H_p(\tau_0) \right\}}{\left\{ F_p(\tau_0) - \frac{\Gamma(p)}{\Gamma(q)} H_q(\tau_0) \right\}}, \quad (7.12)$$

with $q = 3/5$ and $p = 1-q$. Using linear interpolation, we obtain from Tables 7A and 8A of the Appendix that for the numerical values of Table III

$$F_p(\tau_0) = 1.006, \quad H_q(\tau_0) = 0.044, \quad (7.13)$$

$$F_q(\tau_0) = 1.004, \quad H_p(\tau_0) = 0.223,$$

so that (7.12) says that the X force can be annihilated if and only if

$$\frac{x_0}{y_0} < 0.382. \quad (7.14)$$

When the X force can be annihilated, its annihilation time is given by (5.4), which we rewrite here as

$$T_q(\tau(t_a^X)) = \frac{\left\{ \frac{x_0}{y_0 \sqrt{\lambda_R}} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{p-q} F_p(\tau_0) + H_p(\tau_0) \right\}}{\left\{ F_q(\tau_0) + \frac{x_0}{y_0 \sqrt{\lambda_R}} \left(\frac{\lambda_I}{\mu + \nu + 2} \right)^{p-q} H_q(\tau_0) \right\}}, \quad (7.15)$$

whence for the data of Table III

$$T_a(\tau(t_a^X)) = \frac{3.565u_0 + 0.223}{0.156u_0 + 1.004}, \quad (7.16)$$

where $u_0 = x_0/y_0$. Let us also record here that (3.11) yields

$$t = \left(\frac{\{\mu + \nu + 2\}\tau}{2\lambda_I} \right)^{2/(\mu+\nu+2)} - C . \quad (7.17)$$

We will again illustrate further computations for $x_0 = 10$ and $y_0 = 30$.

From (7.14) we see that the X force can be annihilated in finite time (but again we must investigate whether or not $t_a^X \leq t_{\max}$). In this case (7.16) becomes

$$T_q(\tau(t_a^X)) = 1.33651 , \quad (7.18)$$

whence Table 8A of the Appendix and linear interpolation yield

$$\tau(t_a^X) = 1.397 , \quad (7.19)$$

so that by (7.17)

$$t_a^X = 10.63 \text{ minutes} . \quad (7.20)$$

Since $t_{\max} = R_0/v = 9.32$ minutes $< t_a^X$, we see that the attacking Y force overruns the defender's position before annihilation of the X force occurs. Thus, the battle ends with $x_f = x(t_f) > 0$ and $y_f > 0$ at $t_f = t_{\max} = 9.32$ minutes. Corresponding to $t_f = 9.32$ minutes is $\tau_f = 1.1318$, and then Table 8A of the Appendix yields

$$F_q(\tau_f = 1.1318) = 1.589 , \quad H_p(1.1318) = 1.973 , \quad (7.21)$$

whence via (2.4), (4.8), (4.9), and (7.13) we obtain

$$x_f = x(t_f) = x(r = 0) = 1.35 . \quad (7.22)$$

Some further numerical results are given in Table V. Again, these parametric results should be contrasted with the single $\mu = 1$, $\nu = 2$ force-level trajectory shown in Figure 3 of [5].

8. Final Remarks

In the previous section above, we have seen how the LCS functions allow one to conveniently obtain much valuable information about the model (2.1) with power attrition-rate coefficients (3.8) without having to explicitly compute the entire force-level trajectories. Previously we were limited to computing only force-level trajectories (see [4-5]). With the availability of these tabulations of LCS functions (see the Appendix of this report and [8]), we can now tell who is going to be annihilated and when this event will happen without having to compute the trajectories. Not only did we answer questions about the qualitative behavior of the model (e.g. force annihilation) for specific values of, for example, initial force levels but also for a range of values of the initial force ratio (i.e. parametric analysis of model behavior).

Table V. Annihilation of the X Force as a Function
of the Initial Force Ratio for $R_0 = 1250$ meters

(x_0/y_0)	t_a^X (minutes)	r_a^X (meters)
0.333	10.63	_____†
0.250	7.56	235.9
0.200	6.17	422.8

† $t_{\max} = 9.32$ minutes and $x_f = x(r=0) = 1.35$.

The results of this report may be used for other parametric analyses, e.g. parametric dependence of battle outcome on attrition-rate coefficients. Thus, the contents of this report allow one to develop important insights into the dynamics of combat between two homogeneous forces with temporal variations in fire effectiveness. With the availability of tabulations of the LCS functions, one can now analyze such combat modelled by the power attrition-rate coefficients (3.8) with somewhat the same facility as he can for the constant-coefficient case and thus aid in parametric analyses. For further discussions of the significance of such results for military operations research, the reader is directed to [6] and [7].

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APPENDIX: Tabulation of the LCS Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for
 $\alpha = 1/2, 1/3, 2/3, 1/4, 3/4, 1/5, 2/5, 3/5, 4/5, 3/7$, and $4/7$.

x	$F_{1/2}(x)$	$H_{1/2}(x)$	$T_{1/2}(x)$	x	$F_{1/2}(x)$	$H_{1/2}(x)$	$T_{1/2}(x)$	x	$F_{1/2}(x)$	$H_{1/2}(x)$	$T_{1/2}(x)$
0.0	1.00000	0.0	0.0	0.50	1.12763	0.52110	0.46212	1.00	1.53308	1.17520	0.76159
0.01	1.00005	0.01000	0.01000	0.51	1.13289	0.53470	0.46995	1.01	1.54491	1.19099	0.76576
0.02	1.00010	0.02000	0.02000	0.52	1.13827	0.54875	0.47770	1.02	1.55674	1.20699	0.76991
0.03	1.00015	0.03000	0.03000	0.53	1.14377	0.56316	0.48538	1.03	1.56857	1.22323	0.77406
0.04	1.00020	0.04000	0.03999	0.54	1.14938	0.57793	0.49299	1.04	1.58040	1.23973	0.77819
0.05	1.00125	0.05002	0.04996	0.55	1.15510	0.59295	0.50052	1.05	1.60379	1.25646	0.78181
0.06	1.00245	0.06004	0.05993	0.56	1.16097	0.60827	0.50807	1.06	1.62719	1.27341	0.78556
0.07	1.00370	0.07009	0.06993	0.57	1.16729	0.62390	0.51557	1.07	1.65061	1.29059	0.78936
0.08	1.00500	0.08020	0.07993	0.58	1.17397	0.63983	0.52302	1.08	1.67404	1.30801	0.79320
0.09	1.00635	0.09012	0.08976	0.59	1.18116	0.65603	0.53047	1.09	1.69750	1.32567	0.79708
0.10	1.00775	0.10000	0.09956	0.60	1.18884	0.67249	0.53793	1.10	1.72100	1.34357	0.80099
0.11	1.00920	0.11022	0.10954	0.61	1.19697	0.68921	0.54540	1.11	1.74447	1.36171	0.80493
0.12	1.01070	0.12029	0.11943	0.62	1.20550	0.70620	0.55287	1.12	1.76794	1.38009	0.80890
0.13	1.01225	0.13037	0.12927	0.63	1.21448	0.72345	0.56035	1.13	1.79141	1.39871	0.81289
0.14	1.01384	0.14046	0.13909	0.64	1.22392	0.74096	0.56784	1.14	1.81488	1.41757	0.81690
0.15	1.01547	0.15055	0.14889	0.65	1.23382	0.75873	0.57534	1.15	1.83835	1.43667	0.82092
0.16	1.01714	0.16064	0.15893	0.66	1.24418	0.77676	0.58285	1.16	1.86182	1.45599	0.82495
0.17	1.01885	0.17073	0.16908	0.67	1.25500	0.79505	0.59037	1.17	1.88529	1.47553	0.82899
0.18	1.02060	0.18082	0.17935	0.68	1.26628	0.81360	0.59791	1.18	1.90876	1.49529	0.83303
0.19	1.02239	0.19091	0.18975	0.69	1.27802	0.83241	0.60546	1.19	1.93223	1.51527	0.83708
0.20	1.02422	0.20100	0.19997	0.70	1.29022	0.85148	0.61302	1.20	1.95570	1.53547	0.84113
0.21	1.02609	0.21115	0.20997	0.71	1.30288	0.87081	0.62059	1.21	1.97917	1.55589	0.84518
0.22	1.02799	0.22128	0.21997	0.72	1.31600	0.89040	0.62818	1.22	2.00264	1.57653	0.84923
0.23	1.02992	0.23140	0.22997	0.73	1.32958	0.91025	0.63579	1.23	2.02611	1.59739	0.85328
0.24	1.03188	0.24151	0.23997	0.74	1.34362	0.93036	0.64341	1.24	2.04958	1.61847	0.85733
0.25	1.03387	0.25161	0.24997	0.75	1.35812	0.95073	0.65104	1.25	2.07305	1.63977	0.86138
0.26	1.03588	0.26170	0.25997	0.76	1.37308	0.97136	0.65869	1.26	2.09652	1.66129	0.86543
0.27	1.03790	0.27179	0.26997	0.77	1.38850	0.99225	0.66635	1.27	2.12000	1.68303	0.86948
0.28	1.03993	0.28187	0.27997	0.78	1.40438	1.01340	0.67402	1.28	2.14347	1.70499	0.87353
0.29	1.04197	0.29195	0.28997	0.79	1.42072	1.03481	0.68170	1.29	2.16694	1.72717	0.87758
0.30	1.04402	0.30202	0.29997	0.80	1.43752	1.05648	0.68939	1.30	2.19041	1.74957	0.88163
0.31	1.04607	0.31209	0.30997	0.81	1.45478	1.07841	0.69709	1.31	2.21388	1.77219	0.88568
0.32	1.04812	0.32216	0.31997	0.82	1.47250	1.10060	0.70480	1.32	2.23735	1.79493	0.88973
0.33	1.05017	0.33223	0.32997	0.83	1.49067	1.12306	0.71252	1.33	2.26082	1.81779	0.89378
0.34	1.05222	0.34230	0.33997	0.84	1.50930	1.14579	0.72025	1.34	2.28429	1.84077	0.89783
0.35	1.05427	0.35237	0.34997	0.85	1.52838	1.16880	0.72796	1.35	2.30776	1.86387	0.90188
0.36	1.05632	0.36244	0.35997	0.86	1.54792	1.19209	0.73568	1.36	2.33123	1.88707	0.90593
0.37	1.05837	0.37251	0.36997	0.87	1.56792	1.21566	0.74341	1.37	2.35470	1.91037	0.90998
0.38	1.06042	0.38258	0.37997	0.88	1.58838	1.23951	0.75114	1.38	2.37817	1.93377	0.91403
0.39	1.06247	0.39265	0.38997	0.89	1.60930	1.26364	0.75887	1.39	2.40164	1.95727	0.91808
0.40	1.06452	0.40272	0.39997	0.90	1.63068	1.28805	0.76660	1.40	2.42511	1.98087	0.92213
0.41	1.06657	0.41279	0.40997	0.91	1.65252	1.31274	0.77433	1.41	2.44858	2.00457	0.92618
0.42	1.06862	0.42286	0.41997	0.92	1.67482	1.33771	0.78206	1.42	2.47205	2.02837	0.93023
0.43	1.07067	0.43293	0.42997	0.93	1.69758	1.36296	0.78979	1.43	2.49552	2.05217	0.93428
0.44	1.07272	0.44300	0.43997	0.94	1.72080	1.38848	0.79752	1.44	2.51900	2.07597	0.93833
0.45	1.07477	0.45307	0.44997	0.95	1.74448	1.41428	0.80525	1.45	2.54247	2.09977	0.94238
0.46	1.07682	0.46314	0.45997	0.96	1.76862	1.44036	0.81298	1.46	2.56594	2.12357	0.94643
0.47	1.07887	0.47321	0.46997	0.97	1.79322	1.46672	0.82071	1.47	2.58941	2.14737	0.95048
0.48	1.08092	0.48328	0.47997	0.98	1.81828	1.49336	0.82844	1.48	2.61288	2.17117	0.95453
0.49	1.08297	0.49335	0.48997	0.99	1.84380	1.52028	0.83617	1.49	2.63635	2.19497	0.95858
0.50	1.08502	0.50342	0.49997	1.00	1.86978	1.54748	0.84390	1.50	2.65982	2.21877	0.96263

TABLE 1A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 1/2$ and x from 0.00 to 1.50.

x	$F_{1/2}(x)$	$H_{1/2}(x)$	$T_{1/2}(x)$	x	$F_{1/2}(x)$	$H_{1/2}(x)$	$T_{1/2}(x)$	x	$F_{1/2}(x)$	$H_{1/2}(x)$	$T_{1/2}(x)$
1.50	2.35241	2.12928	0.90515	2.0	3.76220	3.62686	0.96403	6.0	201.71264	201.71316	0.99993
1.51	2.37282	2.15291	0.90694	2.1	4.14331	4.02189	0.97045	6.1	222.93001	222.93779	0.99999
1.52	2.39547	2.17676	0.90870	2.2	4.51911	4.39511	0.97574	6.2	246.37351	246.37351	0.99999
1.53	2.41736	2.20082	0.91042	2.3	4.89122	4.76696	0.98010	6.3	272.28667	272.28504	0.99999
1.54	2.43949	2.22510	0.91212	2.4	5.25925	5.14623	0.98367	6.4	300.92335	300.92169	0.99999
1.55	2.46186	2.24961	0.91379	2.5	5.62329	5.52020	0.98661	6.5	332.57157	332.57006	1.00000
1.56	2.48438	2.27434	0.91542	2.6	5.98301	5.89473	0.98903	6.6	367.35697	367.35691	1.00000
1.57	2.50725	2.29930	0.91703	2.7	6.33873	6.26629	0.99101	6.7	406.92430	406.92430	1.00000
1.58	2.53047	2.32449	0.91869	2.8	6.69053	6.63536	0.99396	6.8	449.92406	449.92406	1.00000
1.59	2.55384	2.34991	0.92035	2.9	7.03843	6.97953	0.99696	6.9	496.13766	496.13765	1.00000
1.60	2.57746	2.37557	0.92167	3.0	7.38242	7.32323	0.99905	7.0	545.93704	545.93704	1.00000
1.61	2.60135	2.40146	0.92316	3.1	7.72250	7.66455	0.99955	7.1	598.93395	598.93312	1.00000
1.62	2.62549	2.42760	0.92462	3.2	8.05865	7.99776	0.99968	7.2	654.74516	654.74516	1.00000
1.63	2.64990	2.45397	0.92606	3.3	8.39087	8.32786	0.99977	7.3	713.92320	713.92320	1.00000
1.64	2.67447	2.48059	0.92747	3.4	8.71915	8.65336	0.99977	7.4	776.92322	776.92322	1.00000
1.65	2.69921	2.50746	0.92886	3.5	9.04343	8.97423	0.99977	7.5	843.92322	843.92322	1.00000
1.66	2.72422	2.53459	0.93023	3.6	9.36371	9.29023	0.99977	7.6	914.92322	914.92322	1.00000
1.67	2.74946	2.56196	0.93155	3.7	9.67999	9.60110	0.99977	7.7	989.92322	989.92322	1.00000
1.68	2.77493	2.58959	0.93285	3.8	9.99227	9.90343	0.99977	7.8	1068.92322	1068.92322	1.00000
1.69	2.80060	2.61748	0.93415	3.9	10.30055	10.21435	0.99977	7.9	1151.92322	1151.92322	1.00000
1.70	2.82642	2.64563	0.93541	4.0	10.60483	10.51863	0.99977	8.0	1238.92322	1238.92322	1.00000
1.71	2.85241	2.67405	0.93665	4.1	10.90511	10.81891	0.99977	8.1	1329.92322	1329.92322	1.00000
1.72	2.87856	2.70273	0.93786	4.2	11.20139	11.11919	0.99977	8.2	1424.92322	1424.92322	1.00000
1.73	2.90487	2.73168	0.93906	4.3	11.49367	11.41117	0.99977	8.3	1523.92322	1523.92322	1.00000
1.74	2.93134	2.76091	0.94023	4.4	11.78195	11.70095	0.99977	8.4	1625.92322	1625.92322	1.00000
1.75	2.95799	2.79041	0.94138	4.5	12.06623	12.00095	0.99977	8.5	1730.92322	1730.92322	1.00000
1.76	2.98484	2.82020	0.94250	4.6	12.34651	12.28123	0.99977	8.6	1837.92322	1837.92322	1.00000
1.77	3.01189	2.85026	0.94361	4.7	12.62279	12.56151	0.99977	8.7	1946.92322	1946.92322	1.00000
1.78	3.03914	2.88061	0.94470	4.8	12.89507	12.84179	0.99977	8.8	2057.92322	2057.92322	1.00000
1.79	3.06651	2.91125	0.94576	4.9	13.16335	13.12208	0.99977	8.9	2170.92322	2170.92322	1.00000
1.80	3.09400	2.94217	0.94681	5.0	13.42763	13.38131	0.99977	9.0	2285.92322	2285.92322	1.00000
1.81	3.12161	2.97340	0.94783	5.1	13.68791	13.64095	0.99977	9.1	2401.92322	2401.92322	1.00000
1.82	3.14934	3.00492	0.94884	5.2	13.94419	13.89719	0.99977	9.2	2518.92322	2518.92322	1.00000
1.83	3.17718	3.03674	0.94983	5.3	14.19647	14.15041	0.99977	9.3	2636.92322	2636.92322	1.00000
1.84	3.20513	3.06886	0.95080	5.4	14.44475	14.40431	0.99977	9.4	2755.92322	2755.92322	1.00000
1.85	3.23319	3.10129	0.95175	5.5	14.68903	14.64859	0.99977	9.5	2875.92322	2875.92322	1.00000
1.86	3.26136	3.13403	0.95269	5.6	14.92931	14.88885	0.99977	9.6	2996.92322	2996.92322	1.00000
1.87	3.28964	3.16709	0.95362	5.7	15.16559	15.12515	0.99977	9.7	3118.92322	3118.92322	1.00000
1.88	3.31803	3.20045	0.95455	5.8	15.39787	15.36541	0.99977	9.8	3241.92322	3241.92322	1.00000
1.89	3.34652	3.23415	0.95547	5.9	15.62615	15.59369	0.99977	9.9	3365.92322	3365.92322	1.00000
1.90	3.37513	3.26816	0.95639	6.0	15.85043	15.81823	0.99977	10.0	3490.92322	3490.92322	1.00000
1.91	3.40385	3.30248	0.95730								
1.92	3.43268	3.33718	0.95821								
1.93	3.46161	3.37218	0.95912								
1.94	3.49064	3.40748	0.96003								
1.95	3.51977	3.44308	0.96094								
1.96	3.54899	3.47898	0.96185								
1.97	3.57830	3.51518	0.96276								
1.98	3.60771	3.55168	0.96367								
1.99	3.63722	3.58848	0.96458								
2.00	3.66683	3.62548	0.96549								

TABLE 1B. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 1/2$ and x from 1.50 to 10.0.

$$\alpha = 1/3$$

x	$F_{1/3}(x)$	$H_{2/3}(x)$	$T_{1/3}(x)$	x	$F_{1/3}(x)$	$H_{2/3}(x)$	$T_{1/3}(x)$	x	$F_{1/3}(x)$	$H_{2/3}(x)$	$T_{1/3}(x)$
0.0	1.00000	0.00128	0.00128	0.50	1.19193	0.24220	0.20571	1.00	1.82387	0.68895	0.37789
0.01	1.00008	0.00128	0.00128	0.51	1.19188	0.24214	0.20571	1.01	1.84096	0.70005	0.38009
0.02	1.00020	0.00129	0.00129	0.52	1.19183	0.24208	0.20571	1.02	1.85229	0.71121	0.38229
0.03	1.00036	0.00130	0.00130	0.53	1.19177	0.24202	0.20571	1.03	1.86387	0.72236	0.38448
0.04	1.00054	0.00131	0.00131	0.54	1.19172	0.24196	0.20571	1.04	1.87570	0.73342	0.38674
0.05	1.00073	0.00132	0.00132	0.55	1.19167	0.24190	0.20571	1.05	1.88778	0.74447	0.38895
0.06	1.00093	0.00133	0.00133	0.56	1.19162	0.24184	0.20571	1.06	1.89999	0.75552	0.39113
0.07	1.00114	0.00134	0.00134	0.57	1.19157	0.24178	0.20571	1.07	1.91232	0.76657	0.39328
0.08	1.00136	0.00135	0.00135	0.58	1.19152	0.24172	0.20571	1.08	1.92476	0.77762	0.39540
0.09	1.00159	0.00136	0.00136	0.59	1.19147	0.24166	0.20571	1.09	1.93731	0.78867	0.39748
0.10	1.00183	0.00137	0.00137	0.60	1.19142	0.24160	0.20571	1.10	1.95000	0.79972	0.39952
0.11	1.00208	0.00138	0.00138	0.61	1.19137	0.24154	0.20571	1.11	1.96276	0.81077	0.40156
0.12	1.00233	0.00139	0.00139	0.62	1.19132	0.24148	0.20571	1.12	1.97562	0.82182	0.40359
0.13	1.00259	0.00140	0.00140	0.63	1.19127	0.24142	0.20571	1.13	1.98858	0.83287	0.40559
0.14	1.00285	0.00141	0.00141	0.64	1.19122	0.24136	0.20571	1.14	2.00164	0.84392	0.40759
0.15	1.00311	0.00142	0.00142	0.65	1.19117	0.24130	0.20571	1.15	2.01470	0.85497	0.40955
0.16	1.00337	0.00143	0.00143	0.66	1.19112	0.24124	0.20571	1.16	2.02776	0.86602	0.41148
0.17	1.00363	0.00144	0.00144	0.67	1.19107	0.24118	0.20571	1.17	2.04082	0.87707	0.41338
0.18	1.00389	0.00145	0.00145	0.68	1.19102	0.24112	0.20571	1.18	2.05388	0.88812	0.41525
0.19	1.00415	0.00146	0.00146	0.69	1.19097	0.24106	0.20571	1.19	2.06694	0.89917	0.41709
0.20	1.00441	0.00147	0.00147	0.70	1.19092	0.24100	0.20571	1.20	2.07999	0.91022	0.41890
0.21	1.00467	0.00148	0.00148	0.71	1.19087	0.24094	0.20571	1.21	2.09305	0.92127	0.42068
0.22	1.00493	0.00149	0.00149	0.72	1.19082	0.24088	0.20571	1.22	2.10610	0.93232	0.42243
0.23	1.00519	0.00150	0.00150	0.73	1.19077	0.24082	0.20571	1.23	2.11916	0.94337	0.42416
0.24	1.00545	0.00151	0.00151	0.74	1.19072	0.24076	0.20571	1.24	2.13221	0.95442	0.42587
0.25	1.00571	0.00152	0.00152	0.75	1.19067	0.24070	0.20571	1.25	2.14526	0.96547	0.42756
0.26	1.00597	0.00153	0.00153	0.76	1.19062	0.24064	0.20571	1.26	2.15831	0.97652	0.42923
0.27	1.00623	0.00154	0.00154	0.77	1.19057	0.24058	0.20571	1.27	2.17136	0.98757	0.43088
0.28	1.00649	0.00155	0.00155	0.78	1.19052	0.24052	0.20571	1.28	2.18441	0.99862	0.43251
0.29	1.00675	0.00156	0.00156	0.79	1.19047	0.24046	0.20571	1.29	2.19746	1.00967	0.43413
0.30	1.00701	0.00157	0.00157	0.80	1.19042	0.24040	0.20571	1.30	2.21051	1.02072	0.43574
0.31	1.00727	0.00158	0.00158	0.81	1.19037	0.24034	0.20571	1.31	2.22356	1.03177	0.43734
0.32	1.00753	0.00159	0.00159	0.82	1.19032	0.24028	0.20571	1.32	2.23661	1.04282	0.43893
0.33	1.00779	0.00160	0.00160	0.83	1.19027	0.24022	0.20571	1.33	2.24966	1.05387	0.44051
0.34	1.00805	0.00161	0.00161	0.84	1.19022	0.24016	0.20571	1.34	2.26271	1.06492	0.44208
0.35	1.00831	0.00162	0.00162	0.85	1.19017	0.24010	0.20571	1.35	2.27576	1.07597	0.44364
0.36	1.00857	0.00163	0.00163	0.86	1.19012	0.24004	0.20571	1.36	2.28881	1.08702	0.44519
0.37	1.00883	0.00164	0.00164	0.87	1.19007	0.24000	0.20571	1.37	2.30186	1.09807	0.44673
0.38	1.00909	0.00165	0.00165	0.88	1.19002	0.23994	0.20571	1.38	2.31491	1.10912	0.44826
0.39	1.00935	0.00166	0.00166	0.89	1.19000	0.23988	0.20571	1.39	2.32796	1.12017	0.44978
0.40	1.00961	0.00167	0.00167	0.90	1.18995	0.23982	0.20571	1.40	2.34101	1.13122	0.45129
0.41	1.00987	0.00168	0.00168	0.91	1.18990	0.23976	0.20571	1.41	2.35406	1.14227	0.45279
0.42	1.01013	0.00169	0.00169	0.92	1.18985	0.23970	0.20571	1.42	2.36711	1.15332	0.45428
0.43	1.01039	0.00170	0.00170	0.93	1.18980	0.23964	0.20571	1.43	2.38016	1.16437	0.45576
0.44	1.01065	0.00171	0.00171	0.94	1.18975	0.23958	0.20571	1.44	2.39321	1.17542	0.45723
0.45	1.01091	0.00172	0.00172	0.95	1.18970	0.23952	0.20571	1.45	2.40626	1.18647	0.45869
0.46	1.01117	0.00173	0.00173	0.96	1.18965	0.23946	0.20571	1.46	2.41931	1.19752	0.46014
0.47	1.01143	0.00174	0.00174	0.97	1.18960	0.23940	0.20571	1.47	2.43236	1.20857	0.46158
0.48	1.01169	0.00175	0.00175	0.98	1.18955	0.23934	0.20571	1.48	2.44541	1.21962	0.46301
0.49	1.01195	0.00176	0.00176	0.99	1.18950	0.23928	0.20571	1.49	2.45846	1.23067	0.46443
0.50	1.01221	0.00177	0.00177	1.00	1.18945	0.23922	0.20571	1.50	2.47151	1.24172	0.46584

TABLE 2A. Lanchester-Clifford-Schl fli Functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and

$T_\alpha(x)$ for $\alpha = 1/3$ and x from 0.00 to 1.50.

x	$F_{1/3}(x)$	$H_{2/3}(x)$	$T_{1/3}(x)$	x	$F_{1/3}(x)$	$H_{2/3}(x)$	$T_{1/3}(x)$	x	$F_{1/3}(x)$	$H_{2/3}(x)$	$T_{1/3}(x)$
1.50	3.07330	1.40540	0.45729	2.0	5.23834	2.58494	0.48788	6.0	359.65982	181.79456	0.50547
1.51	3.10669	1.42363	0.45825	2.1	5.99906	2.90136	0.49130	6.1	398.71187	201.53431	0.50546
1.52	3.14047	1.44204	0.45918	2.2	6.58916	3.25333	0.49314	6.2	441.91922	223.40445	0.50546
1.53	3.17464	1.46064	0.46010	2.3	7.34609	3.64236	0.49509	6.3	489.91515	247.63498	0.50547
1.54	3.20920	1.47943	0.46100	2.4	8.18812	4.07426	0.49755	6.4	543.02156	274.47812	0.50547
1.55	3.24416	1.49840	0.46188	2.5	9.12445	4.55335	0.49998	6.5	601.85114	304.21683	0.50547
1.56	3.27957	1.51757	0.46274	2.6	10.15726	5.07921	0.50202	6.6	675.23262	337.319923	0.50547
1.57	3.31543	1.53692	0.46359	2.7	11.28290	5.65394	0.50419	6.7	755.15934	374.10772	0.50547
1.58	3.35174	1.55642	0.46452	2.8	12.50424	6.27944	0.50633	6.8	851.78370	414.20770	0.50547
1.59	3.38852	1.57622	0.46554	2.9	13.82445	6.95355	0.50843	6.9	967.78370	458.85468	0.50547
1.60	3.42581	1.59617	0.46663	3.0	15.24445	7.67921	0.51053	7.0	1095.90333	508.455229	0.50547
1.61	3.46368	1.61632	0.46782	3.1	16.76931	8.45832	0.51263	7.1	1234.58667	563.38731	0.50547
1.62	3.50217	1.63668	0.46909	3.2	18.39319	9.29423	0.51473	7.2	1384.28181	624.23262	0.50547
1.63	3.54124	1.65724	0.47048	3.3	20.11609	10.18417	0.51683	7.3	1545.93846	691.62223	0.50547
1.64	3.58093	1.67800	0.47197	3.4	21.93917	11.12968	0.51893	7.4	1719.9784	766.25801	0.50547
1.65	3.62127	1.69898	0.47356	3.5	23.86248	12.13131	0.52103	7.5	1906.9784	848.91694	0.50547
1.66	3.66226	1.72017	0.47515	3.6	25.88609	13.18955	0.52313	7.6	2107.4784	941.83675	0.50547
1.67	3.70385	1.74150	0.47674	3.7	28.00981	14.30868	0.52523	7.7	2321.9784	1041.10372	0.50547
1.68	3.74605	1.76300	0.47833	3.8	30.23363	15.48417	0.52733	7.8	2550.1952	1148.42680	0.50547
1.69	3.78885	1.78467	0.47992	3.9	32.55745	16.71955	0.52943	7.9	2801.55706	1264.97741	0.50547
1.70	3.83225	1.80650	0.48151	4.0	34.98127	18.01868	0.53153	8.0	3076.15501	1391.45507	0.50547
1.71	3.87625	1.82849	0.48310	4.1	37.50509	19.38417	0.53363	8.1	3374.90155	1528.45507	0.50547
1.72	3.92085	1.85064	0.48469	4.2	40.12891	20.81955	0.53573	8.2	3697.9784	1676.45507	0.50547
1.73	3.96605	1.87294	0.48628	4.3	42.85273	22.32968	0.53783	8.3	4045.9784	1835.45507	0.50547
1.74	4.01185	1.89539	0.48787	4.4	45.67655	23.90955	0.53993	8.4	4419.9784	2006.45507	0.50547
1.75	4.05825	1.91794	0.48946	4.5	48.60037	25.55417	0.54203	8.5	4819.9784	2189.45507	0.50547
1.76	4.10525	1.94064	0.49105	4.6	51.62419	27.26868	0.54413	8.6	5244.9784	2384.45507	0.50547
1.77	4.15285	1.96349	0.49264	4.7	54.74801	29.04868	0.54623	8.7	5694.9784	2591.45507	0.50547
1.78	4.20105	1.98639	0.49423	4.8	57.97183	30.89417	0.54833	8.8	6169.9784	2810.45507	0.50547
1.79	4.24985	2.00944	0.49582	4.9	61.29565	32.80468	0.55043	8.9	6669.9784	3042.45507	0.50547
1.80	4.29925	2.03264	0.49741	5.0	64.71947	34.78468	0.55253	9.0	7194.9784	3287.45507	0.50547
1.81	4.34925	2.05599	0.49899	5.1	68.24329	36.82968	0.55463	9.1	7744.9784	3544.45507	0.50547
1.82	4.39985	2.07949	0.49958	5.2	71.86711	38.94468	0.55673	9.2	8319.9784	3813.45507	0.50547
1.83	4.45105	2.10314	0.50117	5.3	75.59093	41.12968	0.55883	9.3	8919.9784	4094.45507	0.50547
1.84	4.50285	2.12694	0.50276	5.4	79.41475	43.38468	0.56093	9.4	9544.9784	4387.45507	0.50547
1.85	4.55525	2.15084	0.50435	5.5	83.33857	45.70968	0.56303	9.5	10194.9784	4692.45507	0.50547
1.86	4.60825	2.17484	0.50594	5.6	87.36239	48.10468	0.56513	9.6	10869.9784	5009.45507	0.50547
1.87	4.66185	2.19894	0.50753	5.7	91.48621	50.57468	0.56723	9.7	11569.9784	5338.45507	0.50547
1.88	4.71605	2.22314	0.50912	5.8	95.70993	53.11968	0.56933	9.8	12294.9784	5679.45507	0.50547
1.89	4.77085	2.24744	0.51071	5.9	100.03375	55.73968	0.57143	9.9	13044.9784	6032.45507	0.50547
1.90	4.82625	2.27184	0.51230	6.0	104.45757	58.43468	0.57353	10.0	13819.9784	6397.45507	0.50547
1.91	4.88225	2.29634	0.51389								
1.92	4.93885	2.32094	0.51548								
1.93	4.99505	2.34564	0.51707								
1.94	5.05185	2.37044	0.51866								
1.95	5.10925	2.39534	0.52025								
1.96	5.16725	2.42034	0.52184								
1.97	5.22585	2.44544	0.52343								
1.98	5.28505	2.47064	0.52502								
1.99	5.34485	2.49594	0.52661								
2.00	5.40525	2.52134	0.52820								

TABLE 2B. Lanchester-Clifford-Schlafli Functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and $T_\alpha(x)$ for $\alpha = 1/3$ and x from 1.50 to 10.0.

$$\alpha = 2/3$$

x	$F_{2/3}(x)$	$H_{1/3}(x)$	$T_{2/3}(x)$	x	$F_{2/3}(x)$	$H_{1/3}(x)$	$T_{2/3}(x)$	x	$F_{2/3}(x)$	$H_{1/3}(x)$	$T_{2/3}(x)$
0.0	1.00000	0.08772	0.08772	0.50	1.09552	1.2711	1.1937	1.00	1.40923	2.26370	1.61230
0.01	1.00004	0.08772	0.08772	0.51	1.09449	1.26903	1.1937	1.01	1.40726	2.26364	1.61230
0.02	1.00015	0.08770	0.08770	0.52	1.09346	1.26695	1.1937	1.02	1.40529	2.26358	1.61230
0.03	1.00020	0.08761	0.08761	0.53	1.09243	1.26487	1.1937	1.03	1.40332	2.26352	1.61230
0.04	1.00029	0.08752	0.08752	0.54	1.09140	1.26279	1.1937	1.04	1.40135	2.26346	1.61230
0.05	1.00038	0.08743	0.08743	0.55	1.09037	1.26071	1.1937	1.05	1.39938	2.26340	1.61230
0.06	1.00047	0.08734	0.08734	0.56	1.08934	1.25863	1.1937	1.06	1.39741	2.26334	1.61230
0.07	1.00056	0.08725	0.08725	0.57	1.08831	1.25655	1.1937	1.07	1.39544	2.26328	1.61230
0.08	1.00065	0.08716	0.08716	0.58	1.08728	1.25447	1.1937	1.08	1.39347	2.26322	1.61230
0.09	1.00074	0.08707	0.08707	0.59	1.08625	1.25239	1.1937	1.09	1.39150	2.26316	1.61230
0.10	1.00083	0.08698	0.08698	0.60	1.08522	1.25031	1.1937	1.10	1.38953	2.26310	1.61230
0.11	1.00092	0.08689	0.08689	0.61	1.08419	1.24823	1.1937	1.11	1.38756	2.26304	1.61230
0.12	1.00101	0.08680	0.08680	0.62	1.08316	1.24615	1.1937	1.12	1.38559	2.26298	1.61230
0.13	1.00110	0.08671	0.08671	0.63	1.08213	1.24407	1.1937	1.13	1.38362	2.26292	1.61230
0.14	1.00119	0.08662	0.08662	0.64	1.08110	1.24199	1.1937	1.14	1.38165	2.26286	1.61230
0.15	1.00128	0.08653	0.08653	0.65	1.08007	1.23991	1.1937	1.15	1.37968	2.26280	1.61230
0.16	1.00137	0.08644	0.08644	0.66	1.07904	1.23783	1.1937	1.16	1.37771	2.26274	1.61230
0.17	1.00146	0.08635	0.08635	0.67	1.07801	1.23575	1.1937	1.17	1.37574	2.26268	1.61230
0.18	1.00155	0.08626	0.08626	0.68	1.07698	1.23367	1.1937	1.18	1.37377	2.26262	1.61230
0.19	1.00164	0.08617	0.08617	0.69	1.07595	1.23159	1.1937	1.19	1.37180	2.26256	1.61230
0.20	1.00173	0.08608	0.08608	0.70	1.07492	1.22951	1.1937	1.20	1.36983	2.26250	1.61230
0.21	1.00182	0.08599	0.08599	0.71	1.07389	1.22743	1.1937	1.21	1.36786	2.26244	1.61230
0.22	1.00191	0.08590	0.08590	0.72	1.07286	1.22535	1.1937	1.22	1.36589	2.26238	1.61230
0.23	1.00200	0.08581	0.08581	0.73	1.07183	1.22327	1.1937	1.23	1.36392	2.26232	1.61230
0.24	1.00209	0.08572	0.08572	0.74	1.07080	1.22119	1.1937	1.24	1.36195	2.26226	1.61230
0.25	1.00218	0.08563	0.08563	0.75	1.06977	1.21911	1.1937	1.25	1.35998	2.26220	1.61230
0.26	1.00227	0.08554	0.08554	0.76	1.06874	1.21703	1.1937	1.26	1.35801	2.26214	1.61230
0.27	1.00236	0.08545	0.08545	0.77	1.06771	1.21495	1.1937	1.27	1.35604	2.26208	1.61230
0.28	1.00245	0.08536	0.08536	0.78	1.06668	1.21287	1.1937	1.28	1.35407	2.26202	1.61230
0.29	1.00254	0.08527	0.08527	0.79	1.06565	1.21079	1.1937	1.29	1.35210	2.26196	1.61230
0.30	1.00263	0.08518	0.08518	0.80	1.06462	1.20871	1.1937	1.30	1.35013	2.26190	1.61230
0.31	1.00272	0.08509	0.08509	0.81	1.06359	1.20663	1.1937	1.31	1.34816	2.26184	1.61230
0.32	1.00281	0.08500	0.08500	0.82	1.06256	1.20455	1.1937	1.32	1.34619	2.26178	1.61230
0.33	1.00290	0.08491	0.08491	0.83	1.06153	1.20247	1.1937	1.33	1.34422	2.26172	1.61230
0.34	1.00299	0.08482	0.08482	0.84	1.06050	1.20039	1.1937	1.34	1.34225	2.26166	1.61230
0.35	1.00308	0.08473	0.08473	0.85	1.05947	1.19831	1.1937	1.35	1.34028	2.26160	1.61230
0.36	1.00317	0.08464	0.08464	0.86	1.05844	1.19623	1.1937	1.36	1.33831	2.26154	1.61230
0.37	1.00326	0.08455	0.08455	0.87	1.05741	1.19415	1.1937	1.37	1.33634	2.26148	1.61230
0.38	1.00335	0.08446	0.08446	0.88	1.05638	1.19207	1.1937	1.38	1.33437	2.26142	1.61230
0.39	1.00344	0.08437	0.08437	0.89	1.05535	1.19000	1.1937	1.39	1.33240	2.26136	1.61230
0.40	1.00353	0.08428	0.08428	0.90	1.05432	1.18792	1.1937	1.40	1.33043	2.26130	1.61230
0.41	1.00362	0.08419	0.08419	0.91	1.05329	1.18584	1.1937	1.41	1.32846	2.26124	1.61230
0.42	1.00371	0.08410	0.08410	0.92	1.05226	1.18377	1.1937	1.42	1.32649	2.26118	1.61230
0.43	1.00380	0.08401	0.08401	0.93	1.05123	1.18169	1.1937	1.43	1.32452	2.26112	1.61230
0.44	1.00389	0.08392	0.08392	0.94	1.05020	1.17961	1.1937	1.44	1.32255	2.26106	1.61230
0.45	1.00398	0.08383	0.08383	0.95	1.04917	1.17754	1.1937	1.45	1.32058	2.26100	1.61230
0.46	1.00407	0.08374	0.08374	0.96	1.04814	1.17546	1.1937	1.46	1.31861	2.26094	1.61230
0.47	1.00416	0.08365	0.08365	0.97	1.04711	1.17339	1.1937	1.47	1.31664	2.26088	1.61230
0.48	1.00425	0.08356	0.08356	0.98	1.04608	1.17131	1.1937	1.48	1.31467	2.26082	1.61230
0.49	1.00434	0.08347	0.08347	0.99	1.04505	1.16924	1.1937	1.49	1.31270	2.26076	1.61230
0.50	1.00443	0.08338	0.08338	1.00	1.04402	1.16716	1.1937	1.50	1.31073	2.26070	1.61230

TABLE 3A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 2/3$ and x from 0.00 to 1.50.

$\alpha = 2/3$

x	$F_{2/3}(x)$	$H_{1/3}(x)$	$T_{2/3}(x)$	x	$F_{2/3}(x)$	$H_{1/3}(x)$	$T_{2/3}(x)$	x	$F_{2/3}(x)$	$H_{1/3}(x)$	$T_{2/3}(x)$
1.50	1.99654	3.65444	1.83039	2.0	3.01025	5.78325	1.92118	6.0	129.97145	251.7145	1.97834
1.51	2.02781	3.68442	1.83113	2.0	3.28205	6.33842	1.93124	6.1	143.21119	283.32268	1.97835
1.52	2.06795	3.72268	1.83182	2.2	3.58116	6.94779	1.93955	6.2	157.80940	315.20230	1.97835
1.53	2.05975	3.75724	1.83466	2.3	3.91334	7.61702	1.94643	6.3	173.50393	348.04335	1.97835
1.54	2.05975	3.79211	1.84105	2.4	4.27863	8.35230	1.95209	6.4	191.64943	379.15061	1.97836
1.55	2.07598	3.82727	1.84360	2.5	4.68442	9.16046	1.95677	6.5	211.5598	417.80925	1.97836
1.56	2.09237	3.86274	1.84610	2.6	5.12742	10.0399	1.96076	6.6	232.58098	459.50936	1.97836
1.57	2.10955	3.89853	1.84856	2.7	5.60590	11.0009	1.9639	6.7	255.8098	505.71170	1.97836
1.58	2.1270	3.93463	1.85099	2.8	6.11791	12.05351	1.96639	6.8	282.24479	556.95936	1.97836
1.59	2.14262	3.97103	1.85339	2.9	6.67491	13.28351	1.96854	6.9	311.7170	613.71170	1.97836
1.60	2.15973	4.00777	1.85568	3.0	7.28228	14.68471	1.97030	7.0	343.64598	679.85596	1.97836
1.61	2.17701	4.04483	1.85787	3.1	7.94028	16.1669	1.97171	7.1	378.82779	749.45850	1.97836
1.62	2.19448	4.08221	1.86022	3.2	8.64910	17.73281	1.97294	7.2	417.62741	823.21836	1.97836
1.63	2.21213	4.11993	1.86243	3.3	9.40960	19.38776	1.97391	7.3	460.41794	901.81370	1.97836
1.64	2.22996	4.15758	1.86459	3.4	10.22486	21.23775	1.97471	7.4	507.61109	984.23900	1.97836
1.65	2.24798	4.19506	1.86672	3.5	11.0982	23.24063	1.97537	7.5	559.66109	1071.21292	1.97836
1.66	2.26619	4.23230	1.86881	3.6	12.02744	25.4063	1.97597	7.6	617.66916	1162.74706	1.97836
1.67	2.28459	4.26937	1.87086	3.7	13.01644	27.73577	1.97635	7.7	680.28836	1259.05555	1.97836
1.68	2.30319	4.30630	1.87288	3.8	14.06849	30.23364	1.97671	7.8	750.22894	1360.25569	1.97836
1.69	2.32197	4.34307	1.87486	3.9	15.18949	32.90364	1.97701	7.9	827.26421	1466.65967	1.97836
1.70	2.34095	4.37935	1.87681	4.0	16.3744	35.74063	1.97726	8.0	912.23739	1588.73751	1.97836
1.71	2.36012	4.41480	1.87872	4.1	17.6244	37.51199	1.97746	8.1	912.23739	1804.73751	1.97836
1.72	2.37950	4.44780	1.88059	4.2	18.97175	41.26358	1.97762	8.2	1005.26825	1990.11133	1.97836
1.73	2.39907	4.47809	1.88243	4.3	20.42517	45.9611	1.97775	8.3	109.26164	2194.72114	1.97836
1.74	2.41885	4.50568	1.88424	4.4	21.98344	49.64852	1.97786	8.4	123.31598	2420.36220	1.97836
1.75	2.43883	4.53166	1.88591	4.5	23.64867	54.36375	1.97795	8.5	134.92309	2669.24429	1.97836
1.76	2.45901	4.55611	1.88749	4.6	25.42524	60.08916	1.97803	8.6	148.02882	2943.84283	1.97836
1.77	2.47940	4.57917	1.88900	4.7	27.31625	65.76297	1.97809	8.7	161.06220	3243.75232	1.97836
1.78	2.50002	4.60073	1.89049	4.8	29.32517	71.28475	1.97814	8.8	174.106220	3569.95219	1.97836
1.79	2.52082	4.62159	1.89199	4.9	31.45867	77.6601	1.97818	8.9	186.61466	3943.65226	1.97836
1.80	2.54185	4.64130	1.89341	5.0	33.71617	84.82076	1.97821	9.0	202.00570	4356.36923	1.97836
1.81	2.56309	4.65961	1.89481	5.1	36.09866	97.79625	1.97824	9.1	220.02550	4809.10143	1.97836
1.82	2.58455	4.67643	1.89619	5.2	38.61254	107.68763	1.97826	9.2	242.96164	5300.17714	1.97836
1.83	2.60623	4.69245	1.89756	5.3	41.26879	118.58886	1.97828	9.3	270.9046	5846.3194	1.97836
1.84	2.62813	4.70779	1.90059	5.4	44.06825	130.60352	1.97830	9.4	302.96315	6449.0839	1.97836
1.85	2.65025	4.72241	1.90307	5.5	47.01202	143.84591	1.97831	9.5	339.57862	7113.95767	1.97836
1.86	2.67259	4.73632	1.90551	5.6	50.10966	158.44205	1.97832	9.6	381.60960	7847.59643	1.97836
1.87	2.69517	4.74955	1.90793	5.7	53.36287	174.53299	1.97833	9.7	429.27251	8651.09970	1.97836
1.88	2.71791	4.76211	1.91036	5.8	56.78166	192.25599	1.97834	9.8	482.73666	9556.23026	1.97836
1.89	2.74100	4.77401	1.91271	5.9	60.36803	211.81628	1.97834	9.9	542.54829	10535.7517	1.97836
1.90	2.76427	4.78524	1.91504	6.0	64.12145	233.36842	1.97834	10.0	587.514679	11623.18037	1.97836
1.91	2.78777	4.79581	1.91736			257.12820			6481.66483	12823.09399	1.97836
1.92	2.81152	4.80571	1.91960								
1.93	2.83550	4.81495	1.92188								
1.94	2.85972	4.82351	1.92417								
1.95	2.88418	4.83136	1.92645								
1.96	2.90880	4.83855	1.92871								
1.97	2.93359	4.84507	1.93094								
1.98	2.95845	4.85091	1.93316								
1.99	2.98336	4.85607	1.93533								
2.00	3.01025	4.86054	1.93745								

TABLE 3B. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 2/3$ and x from 1.50 to 10.0.

$$\alpha = 1/4$$

x	$F_{1/4}(x)$	$H_{3/4}(x)$	$T_{1/4}(x)$	x	$F_{1/4}(x)$	$H_{3/4}(x)$	$T_{1/4}(x)$	x	$F_{1/4}(x)$	$H_{3/4}(x)$	$T_{1/4}(x)$
0.0	1.00000	0.0	0.00047	0.50	1.25631	0.17259	0.13746	1.00	2.10378	0.54188	0.25757
0.01	1.00010	0.00047	0.00133	0.51	1.26993	0.17815	0.14091	1.01	2.12317	0.55153	0.25916
0.02	1.00040	0.00133	0.00435	0.52	1.27779	0.18368	0.14395	1.02	2.13990	0.56129	0.26074
0.03	1.00090	0.00245	0.00877	0.53	1.28887	0.18946	0.14695	1.03	2.15926	0.57115	0.26234
0.04	1.00160	0.00377	0.01264	0.54	1.30020	0.19496	0.14995	1.04	2.17793	0.58111	0.26394
0.05	1.00250	0.00527	0.01698	0.55	1.31175	0.20071	0.15301	1.05	2.19912	0.59118	0.26554
0.06	1.00360	0.00693	0.02126	0.56	1.32355	0.20666	0.15606	1.06	2.22262	0.60136	0.26714
0.07	1.00490	0.00874	0.02549	0.57	1.33568	0.21282	0.15905	1.07	2.24846	0.61164	0.26874
0.08	1.00640	0.01078	0.02964	0.58	1.34815	0.21923	0.16203	1.08	2.27666	0.62202	0.27034
0.09	1.00810	0.01298	0.03371	0.59	1.36097	0.22583	0.16498	1.09	2.30631	0.63250	0.27194
0.10	1.01001	0.01533	0.03770	0.60	1.37413	0.23264	0.16789	1.10	2.33743	0.64318	0.27354
0.11	1.01211	0.01783	0.04162	0.61	1.38763	0.23964	0.17078	1.11	2.37000	0.65391	0.27514
0.12	1.01442	0.02046	0.04548	0.62	1.39939	0.24693	0.17364	1.12	2.40412	0.66472	0.27674
0.13	1.01693	0.02315	0.04929	0.63	1.41289	0.25443	0.17647	1.13	2.44085	0.67572	0.27834
0.14	1.01964	0.02586	0.05306	0.64	1.42663	0.26215	0.17927	1.14	2.47923	0.68679	0.27994
0.15	1.02255	0.02847	0.05687	0.65	1.44063	0.26999	0.18203	1.15	2.50919	0.69799	0.28154
0.16	1.02567	0.03028	0.06062	0.66	1.45488	0.27800	0.18476	1.16	2.54004	0.70930	0.28314
0.17	1.02898	0.03219	0.06433	0.67	1.46939	0.28623	0.18746	1.17	2.57184	0.72074	0.28474
0.18	1.03251	0.03419	0.06799	0.68	1.48417	0.29467	0.19015	1.18	2.60455	0.73229	0.28634
0.19	1.03623	0.03624	0.07161	0.69	1.49917	0.30331	0.19285	1.19	2.62844	0.74397	0.28794
0.20	1.04016	0.03845	0.07519	0.70	1.51445	0.31214	0.19555	1.20	2.65343	0.75577	0.28954
0.21	1.04429	0.04086	0.07874	0.71	1.52999	0.32114	0.19771	1.21	2.67942	0.76770	0.29114
0.22	1.04863	0.04346	0.08226	0.72	1.54580	0.33031	0.20044	1.22	2.72520	0.77974	0.29274
0.23	1.05318	0.04615	0.08574	0.73	1.56186	0.33964	0.20323	1.23	2.77189	0.79193	0.29434
0.24	1.05793	0.04893	0.08919	0.74	1.57820	0.34915	0.20603	1.24	2.81949	0.80424	0.29594
0.25	1.06289	0.05181	0.09261	0.75	1.59481	0.35883	0.20883	1.25	2.86793	0.81668	0.29754
0.26	1.06806	0.05479	0.09600	0.76	1.61168	0.36867	0.21163	1.26	2.91723	0.82929	0.29914
0.27	1.07343	0.05786	0.09936	0.77	1.62882	0.37867	0.21443	1.27	2.96739	0.84195	0.30074
0.28	1.07902	0.06093	0.10269	0.78	1.64623	0.38883	0.21717	1.28	3.01833	0.85479	0.30234
0.29	1.08481	0.06400	0.10600	0.79	1.66390	0.39915	0.21987	1.29	3.07006	0.86777	0.30394
0.30	1.09081	0.06719	0.07193	0.80	1.68194	0.36964	0.22144	1.30	3.09408	0.88088	0.30554
0.31	1.09703	0.07049	0.07519	0.81	1.70021	0.37664	0.22194	1.31	3.09955	0.89413	0.30714
0.32	1.10345	0.07389	0.07847	0.82	1.71875	0.38470	0.22333	1.32	3.06626	0.90757	0.30874
0.33	1.11009	0.07739	0.08176	0.83	1.73759	0.39286	0.22558	1.33	3.03001	0.92125	0.31034
0.34	1.11694	0.08090	0.08506	0.84	1.75671	0.40070	0.22809	1.34	3.01422	0.93543	0.31194
0.35	1.12401	0.08443	0.08837	0.85	1.77612	0.40822	0.23049	1.35	3.17787	0.94955	0.31354
0.36	1.13129	0.08799	0.09168	0.86	1.79582	0.41543	0.23283	1.36	3.21599	0.96322	0.31514
0.37	1.13878	0.09158	0.09500	0.87	1.81583	0.42252	0.23513	1.37	3.25357	0.97653	0.31674
0.38	1.14639	0.09518	0.09833	0.88	1.83612	0.42959	0.23743	1.38	3.29161	0.99000	0.31834
0.39	1.15412	0.09879	0.10163	0.89	1.85671	0.43663	0.23968	1.39	3.33313	1.00351	0.31994
0.40	1.16258	0.10240	0.10494	0.90	1.87761	0.44371	0.24190	1.40	3.37312	1.01708	0.32154
0.41	1.17094	0.10602	0.10825	0.91	1.89881	0.45081	0.24419	1.41	3.41360	1.03069	0.32314
0.42	1.17953	0.10964	0.11155	0.92	1.92032	0.45793	0.24643	1.42	3.45456	1.04436	0.32474
0.43	1.18834	0.11326	0.11483	0.93	1.94214	0.46506	0.24866	1.43	3.49591	1.05807	0.32634
0.44	1.19738	0.11689	0.11811	0.94	1.96427	0.47221	0.25089	1.44	3.53795	1.07181	0.32794
0.45	1.20663	0.12052	0.12138	0.95	1.98672	0.47936	0.25313	1.45	3.58040	1.08557	0.32954
0.46	1.21611	0.12415	0.12463	0.96	2.00949	0.48653	0.25536	1.46	3.62335	1.09930	0.33114
0.47	1.22582	0.12778	0.12786	0.97	2.03257	0.49371	0.25759	1.47	3.66681	1.11307	0.33274
0.48	1.23574	0.13141	0.13108	0.98	2.05597	0.50090	0.25982	1.48	3.71079	1.12681	0.33434
0.49	1.24586	0.13504	0.13428	0.99	2.07971	0.50810	0.26205	1.49	3.75529	1.14056	0.33594
0.50	1.25631	0.13746	0.13746	1.00	2.10378	0.51538	0.26428	1.50	3.80031	1.15431	0.33754

TABLE 4A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 1/4$ and x from 0.00 to 1.50.

x	$F_{1/4}(x)$	$H_{3/4}(x)$	$T_{1/4}(x)$	x	$F_{1/4}(x)$	$H_{3/4}(x)$	$T_{1/4}(x)$	x	$F_{1/4}(x)$	$H_{3/4}(x)$	$T_{1/4}(x)$	$\alpha = 1/4$
1.50	3.80031	1.17433	0.30901	2.0	9.85616	2.24675	0.37770	6.0	527.67626	178.34720	0.33799	
1.51	3.84587	1.19069	0.30960	2.1	7.70085	2.53835	0.37962	6.1	585.89365	198.0419	0.33799	
1.52	3.89196	1.20722	0.31018	2.2	8.64488	2.86267	0.38118	6.2	650.47799	219.85313	0.33799	
1.53	3.93859	1.22393	0.31075	2.3	9.69099	3.22343	0.38245	6.3	722.2168	240.98805	0.33799	
1.54	3.98577	1.24081	0.31131	2.4	10.86951	3.62347	0.38348	6.4	801.59208	270.92829	0.33799	
1.55	4.03350	1.25788	0.31186	2.5	12.17756	4.07113	0.38431	6.5	889.73967	300.72132	0.33799	
1.56	4.08162	1.27505	0.31242	2.6	15.63112	4.56777	0.38500	6.6	987.50627	333.76568	0.33799	
1.57	4.13007	1.29237	0.31297	2.7	17.25375	5.11007	0.38555	6.7	1095.93811	370.41439	0.33799	
1.58	4.17877	1.30977	0.31354	2.8	19.07686	5.71448	0.38600	6.8	1216.19118	411.05867	0.33799	
1.59	4.22770	1.32727	0.31411	2.9	19.07686	6.41786	0.38637	6.9	1349.54844	456.11200	0.33799	
1.60	4.28065	1.34496	0.31463	3.0	21.32051	7.17792	0.38667	7.0	1497.43105	506.11478	0.33799	
1.61	4.33382	1.36281	0.31518	3.1	23.81195	8.02318	0.38691	7.1	1661.41396	561.53928	0.33799	
1.62	4.38738	1.38075	0.31573	3.2	26.58119	8.96313	0.38711	7.2	1843.24219	622.99219	0.33799	
1.63	4.44139	1.39885	0.31628	3.3	29.64729	10.00828	0.38727	7.3	2043.57739	691.16345	0.33799	
1.64	4.49580	1.41708	0.31683	3.4	33.10675	11.17032	0.38740	7.4	2268.57739	766.88645	0.33799	
1.65	4.55064	1.43542	0.31738	3.5	36.92382	12.46219	0.38751	7.5	2516.20022	850.44792	0.33799	
1.66	4.60592	1.45386	0.31792	3.6	41.16300	13.89826	0.38760	7.6	2790.94770	943.30954	0.33799	
1.67	4.66164	1.47240	0.31846	3.7	45.88641	15.49447	0.38767	7.7	3095.53215	1046.25608	0.33799	
1.68	4.71779	1.49104	0.31900	3.8	51.13136	17.26852	0.38773	7.8	3433.18381	1160.31849	0.33799	
1.69	4.77432	1.50977	0.31954	3.9	56.96608	19.24004	0.38778	7.9	3807.47665	1286.88542	0.33799	
1.70	4.83123	1.52859	0.32007	4.0	63.43934	21.43078	0.38782	8.0	4222.37283	1427.11584	0.33799	
1.71	4.88854	1.54740	0.32060	4.1	70.63317	23.86490	0.38785	8.1	4682.26100	1584.53506	0.33799	
1.72	4.94625	1.56621	0.32112	4.2	78.63360	26.59199	0.38787	8.2	5182.09274	1760.99226	0.33799	
1.73	5.00436	1.58502	0.32164	4.3	87.53252	29.57036	0.38789	8.3	5723.77345	1954.97269	0.33799	
1.74	5.06287	1.60383	0.32216	4.4	97.53940	32.91036	0.38791	8.4	6309.16935	2157.44170	0.33799	
1.75	5.12178	1.62264	0.32268	4.5	108.35735	36.61674	0.38793	8.5	7077.16850	2392.00581	0.33799	
1.76	5.18109	1.64145	0.32319	4.6	120.53320	40.73304	0.38794	8.6	7946.60050	2651.96407	0.33799	
1.77	5.24080	1.66026	0.32370	4.7	134.05121	45.31420	0.38795	8.7	8948.67355	2940.05690	0.33799	
1.78	5.30091	1.67907	0.32421	4.8	149.07356	50.38008	0.38796	8.8	9983.26842	3259.31970	0.33799	
1.79	5.36142	1.69788	0.32472	4.9	165.74688	56.01595	0.38796	8.9	10990.03241	3613.11455	0.33799	
1.80	5.42233	1.71669	0.32523	5.0	184.25850	62.27311	0.38797	9.0	11949.98315	4005.16529	0.33799	
1.81	5.48364	1.73550	0.32574	5.1	204.80949	69.21950	0.38797	9.1	13135.52334	4438.59336	0.33799	
1.82	5.54535	1.75431	0.32625	5.2	227.62717	76.93050	0.38797	9.2	14527.5882	4918.31205	0.33799	
1.83	5.60746	1.77312	0.32676	5.3	252.94573	85.49694	0.38798	9.3	16166.2600	5445.3129	0.33799	
1.84	5.67007	1.79193	0.32727	5.4	281.05178	94.98954	0.38798	9.4	17986.2600	6043.3129	0.33799	
1.85	5.73318	1.81074	0.32778	5.5	312.4535	105.53289	0.38798	9.5	19923.64472	6700.17618	0.33799	
1.86	5.79679	1.82955	0.32829	5.6	346.86301	117.23353	0.38798	9.6	21970.17593	7425.68038	0.33799	
1.87	5.86080	1.84836	0.32880	5.7	385.27813	130.21764	0.38798	9.7	24348.37283	8229.48506	0.33799	
1.88	5.92531	1.86717	0.32931	5.8	427.90458	144.6251	0.38798	9.8	26983.17397	9120.01918	0.33799	
1.89	5.99032	1.88598	0.32982	5.9	475.20112	160.61101	0.38799	9.9	29902.19693	10106.61719	0.33799	
1.90	6.05583	1.90479	0.33033	6.0	527.67626	178.34720	0.33799	10.0	33136.02562	11199.61611	0.33799	
1.91	6.12184	1.92360	0.33084									
1.92	6.18835	1.94241	0.33135									
1.93	6.25536	1.96122	0.33186									
1.94	6.32287	1.98003	0.33237									
1.95	6.39088	1.99884	0.33288									
1.96	6.45939	2.01765	0.33339									
1.97	6.52840	2.03646	0.33390									
1.98	6.59791	2.05527	0.33441									
1.99	6.66792	2.07408	0.33492									
2.00	6.73843	2.09289	0.33543									

TABLE 4B. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 1/4$ and x from 1.50 to 10.0.

x	$F_{3/4}(x)$	$H_{1/4}(x)$	$T_{3/4}(x)$	x	$F_{3/4}(x)$	$H_{1/4}(x)$	$T_{3/4}(x)$	x	$F_{3/4}(x)$	$H_{1/4}(x)$	$T_{3/4}(x)$
0.01	1.00000	0.02884	0.02885	0.50	1.08483	2.10100	1.93707	1.00	1.35789	3.42636	2.52326
0.02	1.00003	0.03998	0.04003	0.51	1.08829	2.12600	1.95391	1.01	1.37440	3.48621	2.53931
0.03	1.00005	0.05164	0.05167	0.52	1.09524	2.17636	1.97021	1.02	1.39113	3.54853	2.55591
0.04	1.00005	0.55558	0.55561	0.53	1.10524	2.20166	1.98688	1.04	1.38832	3.46695	2.55302
0.05	1.00033	0.63277	0.63277	0.54	1.13030	2.22657	2.01869	1.05	1.39443	3.57761	2.56015
0.06	1.00113	0.63249	0.63249	0.55	1.16690	2.25159	2.03423	1.06	1.40664	3.60848	2.56716
0.07	1.00113	0.74784	0.74784	0.56	1.11084	2.27670	2.04953	1.07	1.41395	3.63957	2.57405
0.08	1.00213	0.79932	0.79932	0.57	1.11486	2.30172	2.06459	1.08	1.42336	3.67087	2.58083
0.09	1.00270	0.84761	0.84761	0.58	1.11895	2.32676	2.07942	1.09	1.43368	3.70240	2.58750
0.10	1.00334	0.89232	0.89232	0.59	1.12312	2.35182	2.09401	1.10	1.43350	3.73415	2.59407
0.11	1.00404	0.93957	0.93957	0.60	1.12737	2.37690	2.10831	1.11	1.45705	3.76613	2.60067
0.12	1.00490	0.97172	0.97172	0.61	1.13169	2.40202	2.12230	1.12	1.45599	3.79843	2.60727
0.13	1.00594	1.00323	1.00323	0.62	1.13608	2.42717	2.13611	1.13	1.47703	3.83079	2.61381
0.14	1.00694	1.03455	1.03455	0.63	1.15148	2.45231	2.15011	1.14	1.48414	3.86347	2.62031
0.15	1.00751	1.09218	1.09218	0.64	1.15174	2.47761	2.16353	1.15	1.49640	3.89640	2.62678
0.16	1.00855	1.16173	1.16173	0.65	1.14550	2.50291	2.17693	1.16	1.50844	3.92958	2.63322
0.17	1.01003	1.19886	1.19886	0.66	1.14430	2.52821	2.19033	1.17	1.52081	3.96300	2.63968
0.18	1.01083	1.22779	1.22779	0.67	1.14418	2.55369	2.20373	1.18	1.53136	3.99668	2.64614
0.19	1.01206	1.22700	1.22700	0.68	1.14418	2.57917	2.21713	1.19	1.54158	4.03042	2.65261
0.20	1.01377	1.25231	1.25231	0.69	1.14914	2.60474	2.23053	1.20	1.55132	4.06420	2.65909
0.21	1.01475	1.28861	1.28861	0.70	1.14918	2.63037	2.24393	1.21	1.56132	4.09799	2.66557
0.22	1.01597	1.32491	1.32491	0.71	1.14918	2.65609	2.25733	1.22	1.57152	4.13177	2.67205
0.23	1.01747	1.37514	1.37514	0.72	1.15979	2.68180	2.27073	1.23	1.58179	4.16556	2.67853
0.24	1.01928	1.43195	1.43195	0.73	1.15916	2.70751	2.28413	1.24	1.59188	4.19935	2.68501
0.25	1.02093	1.49479	1.49479	0.74	1.15916	2.73379	2.29753	1.25	1.60198	4.23314	2.69149
0.26	1.02243	1.55565	1.55565	0.75	1.15916	2.75969	2.31093	1.26	1.61173	4.26693	2.69797
0.27	1.02443	1.61330	1.61330	0.76	1.16178	2.78609	2.32433	1.27	1.62155	4.30072	2.70445
0.28	1.02688	1.66641	1.66641	0.77	1.16178	2.81239	2.33773	1.28	1.63136	4.33451	2.71093
0.29	1.02880	1.71178	1.71178	0.78	1.16178	2.83881	2.35113	1.29	1.64117	4.36830	2.71741
0.30	1.03019	1.75099	1.75099	0.79	1.16178	2.86534	2.36453	1.30	1.65099	4.40209	2.72389
0.31	1.03223	1.79441	1.79441	0.80	1.16178	2.89200	2.37793	1.31	1.66082	4.43588	2.73037
0.32	1.03488	1.84241	1.84241	0.81	1.16178	2.91878	2.39133	1.32	1.67065	4.46967	2.73685
0.33	1.03793	1.89450	1.89450	0.82	1.16178	2.94557	2.40473	1.33	1.68048	4.50346	2.74333
0.34	1.04119	1.95061	1.95061	0.83	1.16178	2.97272	2.41813	1.34	1.69031	4.53725	2.74981
0.35	1.04460	1.66766	1.66766	0.84	1.16178	2.99990	2.43153	1.35	1.70014	4.57104	2.75629
0.36	1.04808	1.68661	1.68661	0.85	1.16178	3.02721	2.44493	1.36	1.71001	4.60483	2.76277
0.37	1.05163	1.69005	1.69005	0.86	1.16178	3.05461	2.45833	1.37	1.72001	4.63862	2.76925
0.38	1.05525	1.71110	1.71110	0.87	1.16178	3.08228	2.47173	1.38	1.73001	4.67241	2.77573
0.39	1.05893	1.73178	1.73178	0.88	1.16178	3.11003	2.48513	1.39	1.74001	4.70620	2.78221
0.40	1.06261	1.75209	1.75209	0.89	1.16178	3.13795	2.49853	1.40	1.75001	4.74000	2.78869
0.41	1.06634	1.77204	1.77204	0.90	1.16178	3.16601	2.51193	1.41	1.76001	4.77379	2.79517
0.42	1.07008	1.79199	1.79199	0.91	1.16178	3.19425	2.52533	1.42	1.77001	4.80758	2.80165
0.43	1.07387	1.81194	1.81194	0.92	1.16178	3.22250	2.53873	1.43	1.78001	4.84137	2.80813
0.44	1.07761	1.83189	1.83189	0.93	1.16178	3.25075	2.55213	1.44	1.79001	4.87516	2.81461
0.45	1.08134	1.85184	1.85184	0.94	1.16178	3.27900	2.56553	1.45	1.80001	4.90895	2.82109
0.46	1.08508	1.87179	1.87179	0.95	1.16178	3.30725	2.57893	1.46	1.81001	4.94274	2.82757
0.47	1.08881	1.89174	1.89174	0.96	1.16178	3.33550	2.59233	1.47	1.82001	4.97653	2.83405
0.48	1.09255	1.91169	1.91169	0.97	1.16178	3.36375	2.60573	1.48	1.83001	5.01032	2.84053
0.49	1.09628	1.93164	1.93164	0.98	1.16178	3.39200	2.61913	1.49	1.84001	5.04411	2.84701
0.50	1.10000	1.95159	1.95159	0.99	1.16178	3.42025	2.63253	1.50	1.85001	5.07790	2.85349
0.51	1.10373	1.97154	1.97154	1.00	1.16178	3.44850	2.64593				
0.52	1.10746	1.99149	1.99149								
0.53	1.11119	2.01144	2.01144								
0.54	1.11492	2.03139	2.03139								
0.55	1.11865	2.05134	2.05134								
0.56	1.12238	2.07129	2.07129								
0.57	1.12611	2.09124	2.09124								
0.58	1.12984	2.11119	2.11119								
0.59	1.13357	2.13114	2.13114								
0.60	1.13730	2.15109	2.15109								
0.61	1.14103	2.17104	2.17104								
0.62	1.14476	2.19099	2.19099								
0.63	1.14849	2.21094	2.21094								
0.64	1.15222	2.23089	2.23089								
0.65	1.15595	2.25084	2.25084								
0.66	1.15968	2.27079	2.27079								
0.67	1.16341	2.29074	2.29074								
0.68	1.16714	2.31069	2.31069								
0.69	1.17087	2.33064	2.33064								
0.70	1.17460	2.35059	2.35059								
0.71	1.17833	2.37054	2.37054								
0.72	1.18206	2.39049	2.39049								
0.73	1.18579	2.41044	2.41044								
0.74	1.18952	2.43039	2.43039								
0.75	1.19325	2.45034	2.45034								
0.76	1.19698	2.47029	2.47029								
0.77	1.20071	2.49024	2.49024								
0.78	1.20444	2.51019	2.51019								
0.79	1.20817	2.53014	2.53014								
0.80	1.21190	2.55009	2.55009								
0.81	1.21563	2.57004	2.57004								
0.82	1.21936	2.59000	2.59000								
0.83	1.22309	2.61000	2.61000								
0.84	1.22682	2.63000	2.63000								
0.85	1.23055	2.65000	2.65000								
0.86	1.23428	2.67000	2.67000								
0.87	1.23801	2.69000	2.69000								
0.88	1.24174	2.71000	2.71000								
0.89	1.24547	2.73000	2.73000								
0.90	1.24920	2.75000	2.75000								
0.91	1.25293	2.77000	2.77000								
0.92	1.25666	2.79000	2.79000								
0.93	1.26039	2.81000	2.81000								
0.94	1.26412	2.83000	2.83000								
0.95	1.26785	2.85000	2.85000								
0.96	1.27158	2.87000	2.87000								
0.97	1.27531	2.89000	2.89000								
0.98	1.27904	2.91000	2.91000								
0.99	1.28277	2.93000	2.93000								
1.00	1.28650	2.95000	2.95000								

TABLE 5A. Lanchester-Clifford-Schl fli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 3/4$ and x from 0.00 to 1.50.

$\alpha = 3/4$

x	$F_{3/4}(x)$	$H_{1/4}(x)$	$T_{3/4}(x)$	x	$F_{3/4}(x)$	$H_{1/4}(x)$	$T_{3/4}(x)$	x	$F_{3/4}(x)$	$H_{1/4}(x)$	$T_{3/4}(x)$
1.50	1.87907	5.22942	2.78299	2.0	2.76367	7.98850	2.89054	6.0	107.80972	318.97126	2.95865
1.51	1.89270	5.23750	2.78623	2.1	2.99970	8.70854	2.90347	6.1	118.61917	350.95316	2.95865
1.52	1.90648	5.24558	2.78941	2.2	3.23984	9.42858	2.91635	6.2	129.53525	386.17215	2.95866
1.53	1.92040	5.25366	2.79254	2.3	3.48000	10.14862	2.92927	6.3	140.65132	427.97712	2.95866
1.54	1.93446	5.26174	2.79561	2.4	3.72016	10.86866	2.94212	6.4	151.96900	467.67303	2.95866
1.55	1.94868	5.26982	2.79863	2.5	4.0015	11.58870	2.95497	6.5	173.96977	514.71843	2.95867
1.56	1.96304	5.27790	2.80163	2.6	4.32124	12.30874	2.96782	6.6	191.48319	566.53511	2.95867
1.57	1.97755	5.28598	2.80463	2.7	4.68157	13.02878	2.98067	6.7	216.77365	623.60946	2.95867
1.58	1.99229	5.29406	2.80763	2.8	5.08389	13.74882	2.99352	6.8	243.02366	686.47762	2.95867
1.59	2.00704	5.30214	2.81063	2.9	5.52832	14.46886	2.99489	6.9	271.4908	755.77302	2.95867
1.60	2.02201	5.31022	2.81363	3.0	6.0143	15.18890	2.99626	7.0	281.21402	832.01977	2.95867
1.61	2.03713	5.31830	2.81663	3.1	6.53222	15.90894	2.99763	7.1	306.91993	916.06333	2.95867
1.62	2.05242	5.32638	2.81963	3.2	7.08222	16.62898	2.99900	7.2	330.97444	1008.65430	2.95867
1.63	2.06786	5.33446	2.82263	3.3	7.66381	17.34892	2.99937	7.3	354.97944	1100.65430	2.95867
1.64	2.08346	5.34254	2.82563	3.4	8.28540	18.06896	2.99974	7.4	378.98444	1223.05550	2.95867
1.65	2.09922	5.35062	2.82863	3.5	8.94799	18.78890	2.99974	7.5	402.98944	1346.88978	2.95867
1.66	2.11524	5.35870	2.83163	3.6	9.65158	19.50894	2.99974	7.6	426.99444	1483.33610	2.95867
1.67	2.13147	5.36678	2.83463	3.7	10.39517	20.22898	2.99974	7.7	450.99944	1633.68608	2.95867
1.68	2.14791	5.37486	2.83763	3.8	11.17876	20.94892	2.99974	7.8	474.99944	1799.36002	2.95867
1.69	2.16447	5.38294	2.84063	3.9	12.00235	21.66896	2.99974	7.9	498.99944	1981.92656	2.95867
1.70	2.18124	5.39102	2.84363	4.0	12.86594	22.38890	2.99974	8.0	522.99944	2183.11444	2.95867
1.71	2.19822	5.39910	2.84663	4.1	13.76953	23.10894	2.99974	8.1	546.99944	2403.93066	2.95867
1.72	2.21547	5.40718	2.84963	4.2	14.71312	23.82898	2.99974	8.2	570.99944	2644.74688	2.95867
1.73	2.23299	5.41526	2.85263	4.3	15.69671	24.54892	2.99974	8.3	594.99944	2905.56310	2.95867
1.74	2.25074	5.42334	2.85563	4.4	16.72030	25.26896	2.99974	8.4	618.99944	3186.37932	2.95867
1.75	2.26868	5.43142	2.85863	4.5	17.78389	25.98890	2.99974	8.5	642.99944	3487.19554	2.95867
1.76	2.28683	5.43950	2.86163	4.6	18.88748	26.70894	2.99974	8.6	666.99944	3808.01176	2.95867
1.77	2.30518	5.44758	2.86463	4.7	20.03107	27.42898	2.99974	8.7	690.99944	4148.82798	2.95867
1.78	2.32374	5.45566	2.86763	4.8	21.21466	28.14892	2.99974	8.8	714.99944	4509.64420	2.95867
1.79	2.34249	5.46374	2.87063	4.9	22.43825	28.86896	2.99974	8.9	738.99944	4890.46042	2.95867
1.80	2.36143	5.47182	2.87363	5.0	23.70184	29.58890	2.99974	9.0	762.99944	5291.27664	2.95867
1.81	2.38058	5.47990	2.87663	5.1	25.00543	30.30894	2.99974	9.1	786.99944	5712.09286	2.95867
1.82	2.39993	5.48798	2.87963	5.2	26.34902	31.02898	2.99974	9.2	810.99944	6152.90908	2.95867
1.83	2.41947	5.49606	2.88263	5.3	27.73261	31.74892	2.99974	9.3	834.99944	6613.72530	2.95867
1.84	2.43922	5.50414	2.88563	5.4	29.15620	32.46896	2.99974	9.4	858.99944	7094.54152	2.95867
1.85	2.45917	5.51222	2.88863	5.5	30.62079	33.18890	2.99974	9.5	882.99944	7595.35774	2.95867
1.86	2.47932	5.52030	2.89163	5.6	32.12538	33.90894	2.99974	9.6	906.99944	8116.17396	2.95867
1.87	2.49967	5.52838	2.89463	5.7	33.67097	34.62898	2.99974	9.7	930.99944	8656.99018	2.95867
1.88	2.52032	5.53646	2.89763	5.8	35.25656	35.34892	2.99974	9.8	954.99944	9217.80640	2.95867
1.89	2.54127	5.54454	2.90063	5.9	36.88215	36.06896	2.99974	9.9	978.99944	9798.62262	2.95867
1.90	2.56252	5.55262	2.90363	6.0	38.54774	36.78890	2.99974	10.0	1002.99944	10399.43884	2.95867
1.91	2.58407	5.56070	2.90663								
1.92	2.60592	5.56878	2.90963								
1.93	2.62807	5.57686	2.91263								
1.94	2.65042	5.58494	2.91563								
1.95	2.67307	5.59302	2.91863								
1.96	2.69592	5.60110	2.92163								
1.97	2.71907	5.60918	2.92463								
1.98	2.74242	5.61726	2.92763								
1.99	2.76607	5.62534	2.93063								
2.00	2.79002	5.63342	2.93363								

TABLE 5B. Lanchester-Clifford-Schl fli Functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and $T_\alpha(x)$ for $\alpha = 3/4$ and x from 1.50 to 10.0.

x	$F_{1/5}(x)$	$H_{4/5}(x)$	$T_{1/5}(x)$	x	$F_{1/5}(x)$	$H_{4/5}(x)$	$T_{1/5}(x)$	x	$F_{1/5}(x)$	$H_{4/5}(x)$	$T_{1/5}(x)$
0.0	1.00000	0.00026	0.00026	0.50	1.32072	0.14080	0.10661	1.00	2.38524	0.47223	0.19798
0.01	1.00013	0.00029	0.00029	0.51	1.32422	0.14093	0.10690	1.01	2.38596	0.48109	0.19913
0.02	1.00026	0.00031	0.00031	0.52	1.32762	0.14106	0.10715	1.02	2.38668	0.48987	0.20025
0.03	1.00039	0.00033	0.00033	0.53	1.33102	0.14119	0.10740	1.03	2.38740	0.49865	0.20136
0.04	1.00050	0.00035	0.00035	0.54	1.33442	0.14132	0.10765	1.04	2.38812	0.50743	0.20247
0.05	1.00063	0.00037	0.00037	0.55	1.33782	0.14145	0.10790	1.05	2.38884	0.51621	0.20358
0.06	1.00075	0.00039	0.00039	0.56	1.34122	0.14158	0.10815	1.06	2.38956	0.52499	0.20469
0.07	1.00088	0.00041	0.00041	0.57	1.34462	0.14171	0.10840	1.07	2.39028	0.53377	0.20580
0.08	1.00100	0.00043	0.00043	0.58	1.34802	0.14184	0.10865	1.08	2.39100	0.54255	0.20691
0.09	1.00113	0.00045	0.00045	0.59	1.35142	0.14197	0.10890	1.09	2.39172	0.55133	0.20802
0.10	1.00125	0.00047	0.00047	0.60	1.35482	0.14210	0.10915	1.10	2.39244	0.56011	0.20913
0.11	1.00138	0.00049	0.00049	0.61	1.35822	0.14223	0.10940	1.11	2.39316	0.56889	0.21024
0.12	1.00150	0.00051	0.00051	0.62	1.36162	0.14236	0.10965	1.12	2.39388	0.57767	0.21135
0.13	1.00163	0.00053	0.00053	0.63	1.36502	0.14249	0.10990	1.13	2.39460	0.58645	0.21246
0.14	1.00175	0.00055	0.00055	0.64	1.36842	0.14262	0.11015	1.14	2.39532	0.59523	0.21357
0.15	1.00188	0.00057	0.00057	0.65	1.37182	0.14275	0.11040	1.15	2.39604	0.60401	0.21468
0.16	1.00200	0.00059	0.00059	0.66	1.37522	0.14288	0.11065	1.16	2.39676	0.61279	0.21579
0.17	1.00213	0.00061	0.00061	0.67	1.37862	0.14301	0.11090	1.17	2.39748	0.62157	0.21690
0.18	1.00225	0.00063	0.00063	0.68	1.38202	0.14314	0.11115	1.18	2.39820	0.63035	0.21801
0.19	1.00238	0.00065	0.00065	0.69	1.38542	0.14327	0.11140	1.19	2.39892	0.63913	0.21912
0.20	1.00250	0.00067	0.00067	0.70	1.38882	0.14340	0.11165	1.20	2.39964	0.64791	0.22023
0.21	1.00263	0.00069	0.00069	0.71	1.39222	0.14353	0.11190	1.21	2.40036	0.65669	0.22134
0.22	1.00275	0.00071	0.00071	0.72	1.39562	0.14366	0.11215	1.22	2.40108	0.66547	0.22245
0.23	1.00288	0.00073	0.00073	0.73	1.39902	0.14379	0.11240	1.23	2.40180	0.67425	0.22356
0.24	1.00300	0.00075	0.00075	0.74	1.40242	0.14392	0.11265	1.24	2.40252	0.68303	0.22467
0.25	1.00313	0.00077	0.00077	0.75	1.40582	0.14405	0.11290	1.25	2.40324	0.69181	0.22578
0.26	1.00325	0.00079	0.00079	0.76	1.40922	0.14418	0.11315	1.26	2.40396	0.70059	0.22689
0.27	1.00338	0.00081	0.00081	0.77	1.41262	0.14431	0.11340	1.27	2.40468	0.70937	0.22800
0.28	1.00350	0.00083	0.00083	0.78	1.41602	0.14444	0.11365	1.28	2.40540	0.71815	0.22911
0.29	1.00363	0.00085	0.00085	0.79	1.41942	0.14457	0.11390	1.29	2.40612	0.72693	0.23022
0.30	1.00375	0.00087	0.00087	0.80	1.42282	0.14470	0.11415	1.30	2.40684	0.73571	0.23133
0.31	1.00388	0.00089	0.00089	0.81	1.42622	0.14483	0.11440	1.31	2.40756	0.74449	0.23244
0.32	1.00400	0.00091	0.00091	0.82	1.42962	0.14496	0.11465	1.32	2.40828	0.75327	0.23355
0.33	1.00413	0.00093	0.00093	0.83	1.43302	0.14509	0.11490	1.33	2.40900	0.76205	0.23466
0.34	1.00425	0.00095	0.00095	0.84	1.43642	0.14522	0.11515	1.34	2.40972	0.77083	0.23577
0.35	1.00438	0.00097	0.00097	0.85	1.43982	0.14535	0.11540	1.35	2.41044	0.77961	0.23688
0.36	1.00450	0.00099	0.00099	0.86	1.44322	0.14548	0.11565	1.36	2.41116	0.78839	0.23799
0.37	1.00463	0.00101	0.00101	0.87	1.44662	0.14561	0.11590	1.37	2.41188	0.79717	0.23910
0.38	1.00475	0.00103	0.00103	0.88	1.45002	0.14574	0.11615	1.38	2.41260	0.80595	0.24021
0.39	1.00488	0.00105	0.00105	0.89	1.45342	0.14587	0.11640	1.39	2.41332	0.81473	0.24132
0.40	1.00500	0.00107	0.00107	0.90	1.45682	0.14600	0.11665	1.40	2.41404	0.82351	0.24243
0.41	1.00513	0.00109	0.00109	0.91	1.46022	0.14613	0.11690	1.41	2.41476	0.83229	0.24354
0.42	1.00525	0.00111	0.00111	0.92	1.46362	0.14626	0.11715	1.42	2.41548	0.84107	0.24465
0.43	1.00538	0.00113	0.00113	0.93	1.46702	0.14639	0.11740	1.43	2.41620	0.84985	0.24576
0.44	1.00550	0.00115	0.00115	0.94	1.47042	0.14652	0.11765	1.44	2.41692	0.85863	0.24687
0.45	1.00563	0.00117	0.00117	0.95	1.47382	0.14665	0.11790	1.45	2.41764	0.86741	0.24798
0.46	1.00575	0.00119	0.00119	0.96	1.47722	0.14678	0.11815	1.46	2.41836	0.87619	0.24909
0.47	1.00588	0.00121	0.00121	0.97	1.48062	0.14691	0.11840	1.47	2.41908	0.88497	0.25020
0.48	1.00600	0.00123	0.00123	0.98	1.48402	0.14704	0.11865	1.48	2.41980	0.89375	0.25131
0.49	1.00613	0.00125	0.00125	0.99	1.48742	0.14717	0.11890	1.49	2.42052	0.90253	0.25242
0.50	1.00625	0.00127	0.00127	1.00	1.49082	0.14730	0.11915	1.50	2.42124	0.91131	0.25353

TABLE 6A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 1/5$ and x from 0.00 to 1.50.

$$\alpha = 1/5$$

x	$F_{1/5}(x)$	$H_{4/5}(x)$	$T_{1/5}(x)$	x	$F_{1/5}(x)$	$H_{4/5}(x)$	$T_{1/5}(x)$	x	$F_{1/5}(x)$	$H_{4/5}(x)$	$T_{1/5}(x)$
1.50	4.53040	1.06142	0.23429	2.0	8.2486	2.07992	0.24888	6.0	700.89071	177.74325	0.25360
1.51	4.58820	1.07682	0.23469	2.1	9.10623	2.35595	0.24315	6.1	719.92262	179.25427	0.25360
1.52	4.64669	1.09240	0.23509	2.2	10.01549	2.67072	0.23717	6.2	739.87762	180.76529	0.25360
1.53	4.70578	1.10914	0.23548	2.3	10.96961	3.01928	0.23041	6.3	760.77071	182.27631	0.25360
1.54	4.76578	1.12406	0.23586	2.4	11.96620	3.40326	0.22368	6.4	782.69667	183.78733	0.25360
1.55	4.82639	1.14015	0.23623	2.5	12.99620	3.82303	0.21695	6.5	805.65516	185.29835	0.25360
1.56	4.88771	1.15642	0.23659	2.6	14.05961	4.27886	0.21022	6.6	829.64732	186.80937	0.25360
1.57	4.94976	1.17298	0.23695	2.7	15.15642	4.76992	0.20349	6.7	854.67326	188.32039	0.25360
1.58	5.01258	1.18949	0.23732	2.8	16.28661	5.29628	0.19676	6.8	880.73420	189.83141	0.25360
1.59	5.07607	1.20630	0.23764	2.9	17.44987	5.85795	0.19003	6.9	907.83014	191.34243	0.25360
1.60	5.14035	1.22330	0.23798	3.0	18.64561	6.45503	0.18330	7.0	935.96108	192.85345	0.25360
1.61	5.20538	1.24048	0.23831	3.1	19.87342	7.08751	0.17657	7.1	965.12702	194.36447	0.25360
1.62	5.27117	1.25785	0.23863	3.2	21.13361	7.75548	0.16984	7.2	995.32896	195.87549	0.25360
1.63	5.33774	1.27541	0.23894	3.3	22.42661	8.45895	0.16311	7.3	1026.56690	197.38651	0.25360
1.64	5.40509	1.29316	0.23925	3.4	23.75280	9.19792	0.15638	7.4	1058.84084	198.89753	0.25360
1.65	5.47322	1.31111	0.23955	3.5	25.11240	9.97240	0.14965	7.5	1092.15078	200.40855	0.25360
1.66	5.54166	1.32925	0.23985	3.6	26.50500	10.78248	0.14292	7.6	1126.49672	201.91957	0.25360
1.67	5.61089	1.34759	0.24013	3.7	27.93000	11.62806	0.13619	7.7	1161.87966	203.43059	0.25360
1.68	5.68042	1.36614	0.24041	3.8	29.38661	12.50924	0.12946	7.8	1198.29960	204.94161	0.25360
1.69	5.75062	1.38488	0.24069	3.9	30.87440	13.42602	0.12273	7.9	1235.75654	206.45263	0.25360
1.70	5.82202	1.40383	0.24096	4.0	32.39300	14.37840	0.11600	8.0	1274.25048	207.96365	0.25360
1.71	5.89506	1.42299	0.24122	4.1	33.94240	15.36658	0.10927	8.1	1313.78142	209.47467	0.25360
1.72	5.97095	1.44230	0.24148	4.2	35.52320	16.39076	0.10254	8.2	1354.34936	210.98569	0.25360
1.73	6.04870	1.46174	0.24174	4.3	37.13500	17.45094	0.09581	8.3	1395.95430	212.49671	0.25360
1.74	6.12831	1.48131	0.24198	4.4	38.77760	18.54722	0.08908	8.4	1438.59624	214.00773	0.25360
1.75	6.19779	1.50175	0.24223	4.5	40.45160	19.67960	0.08235	8.5	1482.27518	215.51875	0.25360
1.76	6.27116	1.52195	0.24246	4.6	42.15640	20.84808	0.07562	8.6	1526.99112	217.02977	0.25360
1.77	6.33542	1.54234	0.24270	4.7	43.89240	22.05246	0.06889	8.7	1572.74406	218.54079	0.25360
1.78	6.40042	1.56291	0.24292	4.8	45.65920	23.29284	0.06216	8.8	1619.53300	220.05181	0.25360
1.79	6.45467	1.58402	0.24315	4.9	47.45720	24.56922	0.05543	8.9	1667.35794	221.56283	0.25360
1.80	6.50866	1.60516	0.24337	5.0	49.28680	25.88160	0.04870	9.0	1716.21888	223.07385	0.25360
1.81	6.57159	1.62652	0.24358	5.1	51.14840	27.23008	0.04197	9.1	1766.11582	224.58487	0.25360
1.82	6.63429	1.64812	0.24379	5.2	53.04160	28.61446	0.03524	9.2	1817.04876	226.09589	0.25360
1.83	6.69629	1.66995	0.24399	5.3	54.96680	30.03484	0.02851	9.3	1869.01770	227.60691	0.25360
1.84	6.75797	1.69203	0.24419	5.4	56.92440	31.49122	0.02178	9.4	1922.02264	229.11793	0.25360
1.85	7.01482	1.71434	0.24439	5.5	58.91400	32.98360	0.01505	9.5	1976.06358	230.62895	0.25360
1.86	7.18255	1.73690	0.24458	5.6	60.93600	34.51208	0.00832	9.6	2031.14052	232.14000	0.25360
1.87	7.18255	1.75971	0.24477	5.7	62.99000	36.07646	0.00159	9.7	2087.25346	233.65102	0.25360
1.88	7.22801	1.78276	0.24495	5.8	65.07560	37.67684	0.00000	9.8	2144.40240	235.16204	0.25360
1.89	7.36775	1.80607	0.24513	5.9	67.19240	39.31322	0.00000	9.9	2202.68734	236.67306	0.25360
1.90	7.45951	1.82963	0.24531	6.0	700.89071	177.74325	0.25360	10.0	45288.16157	11484.99634	0.25360
1.91	7.55031	1.85346	0.24548								
1.92	7.64315	1.87754	0.24565								
1.93	7.73805	1.90188	0.24581								
1.94	7.83501	1.92649	0.24598								
1.95	7.92806	1.95137	0.24614								
1.96	8.02119	1.97653	0.24629								
1.97	8.11433	2.00195	0.24644								
1.98	8.20747	2.02766	0.24659								
1.99	8.30235	2.05365	0.24674								
2.00	8.42486	2.07992	0.24688								

TABLE 6B. Lanchester-Clifford-Schl\"afli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 1/5$ and x from 1.50 to 10.0.

x	$F_{2/5}(x)$	$H_{3/5}(x)$	$T_{2/5}(x)$	x	$F_{2/5}(x)$	$H_{3/5}(x)$	$T_{2/5}(x)$	x	$F_{2/5}(x)$	$H_{3/5}(x)$	$T_{2/5}(x)$
0.0	1.00000	0.0	0.0	0.50	1.15977	0.38266	0.23304	1.00	1.62778	0.84339	0.50173
0.01	1.00006	0.00029	0.00289	0.51	1.16337	0.38666	0.23865	1.01	1.63773	0.85707	0.50893
0.02	1.00025	0.00064	0.00643	0.52	1.17312	0.39166	0.24422	1.02	1.64896	0.87096	0.51613
0.03	1.00056	0.00109	0.01093	0.53	1.18001	0.39371	0.24725	1.03	1.65310	0.88358	0.52330
0.04	1.00100	0.00153	0.01533	0.54	1.18704	0.39733	0.25254	1.04	1.65978	0.89718	0.53047
0.05	1.00156	0.00203	0.02030	0.55	1.19222	0.40102	0.31069	1.05	1.66954	0.90991	0.51656
0.06	1.00236	0.00266	0.02666	0.56	1.20062	0.40470	0.31710	1.06	1.67551	0.92116	0.52118
0.07	1.00336	0.00342	0.03422	0.57	1.20644	0.40833	0.32371	1.07	1.67768	0.93153	0.52435
0.08	1.00460	0.00430	0.04306	0.58	1.21241	0.41199	0.32974	1.08	1.68006	0.94001	0.52756
0.09	1.00607	0.00530	0.05306	0.59	1.21841	0.41566	0.33596	1.09	1.84466	0.94761	0.53020
0.10	1.00777	0.00644	0.06444	0.60	1.22322	0.41933	0.33716	1.10	1.84466	0.97334	0.53020
0.11	1.00971	0.00770	0.07700	0.61	1.22409	0.42300	0.34231	1.11	1.84466	0.97334	0.53020
0.12	1.01188	0.00909	0.09090	0.62	1.22481	0.42666	0.34746	1.12	1.84466	1.00416	0.53020
0.13	1.01427	0.01061	0.10610	0.63	1.22558	0.43033	0.35261	1.13	1.84466	1.03496	0.53020
0.14	1.01688	0.01226	0.12260	0.64	1.22630	0.43400	0.35776	1.14	1.84466	1.06576	0.53020
0.15	1.01969	0.01404	0.14040	0.65	1.22702	0.43766	0.36291	1.15	1.84466	1.09656	0.53020
0.16	1.02260	0.01594	0.15940	0.66	1.22774	0.44133	0.36806	1.16	1.84466	1.12736	0.53020
0.17	1.02561	0.01794	0.17940	0.67	1.22846	0.44500	0.37321	1.17	1.84466	1.15816	0.53020
0.18	1.02872	0.01994	0.19940	0.68	1.22918	0.44866	0.37836	1.18	1.84466	1.18896	0.53020
0.19	1.03193	0.02204	0.22040	0.69	1.22990	0.45233	0.38351	1.19	1.84466	1.21976	0.53020
0.20	1.03524	0.02424	0.24240	0.70	1.23062	0.45600	0.38866	1.20	1.84466	1.25056	0.53020
0.21	1.03865	0.02654	0.26540	0.71	1.23134	0.45966	0.39381	1.21	1.84466	1.28136	0.53020
0.22	1.04216	0.02894	0.28940	0.72	1.23206	0.46333	0.39896	1.22	1.84466	1.31216	0.53020
0.23	1.04577	0.03144	0.31440	0.73	1.23278	0.46700	0.40411	1.23	1.84466	1.34296	0.53020
0.24	1.04938	0.03404	0.34040	0.74	1.23350	0.47066	0.40926	1.24	1.84466	1.37376	0.53020
0.25	1.05309	0.03674	0.36740	0.75	1.23422	0.47433	0.41441	1.25	1.84466	1.40456	0.53020
0.26	1.05680	0.03954	0.39540	0.76	1.23494	0.47800	0.41956	1.26	1.84466	1.43536	0.53020
0.27	1.06051	0.04234	0.42340	0.77	1.23566	0.48166	0.42471	1.27	1.84466	1.46616	0.53020
0.28	1.06422	0.04514	0.45140	0.78	1.23638	0.48533	0.42986	1.28	1.84466	1.49696	0.53020
0.29	1.06793	0.04794	0.47940	0.79	1.23710	0.48900	0.43501	1.29	1.84466	1.52776	0.53020
0.30	1.07164	0.05074	0.50740	0.80	1.23782	0.49266	0.44016	1.30	1.84466	1.55856	0.53020
0.31	1.07535	0.05354	0.53540	0.81	1.23854	0.49633	0.44531	1.31	1.84466	1.58936	0.53020
0.32	1.07906	0.05634	0.56340	0.82	1.23926	0.50000	0.45046	1.32	1.84466	1.62016	0.53020
0.33	1.08277	0.05914	0.59140	0.83	1.24000	0.50366	0.45561	1.33	1.84466	1.65096	0.53020
0.34	1.08648	0.06194	0.61940	0.84	1.24072	0.50733	0.46076	1.34	1.84466	1.68176	0.53020
0.35	1.09019	0.06474	0.64740	0.85	1.24144	0.51100	0.46591	1.35	1.84466	1.71256	0.53020
0.36	1.09390	0.06754	0.67540	0.86	1.24216	0.51466	0.47106	1.36	1.84466	1.74336	0.53020
0.37	1.09761	0.07034	0.70340	0.87	1.24288	0.51833	0.47621	1.37	1.84466	1.77416	0.53020
0.38	1.10132	0.07314	0.73140	0.88	1.24360	0.52200	0.48136	1.38	1.84466	1.80496	0.53020
0.39	1.10503	0.07594	0.75940	0.89	1.24432	0.52566	0.48651	1.39	1.84466	1.83576	0.53020
0.40	1.10874	0.07874	0.78740	0.90	1.24504	0.52933	0.49166	1.40	1.84466	1.86656	0.53020
0.41	1.11245	0.08154	0.81540	0.91	1.24576	0.53300	0.49681	1.41	1.84466	1.89736	0.53020
0.42	1.11616	0.08434	0.84340	0.92	1.24648	0.53666	0.50196	1.42	1.84466	1.92816	0.53020
0.43	1.11987	0.08714	0.87140	0.93	1.24720	0.54033	0.50711	1.43	1.84466	1.95896	0.53020
0.44	1.12358	0.08994	0.89940	0.94	1.24792	0.54400	0.51226	1.44	1.84466	1.98976	0.53020
0.45	1.12729	0.09274	0.92740	0.95	1.24864	0.54766	0.51741	1.45	1.84466	2.02056	0.53020
0.46	1.13100	0.09554	0.95540	0.96	1.24936	0.55133	0.52256	1.46	1.84466	2.05136	0.53020
0.47	1.13471	0.09834	0.98340	0.97	1.25008	0.55500	0.52771	1.47	1.84466	2.08216	0.53020
0.48	1.13842	0.10114	0.10114	0.98	1.25080	0.55866	0.53286	1.48	1.84466	2.11296	0.53020
0.49	1.14213	0.10394	0.10394	0.99	1.25152	0.56233	0.53801	1.49	1.84466	2.14376	0.53020
0.50	1.14584	0.10674	0.10674	1.00	1.25224	0.56600	0.54316	1.50	1.84466	2.17456	0.53020

TABLE 7A. Lanchester-Clifford-Schlafli Functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and

$T_\alpha(x)$ for $\alpha = 2/5$ and x from 0.00 to 1.50.

$$\alpha = 2/5$$

x	F _{2/5} (x)	H _{3/5} (x)	T _{2/5} (x)	x	F _{2/5} (x)	T _{2/5} (x)	x	F _{2/5} (x)	H _{3/5} (x)	T _{2/5} (x)	x	F _{2/5} (x)	H _{3/5} (x)	T _{2/5} (x)
1.0	2.71176	1.64228	0.60561	2.0	4.52641	3.29225	6.0	278.92057	187.25297	0.67136	6.0	278.92057	187.25297	0.67136
1.1	2.73913	1.66234	0.60689	2.1	4.59218	3.29225	6.1	308.91621	207.52297	0.67136	6.1	308.91621	207.52297	0.67136
1.2	2.76681	1.68236	0.60813	2.2	4.65778	3.29225	6.2	338.91621	227.79297	0.67136	6.2	338.91621	227.79297	0.67136
1.3	2.79432	1.70239	0.60936	2.3	4.72331	3.29225	6.3	368.91621	248.06297	0.67136	6.3	368.91621	248.06297	0.67136
1.4	2.82175	1.72243	0.61060	2.4	4.78881	3.29225	6.4	398.91621	268.33297	0.67136	6.4	398.91621	268.33297	0.67136
1.5	2.84918	1.74247	0.61184	2.5	4.85431	3.29225	6.5	428.91621	288.60297	0.67136	6.5	428.91621	288.60297	0.67136
1.6	2.87661	1.76251	0.61308	2.6	4.91981	3.29225	6.6	458.91621	308.87297	0.67136	6.6	458.91621	308.87297	0.67136
1.7	2.90404	1.78255	0.61432	2.7	4.98531	3.29225	6.7	488.91621	329.14297	0.67136	6.7	488.91621	329.14297	0.67136
1.8	2.93147	1.80259	0.61556	2.8	5.05081	3.29225	6.8	518.91621	349.41297	0.67136	6.8	518.91621	349.41297	0.67136
1.9	2.95890	1.82263	0.61680	2.9	5.11631	3.29225	6.9	548.91621	369.68297	0.67136	6.9	548.91621	369.68297	0.67136
2.0	2.98633	1.84267	0.61804	3.0	5.18181	3.29225	7.0	578.91621	389.95297	0.67136	7.0	578.91621	389.95297	0.67136
2.1	3.01376	1.86271	0.61928	3.1	5.24731	3.29225	7.1	608.91621	410.22297	0.67136	7.1	608.91621	410.22297	0.67136
2.2	3.04119	1.88275	0.62052	3.2	5.31281	3.29225	7.2	638.91621	430.49297	0.67136	7.2	638.91621	430.49297	0.67136
2.3	3.06862	1.90279	0.62176	3.3	5.37831	3.29225	7.3	668.91621	450.76297	0.67136	7.3	668.91621	450.76297	0.67136
2.4	3.09605	1.92283	0.62300	3.4	5.44381	3.29225	7.4	698.91621	471.03297	0.67136	7.4	698.91621	471.03297	0.67136
2.5	3.12348	1.94287	0.62424	3.5	5.50931	3.29225	7.5	728.91621	491.30297	0.67136	7.5	728.91621	491.30297	0.67136
2.6	3.15091	1.96291	0.62548	3.6	5.57481	3.29225	7.6	758.91621	511.57297	0.67136	7.6	758.91621	511.57297	0.67136
2.7	3.17834	1.98295	0.62672	3.7	5.64031	3.29225	7.7	788.91621	531.84297	0.67136	7.7	788.91621	531.84297	0.67136
2.8	3.20577	2.00299	0.62796	3.8	5.70581	3.29225	7.8	818.91621	552.11297	0.67136	7.8	818.91621	552.11297	0.67136
2.9	3.23320	2.02303	0.62920	3.9	5.77131	3.29225	7.9	848.91621	572.38297	0.67136	7.9	848.91621	572.38297	0.67136
3.0	3.26063	2.04307	0.63044	4.0	5.83681	3.29225	8.0	878.91621	592.65297	0.67136	8.0	878.91621	592.65297	0.67136
3.1	3.28806	2.06311	0.63168	4.1	5.90231	3.29225	8.1	908.91621	612.92297	0.67136	8.1	908.91621	612.92297	0.67136
3.2	3.31549	2.08315	0.63292	4.2	5.96781	3.29225	8.2	938.91621	633.19297	0.67136	8.2	938.91621	633.19297	0.67136
3.3	3.34292	2.10319	0.63416	4.3	6.03331	3.29225	8.3	968.91621	653.46297	0.67136	8.3	968.91621	653.46297	0.67136
3.4	3.37035	2.12323	0.63540	4.4	6.09881	3.29225	8.4	998.91621	673.73297	0.67136	8.4	998.91621	673.73297	0.67136
3.5	3.39778	2.14327	0.63664	4.5	6.16431	3.29225	8.5	1028.91621	694.00297	0.67136	8.5	1028.91621	694.00297	0.67136
3.6	3.42521	2.16331	0.63788	4.6	6.22981	3.29225	8.6	1058.91621	714.27297	0.67136	8.6	1058.91621	714.27297	0.67136
3.7	3.45264	2.18335	0.63912	4.7	6.29531	3.29225	8.7	1088.91621	734.54297	0.67136	8.7	1088.91621	734.54297	0.67136
3.8	3.48007	2.20339	0.64036	4.8	6.36081	3.29225	8.8	1118.91621	754.81297	0.67136	8.8	1118.91621	754.81297	0.67136
3.9	3.50750	2.22343	0.64160	4.9	6.42631	3.29225	8.9	1148.91621	775.08297	0.67136	8.9	1148.91621	775.08297	0.67136
4.0	3.53493	2.24347	0.64284	5.0	6.49181	3.29225	9.0	1178.91621	795.35297	0.67136	9.0	1178.91621	795.35297	0.67136
4.1	3.56236	2.26351	0.64408	5.1	6.55731	3.29225	9.1	1208.91621	815.62297	0.67136	9.1	1208.91621	815.62297	0.67136
4.2	3.58979	2.28355	0.64532	5.2	6.62281	3.29225	9.2	1238.91621	835.89297	0.67136	9.2	1238.91621	835.89297	0.67136
4.3	3.61722	2.30359	0.64656	5.3	6.68831	3.29225	9.3	1268.91621	856.16297	0.67136	9.3	1268.91621	856.16297	0.67136
4.4	3.64465	2.32363	0.64780	5.4	6.75381	3.29225	9.4	1298.91621	876.43297	0.67136	9.4	1298.91621	876.43297	0.67136
4.5	3.67208	2.34367	0.64904	5.5	6.81931	3.29225	9.5	1328.91621	896.70297	0.67136	9.5	1328.91621	896.70297	0.67136
4.6	3.69951	2.36371	0.65028	5.6	6.88481	3.29225	9.6	1358.91621	916.97297	0.67136	9.6	1358.91621	916.97297	0.67136
4.7	3.72694	2.38375	0.65152	5.7	6.95031	3.29225	9.7	1388.91621	937.24297	0.67136	9.7	1388.91621	937.24297	0.67136
4.8	3.75437	2.40379	0.65276	5.8	7.01581	3.29225	9.8	1418.91621	957.51297	0.67136	9.8	1418.91621	957.51297	0.67136
4.9	3.78180	2.42383	0.65400	5.9	7.08131	3.29225	9.9	1448.91621	977.78297	0.67136	9.9	1448.91621	977.78297	0.67136
5.0	3.80923	2.44387	0.65524	6.0	7.14681	3.29225	10.0	1478.91621	998.05297	0.67136	10.0	1478.91621	998.05297	0.67136
5.1	3.83666	2.46391	0.65648											
5.2	3.86409	2.48395	0.65772											
5.3	3.89152	2.50399	0.65896											
5.4	3.91895	2.52403	0.66020											
5.5	3.94638	2.54407	0.66144											
5.6	3.97381	2.56411	0.66268											
5.7	4.00124	2.58415	0.66392											
5.8	4.02867	2.60419	0.66516											
5.9	4.05610	2.62423	0.66640											
6.0	4.08353	2.64427	0.66764											
6.1	4.11096	2.66431	0.66888											
6.2	4.13839	2.68435	0.67012											
6.3	4.16582	2.70439	0.67136											
6.4	4.19325	2.72443	0.67260											
6.5	4.22068	2.74447	0.67384											
6.6	4.24811	2.76451	0.67508											
6.7	4.27554	2.78455	0.67632											
6.8	4.30297	2.80459	0.67756											
6.9	4.33040	2.82463	0.67880											
7.0	4.35783	2.84467	0.68004											
7.1	4.38526	2.86471	0.68128											
7.2	4.41269	2.88475	0.68252											
7.3	4.44012	2.90479	0.68376											
7.4	4.46755	2.92483	0.68500											
7.5	4.49498	2.94487	0.68624											
7.6	4.52241	2.96491	0.68748											
7.7	4.54984	2.98495	0.68872											
7.8	4.57727	3.00499	0.68996											
7.9	4.60470	3.02503	0.69120											
8.0	4.63213	3.04507	0.69244											
8.1	4.65956	3.06511	0.69368											
8.2	4.68699	3.08515	0.69492											
8.3	4.71442	3.10519	0.69616											
8.4	4.74185	3.12523	0.69740											
8.5	4.76928	3.14527	0.69864											
8.6	4.79671	3.16531	0.69988											
8.7	4.82414	3.18535	0.70112											
8.8	4.85157	3.20539	0.70236											
8.9	4.87900	3.22543	0.70360											
9.0	4.90643	3.24547	0.70484											
9.1	4.93386	3.26551	0.70608											
9.2	4.96129	3.28555	0.70732											
9.3	4.98872	3.30559	0.70856											
9.4	5.01615	3.32563	0.70980											
9.5	5.04358	3.34567	0.71104											
9.6	5.07101	3.36571	0.71228											
9.7	5.09844	3.38575	0.71352											
9.8	5.12587	3.40579	0.71476											
9.9	5.15330	3.42583	0.71600											
10.0	5.18073	3.44587	0.71724											

TABLE 7B. Lanchester-Clifford-Schl fli Functions F_α(x), H_{1-α}(x), and

x	$F_{3/5}(x)$	$H_{2/5}(x)$	$T_{3/5}(x)$	x	$F_{3/5}(x)$	$H_{2/5}(x)$	$T_{3/5}(x)$	x	$F_{3/5}(x)$	$H_{2/5}(x)$	$T_{3/5}(x)$
0.0	1.00000	0.0	0.0	0.50	1.10222	0.89199	0.77922	1.00	1.45028	1.70597	1.17030
0.01	1.00004	0.00007	0.00007	0.51	1.10273	0.89251	0.78047	1.01	1.45094	1.70636	1.17072
0.02	1.00013	0.00024	0.00024	0.52	1.10324	0.89303	0.78100	1.02	1.45160	1.70675	1.17114
0.03	1.00029	0.00057	0.00057	0.53	1.10375	0.89354	0.78153	1.03	1.45226	1.70714	1.17156
0.04	1.00057	0.00109	0.00109	0.54	1.10426	0.89405	0.78206	1.04	1.45292	1.70753	1.17198
0.05	1.00090	0.00183	0.00183	0.55	1.10477	0.89456	0.78259	1.05	1.45358	1.70792	1.17240
0.06	1.00129	0.00283	0.00283	0.56	1.10528	0.89507	0.78312	1.06	1.45424	1.70831	1.17282
0.07	1.00175	0.00413	0.00413	0.57	1.10579	0.89558	0.78365	1.07	1.45490	1.70870	1.17324
0.08	1.00227	0.00577	0.00577	0.58	1.10630	0.89609	0.78418	1.08	1.45556	1.70909	1.17366
0.09	1.00287	0.00780	0.00780	0.59	1.10681	0.89660	0.78471	1.09	1.45622	1.70948	1.17408
0.10	1.00353	0.01027	0.01027	0.60	1.10732	0.89711	0.78524	1.10	1.45688	1.70987	1.17450
0.11	1.00427	0.01373	0.01373	0.61	1.10783	0.89762	0.78577	1.11	1.45754	1.71026	1.17492
0.12	1.00507	0.01823	0.01823	0.62	1.10834	0.89813	0.78630	1.12	1.45820	1.71065	1.17534
0.13	1.00593	0.02383	0.02383	0.63	1.10885	0.89864	0.78683	1.13	1.45886	1.71104	1.17576
0.14	1.00685	0.03060	0.03060	0.64	1.10936	0.89915	0.78736	1.14	1.45952	1.71143	1.17618
0.15	1.00793	0.03869	0.03869	0.65	1.10987	0.89966	0.78789	1.15	1.46018	1.71182	1.17660
0.16	1.00917	0.04825	0.04825	0.66	1.11038	0.90017	0.78842	1.16	1.46084	1.71221	1.17702
0.17	1.01057	0.05953	0.05953	0.67	1.11089	0.90068	0.78895	1.17	1.46150	1.71260	1.17744
0.18	1.01213	0.07271	0.07271	0.68	1.11140	0.90119	0.78948	1.18	1.46216	1.71299	1.17786
0.19	1.01385	0.08795	0.08795	0.69	1.11191	0.90170	0.78999	1.19	1.46282	1.71338	1.17828
0.20	1.01573	0.10541	0.10541	0.70	1.11242	0.90221	0.79052	1.20	1.46348	1.71377	1.17870
0.21	1.01777	0.12535	0.12535	0.71	1.11293	0.90272	0.79105	1.21	1.46414	1.71416	1.17912
0.22	1.01997	0.14793	0.14793	0.72	1.11344	0.90323	0.79158	1.22	1.46480	1.71455	1.17954
0.23	1.02233	0.17341	0.17341	0.73	1.11395	0.90374	0.79211	1.23	1.46546	1.71494	1.17996
0.24	1.02485	0.20205	0.20205	0.74	1.11446	0.90425	0.79264	1.24	1.46612	1.71533	1.18038
0.25	1.02753	0.23413	0.23413	0.75	1.11497	0.90476	0.79317	1.25	1.46678	1.71572	1.18080
0.26	1.03037	0.26993	0.26993	0.76	1.11548	0.90527	0.79370	1.26	1.46744	1.71611	1.18122
0.27	1.03337	0.30971	0.30971	0.77	1.11599	0.90578	0.79423	1.27	1.46810	1.71650	1.18164
0.28	1.03653	0.35383	0.35383	0.78	1.11650	0.90629	0.79476	1.28	1.46876	1.71689	1.18206
0.29	1.03985	0.40265	0.40265	0.79	1.11701	0.90680	0.79529	1.29	1.46942	1.71728	1.18248
0.30	1.04333	0.45653	0.45653	0.80	1.11752	0.90731	0.79582	1.30	1.47008	1.71767	1.18290
0.31	1.04697	0.51593	0.51593	0.81	1.11803	0.90782	0.79635	1.31	1.47074	1.71806	1.18332
0.32	1.05077	0.58121	0.58121	0.82	1.11854	0.90833	0.79688	1.32	1.47140	1.71845	1.18374
0.33	1.05473	0.65283	0.65283	0.83	1.11905	0.90884	0.79741	1.33	1.47206	1.71884	1.18416
0.34	1.05885	0.73125	0.73125	0.84	1.11956	0.90935	0.79794	1.34	1.47272	1.71923	1.18458
0.35	1.06313	0.81693	0.81693	0.85	1.12007	0.90986	0.79847	1.35	1.47338	1.71962	1.18500
0.36	1.06757	0.91043	0.91043	0.86	1.12058	0.91037	0.79900	1.36	1.47404	1.72001	1.18542
0.37	1.07217	1.01231	1.01231	0.87	1.12109	0.91088	0.79953	1.37	1.47470	1.72040	1.18584
0.38	1.07693	1.12305	1.12305	0.88	1.12160	0.91139	0.79999	1.38	1.47536	1.72079	1.18626
0.39	1.08185	1.24313	1.24313	0.89	1.12211	0.91190	0.80052	1.39	1.47602	1.72118	1.18668
0.40	1.08693	1.37305	1.37305	0.90	1.12262	0.91241	0.80105	1.40	1.47668	1.72157	1.18710
0.41	1.09217	1.51321	1.51321	0.91	1.12313	0.91292	0.80158	1.41	1.47734	1.72196	1.18752
0.42	1.09757	1.66403	1.66403	0.92	1.12364	0.91343	0.80211	1.42	1.47800	1.72235	1.18794
0.43	1.10313	1.82601	1.82601	0.93	1.12415	0.91394	0.80264	1.43	1.47866	1.72274	1.18836
0.44	1.10885	2.00065	2.00065	0.94	1.12466	0.91445	0.80317	1.44	1.47932	1.72313	1.18878
0.45	1.11473	2.18843	2.18843	0.95	1.12517	0.91496	0.80370	1.45	1.48000	1.72352	1.18920
0.46	1.12077	2.39085	2.39085	0.96	1.12568	0.91547	0.80423	1.46	1.48066	1.72391	1.18962
0.47	1.12697	2.60843	2.60843	0.97	1.12619	0.91598	0.80476	1.47	1.48132	1.72430	1.19004
0.48	1.13333	2.84165	2.84165	0.98	1.12670	0.91649	0.80529	1.48	1.48198	1.72469	1.19046
0.49	1.13985	3.09101	3.09101	0.99	1.12721	0.91700	0.80582	1.49	1.48264	1.72508	1.19088
0.50	1.14653	3.35699	3.35699	1.00	1.12772	0.91751	0.80635	1.50	1.48330	1.72547	1.19130

TABLE 8A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 3/5$ and x from 0.00 to 1.50.

$$\alpha = 3/5$$

x	$F_{3/5}(x)$	$H_{2/5}(x)$	$T_{3/5}(x)$	x	$F_{3/5}(x)$	$H_{2/5}(x)$	$T_{3/5}(x)$	x	$F_{3/5}(x)$	$H_{2/5}(x)$	$T_{3/5}(x)$
1.50	2.11465	2.88285	1.36327	2.0	3.25912	4.69636	1.44099	6.0	153.10773	28.09212	1.48949
1.51	2.12114	2.91169	1.36562	2.1	3.56725	5.19257	1.44951	6.1	186.97493	27.41383	1.48949
1.52	2.12770	2.94080	1.36792	2.2	3.90199	5.66611	1.45665	6.2	205.57172	26.74295	1.48949
1.53	2.13436	2.97015	1.37019	2.3	4.28053	6.06617	1.46249	6.3	220.57172	26.07198	1.48950
1.54	2.14112	2.99977	1.37241	2.4	4.68063	6.39726	1.46731	6.4	232.60917	25.40134	1.48950
1.55	2.14797	3.02955	1.37460	2.5	5.10066	7.59185	1.47121	6.5	250.24413	312.73492	1.48950
1.56	2.15490	3.05959	1.37675	2.6	5.54271	8.25659	1.47453	6.6	276.10208	411.25538	1.48950
1.57	2.16192	3.08982	1.37886	2.7	6.00689	9.28448	1.47723	6.7	304.64504	533.79071	1.48950
1.58	2.16903	3.12026	1.38093	2.8	6.49325	10.14858	1.47925	6.8	336.44699	500.60771	1.48950
1.59	2.17624	3.15083	1.38296	2.9	7.00255	11.14858	1.48123	6.9	369.91477	552.47842	1.48950
1.60	2.18355	3.18156	1.38496	3.0	7.53444	12.37860	1.48272	7.0	409.28911	692.47359	1.48950
1.61	2.19096	3.21243	1.38692	3.1	8.09823	13.14608	1.48394	7.1	451.24444	774.32760	1.48950
1.62	2.19847	3.24343	1.38885	3.2	8.69399	14.18417	1.48494	7.2	498.99513	845.22251	1.48950
1.63	2.20607	3.27453	1.39074	3.3	9.32177	15.05476	1.48577	7.3	549.99647	916.22251	1.48950
1.64	2.21376	3.30573	1.39260	3.4	1.01463	16.05476	1.48644	7.4	606.95477	904.05134	1.48950
1.65	2.22154	3.33704	1.39442	3.5	1.37444	19.88770	1.48699	7.5	669.82494	997.70714	1.48950
1.66	2.22941	3.36845	1.39617	3.6	1.62707	21.50968	1.48743	7.6	739.22222	1101.01568	1.48950
1.67	2.23737	3.40000	1.39787	3.7	1.86890	24.04419	1.48788	7.7	815.82701	1215.18071	1.48950
1.68	2.24542	3.43166	1.40039	3.8	2.09849	26.50442	1.48837	7.8	900.38700	1341.13066	1.48950
1.69	2.25356	3.46345	1.40287	3.9	2.31467	29.00427	1.48889	7.9	993.73002	1480.16606	1.48950
1.70	2.26179	3.49536	1.40532	4.0	2.51778	32.29303	1.48958	8.0	1096.77089	1633.44526	1.48950
1.71	2.26999	3.52740	1.40774	4.1	2.70779	35.58913	1.48974	8.1	1210.51490	1803.07095	1.48950
1.72	2.27826	3.55956	1.41012	4.2	2.88495	39.22452	1.48988	8.2	1336.08310	1990.19239	1.48950
1.73	2.28659	3.59185	1.41248	4.3	3.04995	43.34429	1.48999	8.3	1474.69908	2196.51164	1.48950
1.74	2.29497	3.62426	1.41481	4.4	3.20389	47.95715	1.48999	8.4	1627.72303	2444.50162	1.48950
1.75	2.30339	3.65680	1.41712	4.5	3.34778	52.95584	1.48916	8.5	1796.65455	2676.12602	1.48950
1.76	2.31185	3.68945	1.41941	4.6	3.48179	58.35457	1.48922	8.6	1983.14952	2933.41126	1.48950
1.77	2.32035	3.72220	1.42167	4.7	3.60609	64.15432	1.48935	8.7	2189.03322	3200.90733	1.48950
1.78	2.32889	3.75505	1.42391	4.8	3.72179	70.35432	1.48945	8.8	2409.33389	3592.41466	1.48950
1.79	2.33747	3.78799	1.42612	4.9	3.82978	77.05432	1.48945	8.9	2647.27158	3992.41466	1.48950
1.80	2.34609	3.82103	1.42831	5.0	3.92978	85.59991	1.48938	9.0	2904.1108	4395.56643	1.48950
1.81	2.35475	3.85417	1.43048	5.1	4.02179	94.99703	1.48940	9.1	3181.17066	4811.45580	1.48950
1.82	2.36345	3.88740	1.43263	5.2	4.10609	104.99703	1.48942	9.2	3478.85955	5244.73758	1.48950
1.83	2.37219	3.92073	1.43477	5.3	4.18279	114.90242	1.48944	9.3	3800.66505	5699.53758	1.48950
1.84	2.38097	3.95417	1.43689	5.4	4.25179	126.61432	1.48945	9.4	4147.28441	6182.53463	1.48950
1.85	2.38979	3.98770	1.43896	5.5	4.31279	139.47442	1.48946	9.5	4518.68404	6692.53463	1.48950
1.86	2.39864	4.02133	1.44101	5.6	4.36679	153.07773	1.48949	9.6	4826.73338	7189.44720	1.48950
1.87	2.40751	4.05506	1.44304	5.7	4.41379	167.07773	1.48949	9.7	5168.73338	7720.44720	1.48950
1.88	2.41641	4.08889	1.44507	5.8	4.45379	181.07773	1.48949	9.8	5538.73338	8281.44720	1.48950
1.89	2.42533	4.12272	1.44709	5.9	4.48679	195.07773	1.48948	9.9	5938.73338	8861.44720	1.48950
1.90	2.43427	4.15665	1.44909	6.0	4.51379	209.07773	1.48948	10.0	6368.73338	9461.44720	1.48950
1.91	2.44323	4.19067	1.45107								
1.92	2.45221	4.22479	1.45304								
1.93	2.46121	4.25899	1.45500								
1.94	2.47023	4.29326	1.45695								
1.95	2.47927	4.32760	1.45889								
1.96	2.48833	4.36200	1.46082								
1.97	2.49741	4.39645	1.46274								
1.98	2.50651	4.43094	1.46465								
1.99	2.51563	4.46547	1.46655								
2.00	2.52477	4.50000	1.46845								

TABLE 8B. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 3/5$ and x from 1.50 to 10.0.

$$\alpha = 4/5$$

x	$F_{4/5}(x)$	$H_{1/5}(x)$	$T_{4/5}(x)$	x	$F_{4/5}(x)$	$H_{1/5}(x)$	$T_{4/5}(x)$	x	$F_{4/5}(x)$	$H_{1/5}(x)$	$T_{4/5}(x)$
0.0	1.00000	0.0	0.0	0.50	1.07949	3.02345	2.80081	1.00	1.3486	4.62477	3.44761
0.01	1.00003	0.50257	0.00059	0.51	1.08246	3.08377	2.82325	1.01	1.35206	4.62477	3.44761
0.02	1.00015	0.92515	0.00241	0.52	1.08610	3.14803	2.84595	1.02	1.35549	4.62477	3.44761
0.03	1.00028	0.92215	0.00347	0.53	1.08951	3.21424	2.86799	1.03	1.35893	4.62477	3.44761
0.04	1.00050	1.04599	1.04599	0.54	1.09259	3.28144	2.88999	1.04	1.36237	4.62477	3.44761
0.05	1.00078	1.13088	1.13088	0.55	1.09544	3.34864	2.91199	1.05	1.36581	4.62477	3.44761
0.06	1.00115	1.15093	1.15093	0.56	1.09805	3.41584	2.93399	1.06	1.36925	4.62477	3.44761
0.07	1.00153	1.15381	1.15381	0.57	1.10038	3.48304	2.95599	1.07	1.37269	4.62477	3.44761
0.08	1.00200	1.14487	1.14487	0.58	1.10241	3.55024	2.97799	1.08	1.37613	4.62477	3.44761
0.09	1.00253	1.11699	1.11699	0.59	1.10414	3.61744	2.99999	1.09	1.37957	4.62477	3.44761
0.10	1.00313	1.07112	1.07112	0.60	1.10557	3.68464	3.02199	1.10	1.38301	4.62477	3.44761
0.11	1.00379	1.00734	1.00734	0.61	1.10670	3.75184	3.04399	1.11	1.38645	4.62477	3.44761
0.12	1.00450	1.02137	1.02137	0.62	1.10763	3.81904	3.06599	1.12	1.38989	4.62477	3.44761
0.13	1.00529	1.01537	1.01537	0.63	1.10836	3.88624	3.08799	1.13	1.39333	4.62477	3.44761
0.14	1.00613	1.00813	1.00813	0.64	1.10889	3.95344	3.10999	1.14	1.39677	4.62477	3.44761
0.15	1.00704	0.99704	0.99704	0.65	1.10922	4.02064	3.13199	1.15	1.40021	4.62477	3.44761
0.16	1.00801	0.98203	0.98203	0.66	1.10935	4.08784	3.15399	1.16	1.40365	4.62477	3.44761
0.17	1.00905	0.96481	0.96481	0.67	1.10928	4.15504	3.17599	1.17	1.40709	4.62477	3.44761
0.18	1.01015	0.94481	0.94481	0.68	1.10891	4.22224	3.19799	1.18	1.41053	4.62477	3.44761
0.19	1.01131	0.92169	0.92169	0.69	1.10824	4.28944	3.21999	1.19	1.41397	4.62477	3.44761
0.20	1.01253	0.89580	0.89580	0.70	1.10727	4.35664	3.24199	1.20	1.41741	4.62477	3.44761
0.21	1.01382	0.86780	0.86780	0.71	1.10590	4.42384	3.26399	1.21	1.42085	4.62477	3.44761
0.22	1.01518	0.83825	0.83825	0.72	1.10413	4.49104	3.28599	1.22	1.42429	4.62477	3.44761
0.23	1.01659	0.80769	0.80769	0.73	1.10196	4.55824	3.30799	1.23	1.42773	4.62477	3.44761
0.24	1.01807	0.77569	0.77569	0.74	1.10039	4.62544	3.32999	1.24	1.43117	4.62477	3.44761
0.25	1.01962	0.74282	0.74282	0.75	1.09842	4.69264	3.35199	1.25	1.43461	4.62477	3.44761
0.26	1.02122	0.70965	0.70965	0.76	1.09606	4.75984	3.37399	1.26	1.43805	4.62477	3.44761
0.27	1.02290	0.67580	0.67580	0.77	1.09330	4.82704	3.39599	1.27	1.44149	4.62477	3.44761
0.28	1.02464	0.64124	0.64124	0.78	1.09013	4.89424	3.41799	1.28	1.44493	4.62477	3.44761
0.29	1.02644	0.60647	0.60647	0.79	1.08656	4.96144	3.43999	1.29	1.44837	4.62477	3.44761
0.30	1.02830	0.57130	0.57130	0.80	1.08259	5.02864	3.46199	1.30	1.45181	4.62477	3.44761
0.31	1.03023	0.53580	0.53580	0.81	1.07822	5.09584	3.48399	1.31	1.45525	4.62477	3.44761
0.32	1.03223	0.49995	0.49995	0.82	1.07345	5.16304	3.50599	1.32	1.45869	4.62477	3.44761
0.33	1.03429	0.46382	0.46382	0.83	1.06828	5.23024	3.52799	1.33	1.46213	4.62477	3.44761
0.34	1.03642	0.42739	0.42739	0.84	1.06271	5.29744	3.54999	1.34	1.46557	4.62477	3.44761
0.35	1.03861	0.39067	0.39067	0.85	1.05674	5.36464	3.57199	1.35	1.46901	4.62477	3.44761
0.36	1.04087	0.35363	0.35363	0.86	1.05037	5.43184	3.59399	1.36	1.47245	4.62477	3.44761
0.37	1.04319	0.31630	0.31630	0.87	1.04360	5.49904	3.61599	1.37	1.47589	4.62477	3.44761
0.38	1.04556	0.27867	0.27867	0.88	1.03643	5.56624	3.63799	1.38	1.47933	4.62477	3.44761
0.39	1.04804	0.24082	0.24082	0.89	1.02886	5.63344	3.65999	1.39	1.48277	4.62477	3.44761
0.40	1.05056	0.20288	0.20288	0.90	1.02096	5.70064	3.68199	1.40	1.48621	4.62477	3.44761
0.41	1.05315	0.16463	0.16463	0.91	1.01269	5.76784	3.70399	1.41	1.48965	4.62477	3.44761
0.42	1.05580	0.12610	0.12610	0.92	1.00403	5.83504	3.72599	1.42	1.49309	4.62477	3.44761
0.43	1.05853	0.08735	0.08735	0.93	1.00499	5.90224	3.74799	1.43	1.49653	4.62477	3.44761
0.44	1.06132	0.04842	0.04842	0.94	1.00552	5.96944	3.76999	1.44	1.50000	4.62477	3.44761
0.45	1.06418	0.00918	0.00918	0.95	1.00565	6.03664	3.79199	1.45	1.50344	4.62477	3.44761
0.46	1.06707	0.03017	0.03017	0.96	1.00538	6.10384	3.81399	1.46	1.50688	4.62477	3.44761
0.47	1.07010	0.09120	0.09120	0.97	1.00471	6.17104	3.83599	1.47	1.51032	4.62477	3.44761
0.48	1.07316	0.25263	0.25263	0.98	1.00364	6.23824	3.85799	1.48	1.51376	4.62477	3.44761
0.49	1.07629	0.51406	0.51406	0.99	1.00217	6.30544	3.87999	1.49	1.51720	4.62477	3.44761
0.50	1.07949	0.77549	0.77549	1.00	1.00040	6.37264	3.90199	1.50	1.52064	4.62477	3.44761

TABLE 9A. Lanchester-Clifford-Schl fli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 4/5$ and x from 0.00 to 1.50.

$\alpha = 4/5$

x	$F_{4/5}(x)$	$H_{1/5}(x)$	$T_{4/5}(x)$
1.50	1.82062	6.89373	3.75077
1.51	1.83320	6.88281	3.75432
1.52	1.84642	6.87191	3.75787
1.53	1.85974	6.86124	3.76161
1.54	1.87324	6.85083	3.76553
1.55	1.88686	6.84063	3.76961
1.56	1.89982	6.83063	3.77383
1.57	1.91221	6.82083	3.77819
1.58	1.92524	6.81124	3.78269
1.59	1.93904	6.80183	3.78733
1.60	1.95333	6.79259	3.79211
1.61	1.96786	6.78354	3.79701
1.62	1.98192	6.77463	3.80201
1.63	1.99602	6.76583	3.80711
1.64	2.01022	6.75719	3.81231
1.65	2.02517	6.74863	3.81761
1.66	2.04006	6.74019	3.82301
1.67	2.05500	6.73183	3.82851
1.68	2.07009	6.72354	3.83411
1.69	2.08533	6.71534	3.83981
1.70	2.10073	6.70719	3.84561
1.71	2.11628	6.69909	3.85151
1.72	2.13200	6.69109	3.85751
1.73	2.14786	6.68319	3.86361
1.74	2.16386	6.67539	3.86981
1.75	2.18000	6.66769	3.87611
1.76	2.19628	6.66009	3.88251
1.77	2.21269	6.65259	3.88901
1.78	2.22924	6.64519	3.89561
1.79	2.24600	6.63789	3.90231
1.80	2.26339	6.63069	3.90911
1.81	2.28100	6.62359	3.91601
1.82	2.29892	6.61659	3.92301
1.83	2.31706	6.60969	3.93011
1.84	2.33542	6.60289	3.93731
1.85	2.35400	6.59619	3.94461
1.86	2.37280	6.58959	3.95201
1.87	2.39182	6.58309	3.95951
1.88	2.41106	6.57669	3.96711
1.89	2.43052	6.57039	3.97481
1.90	2.45020	6.56419	3.98261
1.91	2.47010	6.55809	3.99051
1.92	2.49022	6.55209	3.99851
1.93	2.51056	6.54619	4.00661
1.94	2.53112	6.54039	4.01481
1.95	2.55190	6.53469	4.02311
1.96	2.57290	6.52909	4.03151
1.97	2.59412	6.52359	4.04001
1.98	2.61556	6.51819	4.04861
1.99	2.63722	6.51289	4.05731
2.00	2.66139	6.50769	4.06611

TABLE 9B. Lanchester-Clifford-Schlöfli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 4/5$ and x from 1.50 to 10.0.

x	$F_{4/5}(x)$	$H_{1/5}(x)$	$T_{4/5}(x)$
6.0	97.15298	33.09540	3.86856
6.2	107.49121	42.04242	3.88162
6.4	119.10551	50.98696	3.89473
6.6	141.96489	59.93068	3.90813
6.8	156.11948	61.51587	3.91488
7.0	171.69334	67.05126	3.92483
7.2	188.84889	74.67626	3.93432
7.4	207.72620	81.911820	3.94324
7.6	227.42056	90.166828	3.95196
7.8	251.38912	99.28809	3.96048
8.0	272.11318	109.49849	3.96889
8.2	293.90321	119.97911	3.97711
8.4	317.05848	130.73933	3.98524
8.6	341.95823	141.78408	3.99324
8.8	368.47431	153.12361	3.99996
9.0	405.50035	158.98691	3.99996
9.2	446.27361	175.76597	3.99996
9.4	491.77191	193.62395	3.99996
9.6	540.62505	213.11696	3.99996
9.8	593.08607	234.67937	3.99996
10.0	650.9312	259.39512	3.99996
10.2	713.13118	289.30849	3.99996
10.4	780.90321	310.55508	3.99996
10.6	854.45848	346.67201	3.99996
10.8	934.95823	379.47937	3.99996
11.0	1059.61611	418.832639	3.99996
11.2	1166.76302	460.83300	3.99996
11.4	1284.80185	506.58910	3.99996
11.6	1414.84405	557.97745	3.99996
11.8	1558.11484	614.02939	3.99996
12.0	1715.96487	676.47083	3.99996
12.2	1889.88308	744.73207	3.99996
12.4	2081.51091	820.50869	3.99996
12.6	2292.65195	904.01334	3.99996
12.8	2525.31320	997.95379	3.99996
13.0	2781.69411	1098.90301	3.99996
13.2	3064.20757	1208.22298	3.99996
13.4	3375.5310	1310.55601	3.99996
13.6	3718.61333	1463.52546	3.99996
13.8	4096.71333	1613.34682	3.99996
14.0	4513.40140	1779.45024	3.99996

x	$F_{3/7}(x)$	$H_{4/7}(x)$	$T_{3/7}(x)$	x	$F_{3/7}(x)$	$H_{4/7}(x)$	$T_{3/7}(x)$	x	$F_{3/7}(x)$	$H_{4/7}(x)$	$T_{3/7}(x)$
0.0	1.00000	0.00410	0.0	0.50	1.14905	0.37335	0.32492	1.0	1.3619	0.92366	0.5626
0.1	1.00006	0.00906	0.00410	0.51	1.15211	0.37250	0.32311	1.01	1.36417	0.92379	0.56263
0.2	1.00023	0.01441	0.00906	0.52	1.15493	0.37161	0.32185	1.02	1.36647	0.92392	0.56266
0.3	1.00053	0.02002	0.01441	0.53	1.15748	0.37063	0.32057	1.03	1.36884	0.92405	0.56269
0.4	1.00095	0.02584	0.02002	0.54	1.15978	0.36957	0.31927	1.04	1.37127	0.92418	0.56272
0.5	1.00146	0.03183	0.02584	0.55	1.16188	0.36843	0.31794	1.05	1.37384	0.92431	0.56275
0.6	1.00210	0.03797	0.03183	0.56	1.16374	0.36721	0.31657	1.06	1.37654	0.92444	0.56278
0.7	1.00286	0.04424	0.03797	0.57	1.16536	0.36591	0.31519	1.07	1.37936	0.92457	0.56281
0.8	1.00374	0.05063	0.04424	0.58	1.16674	0.36453	0.31377	1.08	1.38229	0.92470	0.56284
0.9	1.00473	0.05713	0.05063	0.59	1.16788	0.36308	0.31231	1.09	1.38532	0.92483	0.56287
1.0	1.00584	0.06372	0.05713	0.60	1.16878	0.36157	0.31081	1.10	1.38844	0.92496	0.56290
0.1	1.00707	0.07041	0.06372	0.61	1.16944	0.36000	0.30927	1.11	1.39164	0.92509	0.56293
0.2	1.00841	0.07719	0.07041	0.62	1.16986	0.35837	0.30769	1.12	1.39491	0.92522	0.56296
0.3	1.00985	0.08404	0.07719	0.63	1.17004	0.35668	0.30607	1.13	1.39824	0.92535	0.56299
0.4	1.01145	0.09098	0.08404	0.64	1.17000	0.35493	0.30441	1.14	1.40164	0.92548	0.56302
0.5	1.01315	0.09799	0.09098	0.65	1.16974	0.35312	0.30271	1.15	1.40511	0.92561	0.56305
0.6	1.01490	0.10508	0.09799	0.66	1.16926	0.35125	0.30097	1.16	1.40864	0.92574	0.56308
0.7	1.01669	0.11223	0.10508	0.67	1.16856	0.34932	0.29919	1.17	1.41224	0.92587	0.56311
0.8	1.01855	0.11946	0.11223	0.68	1.16763	0.34733	0.29737	1.18	1.41590	0.92600	0.56314
0.9	1.02042	0.12675	0.11946	0.69	1.16647	0.34528	0.29551	1.19	1.41962	0.92613	0.56317
1.0	1.02234	0.13410	0.12675	0.70	1.16508	0.34317	0.29361	1.20	1.42340	0.92626	0.56320
0.1	1.02432	0.14152	0.13410	0.71	1.16346	0.34100	0.29167	1.21	1.42724	0.92639	0.56323
0.2	1.02635	0.14900	0.14152	0.72	1.16152	0.33877	0.28969	1.22	1.43114	0.92652	0.56326
0.3	1.02843	0.15653	0.14900	0.73	1.15926	0.33648	0.28767	1.23	1.43509	0.92665	0.56329
0.4	1.03067	0.16415	0.15653	0.74	1.15667	0.33413	0.28561	1.24	1.43909	0.92678	0.56332
0.5	1.03299	0.17182	0.16415	0.75	1.15374	0.33172	0.28351	1.25	1.44314	0.92691	0.56335
0.6	1.03540	0.17954	0.17182	0.76	1.15048	0.32925	0.28137	1.26	1.44724	0.92704	0.56338
0.7	1.03789	0.18732	0.17954	0.77	1.14689	0.32672	0.27919	1.27	1.45139	0.92717	0.56341
0.8	1.04046	0.19516	0.18732	0.78	1.14296	0.32413	0.27697	1.28	1.45559	0.92730	0.56344
0.9	1.04311	0.20306	0.19516	0.79	1.13869	0.32148	0.27471	1.29	1.45984	0.92743	0.56347
1.0	1.04584	0.21102	0.20306	0.80	1.13408	0.31877	0.27241	1.30	1.46414	0.92756	0.56350
0.1	1.04864	0.21904	0.21102	0.81	1.12914	0.31600	0.27007	1.31	1.46849	0.92769	0.56353
0.2	1.05151	0.22711	0.21904	0.82	1.12386	0.31317	0.26769	1.32	1.47289	0.92782	0.56356
0.3	1.05444	0.23524	0.22711	0.83	1.11824	0.31028	0.26527	1.33	1.47734	0.92795	0.56359
0.4	1.05743	0.24343	0.23524	0.84	1.11228	0.30733	0.26281	1.34	1.48184	0.92808	0.56362
0.5	1.06048	0.25167	0.24343	0.85	1.10600	0.30432	0.26031	1.35	1.48639	0.92821	0.56365
0.6	1.06359	0.25994	0.25167	0.86	1.10040	0.30125	0.25777	1.36	1.49099	0.92834	0.56368
0.7	1.06676	0.26826	0.25994	0.87	1.09447	0.29812	0.25519	1.37	1.49564	0.92847	0.56371
0.8	1.06999	0.27663	0.26826	0.88	1.08821	0.29494	0.25257	1.38	1.50034	0.92860	0.56374
0.9	1.07328	0.28504	0.27663	0.89	1.08162	0.29171	0.24991	1.39	1.50509	0.92873	0.56377
1.0	1.07663	0.29349	0.28504	0.90	1.07569	0.28843	0.24721	1.40	1.50989	0.92886	0.56380
0.1	1.07995	0.30198	0.29349	0.91	1.06942	0.28510	0.24447	1.41	1.51464	0.92899	0.56383
0.2	1.08333	0.31050	0.30198	0.92	1.06280	0.28172	0.24169	1.42	1.51944	0.92912	0.56386
0.3	1.08676	0.31904	0.31050	0.93	1.05584	0.27829	0.23887	1.43	1.52429	0.92925	0.56389
0.4	1.09024	0.32761	0.31904	0.94	1.04854	0.27481	0.23601	1.44	1.52909	0.92938	0.56392
0.5	1.09377	0.33621	0.32761	0.95	1.04090	0.27128	0.23311	1.45	1.53384	0.92951	0.56395
0.6	1.09734	0.34484	0.33621	0.96	1.03292	0.26770	0.23017	1.46	1.53864	0.92964	0.56398
0.7	1.10095	0.35350	0.34484	0.97	1.02460	0.26407	0.22719	1.47	1.54339	0.92977	0.56401
0.8	1.10460	0.36219	0.35350	0.98	1.01594	0.26040	0.22417	1.48	1.54809	0.92990	0.56404
0.9	1.10829	0.37091	0.36219	0.99	1.00694	0.25668	0.22111	1.49	1.55274	0.93003	0.56407
1.0	1.11202	0.37966	0.37091	1.00	1.00000	0.25291	0.21800	1.50	1.55734	0.93016	0.56410

TABLE 10A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and

$T_{\alpha}(x)$ for $\alpha = 3/7$ and x from 0.00 to 1.50.

x	$F_{3/7}(x)$	$H_{4/7}(x)$	$T_{3/7}(x)$	x	$F_{3/7}(x)$	$H_{4/7}(x)$	$T_{3/7}(x)$	x	$F_{3/7}(x)$	$H_{4/7}(x)$	$T_{3/7}(x)$
1.50	2.59169	1.76254	0.68007	2.0	4.27065	3.10153	0.72624	6.0	252.71968	190.50872	0.75384
1.51	2.71709	1.80250	0.68009	2.1	4.78017	3.55728	0.73122	6.1	279.66487	210.80709	0.75384
1.52	2.84223	1.80266	0.68010	2.2	5.23611	3.88017	0.73551	6.2	309.47366	233.29292	0.75384
1.53	2.96688	1.80262	0.68011	2.3	5.65002	4.08421	0.73866	6.3	342.45369	258.15445	0.75384
1.54	3.09092	1.80268	0.68012	2.4	6.02568	4.26410	0.74142	6.4	378.93865	285.05902	0.75384
1.55	3.21415	1.86636	0.68690	2.5	7.11957	5.29460	0.74367	6.5	419.30250	316.08721	0.75384
1.56	3.33828	1.89133	0.68819	2.6	7.89889	5.88124	0.74552	6.6	463.55688	349.74932	0.75384
1.57	3.46352	1.91352	0.68945	2.7	8.74163	6.33024	0.74703	6.7	513.55352	386.98855	0.75384
1.58	3.58949	1.93291	0.69069	2.8	9.68661	7.42412	0.74827	6.8	568.90223	429.18449	0.75384
1.59	3.71659	1.95052	0.69191	2.9	10.73361	8.42817	0.74928	6.9	628.45630	473.75698	0.75384
1.60	3.84486	1.98135	0.69311	3.0	11.89347	9.32139	0.75019	7.0	695.33122	524.17032	0.75384
1.61	3.97439	2.00640	0.69428	3.1	13.17819	9.99402	0.75074	7.1	769.93805	579.93806	0.75384
1.62	4.10517	2.02766	0.69544	3.2	14.60103	10.57038	0.75130	7.2	851.14220	641.62815	0.75384
1.63	4.23739	2.04515	0.69657	3.3	16.17669	12.16159	0.75180	7.3	941.66633	709.86875	0.75384
1.64	4.37146	2.07167	0.69768	3.4	17.92145	13.47996	0.75217	7.4	1041.80018	785.35455	0.75384
1.65	4.50760	2.09881	0.69877	3.5	19.85328	14.32907	0.75247	7.5	1152.56566	868.05388	0.75384
1.66	4.64583	2.12704	0.69983	3.6	21.99312	16.55396	0.75272	7.6	1273.08738	961.21644	0.75384
1.67	4.78616	2.15657	0.70088	3.7	24.36000	18.34129	0.75293	7.7	1403.91328	1065.38196	0.75384
1.68	4.92859	2.18740	0.70191	3.8	26.96831	20.70947	0.75303	7.8	1545.95328	1180.38196	0.75384
1.69	5.07312	2.21962	0.70292	3.9	29.82804	22.50879	0.75323	7.9	1700.35862	1316.38913	0.75384
1.70	5.21975	2.25341	0.70391	4.0	33.09505	24.93186	0.75334	8.0	1909.74766	1439.61124	0.75384
1.71	5.36848	2.28797	0.70488	4.1	36.65935	27.11256	0.75344	8.1	2112.61531	1592.51180	0.75384
1.72	5.51921	2.32409	0.70583	4.2	40.58552	30.58148	0.75351	8.2	2337.00362	1771.73543	0.75384
1.73	5.67194	2.36186	0.70677	4.3	44.94997	33.86609	0.75357	8.3	2585.19233	1948.83137	0.75384
1.74	5.82667	2.40129	0.70769	4.4	49.76144	37.50113	0.75362	8.4	2859.70505	2155.77090	0.75384
1.75	5.98340	2.44241	0.70858	4.5	55.09643	41.52393	0.75369	8.5	3163.32988	2384.65663	0.75384
1.76	6.14213	2.48547	0.70947	4.6	61.00697	45.97576	0.75376	8.6	3497.53148	2637.81382	0.75384
1.77	6.30286	2.53047	0.71033	4.7	67.53473	50.92229	0.75372	8.7	3880.39133	2917.50874	0.75384
1.78	6.46559	2.57747	0.71118	4.8	74.76325	56.36688	0.75376	8.8	4315.75219	3227.50874	0.75384
1.79	6.63032	2.62654	0.71201	4.9	82.76144	62.36688	0.75376	8.9	4805.75219	3570.0821	0.75384
1.80	6.79705	2.67770	0.71283	5.0	91.62221	69.06263	0.75378	9.0	5238.27950	3948.84475	0.75384
1.81	6.96578	2.73090	0.71363	5.1	101.42059	76.44966	0.75379	9.1	5794.07225	4367.82572	0.75384
1.82	7.13651	2.78619	0.71442	5.2	111.26692	84.62362	0.75380	9.2	6408.77225	4831.21380	0.75384
1.83	7.30924	2.84351	0.71519	5.3	122.26019	93.68816	0.75381	9.3	7088.61905	5343.71190	0.75384
1.84	7.48397	2.90283	0.71592	5.4	134.53514	103.67584	0.75381	9.4	7840.51021	5910.52045	0.75384
1.85	7.66070	2.96416	0.71669	5.5	148.22357	114.74901	0.75382	9.5	8672.07486	6537.33037	0.75384
1.86	7.83943	3.02749	0.71742	5.6	163.35370	127.00092	0.75382	9.6	9579.87950	7259.08112	0.75384
1.87	8.02016	3.09282	0.71813	5.7	180.03570	142.22890	0.75383	9.7	10601.67950	8055.56886	0.75384
1.88	8.20289	3.16015	0.71883	5.8	199.36345	159.52934	0.75383	9.8	11733.72807	8955.20173	0.75384
1.89	8.38762	3.22948	0.71952	5.9	221.56345	179.22934	0.75383	9.9	12977.75603	9733.20173	0.75384
1.90	8.57435	3.30081	0.72019	6.0	252.71948	190.50872	0.75383	10.0	14353.55964	10820.34285	0.75384
1.91	8.76308	3.37414	0.72085								
1.92	8.95381	3.44947	0.72150								
1.93	9.14654	3.52680	0.72215								
1.94	9.34127	3.60613	0.72279								
1.95	9.53800	3.68746	0.72342								
1.96	9.73673	3.77079	0.72405								
1.97	9.93756	3.85612	0.72467								
1.98	10.14039	3.94345	0.72528								
1.99	10.34522	4.03278	0.72589								
2.00	10.55205	4.12411	0.72649								

TABLE 10B. Lanchester-Clifford-Schlafli Functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and

$T_\alpha(x)$ for $\alpha = 3/7$ and x from 1.50 to 10.0.

x	$F_{4/7}(x)$	$F_{3/7}(x)$	$T_{4/7}(x)$	x	$F_{4/7}(x)$	$H_{3/7}(x)$	$T_{4/7}(x)$	x	$F_{4/7}(x)$	$H_{3/7}(x)$	$T_{4/7}(x)$
0.0	1.00000	0.0	0.0	0.50	1.1157	0.74260	0.66807	1.00	1.47345	1.52541	1.05227
0.01	1.00004	0.02487	0.02487	0.51	1.1177	0.75693	0.67788	1.01	1.48373	1.53557	1.06231
0.02	1.00019	0.04905	0.04904	0.52	1.12087	0.77069	0.68759	1.02	1.49414	1.54580	1.07248
0.03	1.00037	0.07376	0.07376	0.53	1.12566	0.78479	0.69719	1.03	1.50468	1.55618	1.08279
0.04	1.00070	0.08163	0.08157	0.54	1.13056	0.79893	0.70666	1.04	1.51537	1.56670	1.09323
0.05	1.00109	0.09885	0.09874	0.55	1.13566	0.81311	0.71604	1.05	1.52619	1.57736	1.10374
0.06	1.00158	0.11595	0.11541	0.56	1.14096	0.82734	0.72532	1.06	1.53714	1.58808	1.11441
0.07	1.00214	0.13195	0.13167	0.57	1.14646	0.84161	0.73459	1.07	1.54824	1.59886	1.12517
0.08	1.00280	0.14795	0.14758	0.58	1.15216	0.85593	0.74387	1.08	1.55949	1.60969	1.13604
0.09	1.00353	0.16396	0.16319	0.59	1.15806	0.87029	0.75315	1.09	1.57089	1.62056	1.14691
0.10	1.00430	0.17929	0.17851	0.60	1.16406	0.88470	0.76243	1.10	1.58249	1.63146	1.15779
0.11	1.00530	0.19463	0.19360	0.61	1.17016	0.89914	0.77171	1.11	1.59424	1.64239	1.16866
0.12	1.00631	0.20978	0.20847	0.62	1.17636	0.91361	0.78100	1.12	1.60614	1.65334	1.17954
0.13	1.00740	0.22478	0.22313	0.63	1.18266	0.92811	0.79029	1.13	1.61819	1.66431	1.19041
0.14	1.00859	0.23963	0.23759	0.64	1.18916	0.94263	0.80000	1.14	1.63039	1.67530	1.20128
0.15	1.00986	0.25436	0.25188	0.65	1.19576	0.95717	0.80971	1.15	1.64274	1.68631	1.21215
0.16	1.01122	0.26899	0.26599	0.66	1.20246	0.97173	0.81936	1.16	1.65524	1.69734	1.22302
0.17	1.01267	0.28348	0.27944	0.67	1.20926	0.98631	0.82896	1.17	1.66789	1.70839	1.23389
0.18	1.01421	0.29789	0.29357	0.68	1.21616	0.99990	0.83851	1.18	1.68069	1.71946	1.24476
0.19	1.01584	0.31223	0.30757	0.69	1.22316	1.01350	0.84800	1.19	1.69364	1.73054	1.25563
0.20	1.01756	0.32649	0.32096	0.70	1.23026	1.02710	0.85744	1.20	1.70674	1.74163	1.26650
0.21	1.01936	0.34068	0.33472	0.71	1.23746	1.04069	0.86683	1.21	1.71994	1.75272	1.27737
0.22	1.02126	0.35480	0.34842	0.72	1.24476	1.05429	0.87617	1.22	1.73324	1.76381	1.28824
0.23	1.02326	0.36887	0.36049	0.73	1.25216	1.06788	0.88546	1.23	1.74664	1.77490	1.29911
0.24	1.02532	0.38289	0.37343	0.74	1.25966	1.08147	0.89470	1.24	1.76014	1.78600	1.31000
0.25	1.02749	0.39686	0.38625	0.75	1.26726	1.09506	0.90389	1.25	1.77374	1.79710	1.32089
0.26	1.02973	0.41079	0.39993	0.76	1.27496	1.10865	0.91303	1.26	1.78744	1.80820	1.33178
0.27	1.03208	0.42469	0.41449	0.77	1.28276	1.12224	0.92212	1.27	1.80124	1.81930	1.34267
0.28	1.03451	0.43856	0.42824	0.78	1.29066	1.13583	0.93117	1.28	1.81514	1.83040	1.35356
0.29	1.03704	0.45240	0.44204	0.79	1.29866	1.14942	0.94022	1.29	1.82914	1.84150	1.36445
0.30	1.03969	0.46622	0.45583	0.80	1.30676	1.16301	0.94927	1.30	1.84324	1.85260	1.37534
0.31	1.04237	0.48000	0.46960	0.81	1.31496	1.17660	0.95832	1.31	1.85744	1.86370	1.38623
0.32	1.04517	0.49380	0.48346	0.82	1.32326	1.19019	0.96737	1.32	1.87174	1.87480	1.39712
0.33	1.04806	0.50757	0.49729	0.83	1.33166	1.20378	0.97642	1.33	1.88614	1.88590	1.40801
0.34	1.05104	0.52133	0.49601	0.84	1.34016	1.21737	0.98547	1.34	1.90064	1.89700	1.41890
0.35	1.05412	0.53508	0.50761	0.85	1.34876	1.23096	0.99452	1.35	1.91524	1.90810	1.42979
0.36	1.05729	0.54884	0.51910	0.86	1.35746	1.24455	1.00357	1.36	1.93004	1.91920	1.44068
0.37	1.06055	0.56259	0.53047	0.87	1.36626	1.25814	1.01262	1.37	1.94494	1.93030	1.45157
0.38	1.06390	0.57635	0.54187	0.88	1.37516	1.27173	1.02167	1.38	1.96004	1.94140	1.46246
0.39	1.06733	0.58911	0.55327	0.89	1.38416	1.28532	1.03072	1.39	1.97524	1.95250	1.47335
0.40	1.07090	0.60388	0.56390	0.90	1.39326	1.29891	1.03977	1.40	1.99054	1.96360	1.48424
0.41	1.07453	0.61866	0.57482	0.91	1.40246	1.31250	1.04882	1.41	2.00594	1.97470	1.49513
0.42	1.07826	0.63346	0.58562	0.92	1.41176	1.32609	1.05787	1.42	2.02144	1.98580	1.50602
0.43	1.08209	0.64827	0.59632	0.93	1.42116	1.33968	1.06692	1.43	2.03704	1.99690	1.51691
0.44	1.08601	0.66310	0.60690	0.94	1.43066	1.35327	1.07597	1.44	2.05274	2.00800	1.52780
0.45	1.09003	0.67795	0.61737	0.95	1.44026	1.36686	1.08502	1.45	2.06854	2.01910	1.53869
0.46	1.09414	0.69283	0.62773	0.96	1.45006	1.38045	1.09407	1.46	2.08444	2.03020	1.54958
0.47	1.09835	0.70773	0.63799	0.97	1.46006	1.39404	1.10312	1.47	2.10044	2.04130	1.56047
0.48	1.10269	0.72266	0.64812	0.98	1.47026	1.40763	1.11217	1.48	2.11654	2.05240	1.57136
0.49	1.10707	0.73761	0.65815	0.99	1.48056	1.42122	1.12122	1.49	2.13274	2.06350	1.58225
0.50	1.11157	0.75260	0.66807	1.00	1.49106	1.43481	1.13027	1.50	2.14904	2.07460	1.59314

TABLE 11A. Lanchester-Clifford-Schlafli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha = 4/7$ and x from 0.00 to 1.50.

x	$F_{4/7}(x)$	$H_{3/7}(x)$	$T_{4/7}(x)$	x	$F_{4/7}(x)$	$H_{3/7}(x)$	$T_{4/7}(x)$	x	$F_{4/7}(x)$	$H_{3/7}(x)$	$T_{4/7}(x)$
1.50	2.17392	2.65223	1.20944	2.0	3.38428	4.33755	1.28168	6.0	164.99174	218.86467	1.32652
1.51	2.19233	2.65977	1.21167	2.1	3.71076	4.38578	1.28962	6.1	182.10854	241.57070	1.32652
1.52	2.21105	2.66722	1.21377	2.2	4.07160	4.43401	1.29692	6.2	201.00541	266.63852	1.32652
1.53	2.22992	2.67477	1.21588	2.3	4.47130	4.48221	1.30421	6.3	221.86839	294.31420	1.32652
1.54	2.24900	2.68233	1.21795	2.4	4.91330	4.53044	1.31159	6.4	244.90215	324.86946	1.32653
1.55	2.26828	2.69000	1.21999	2.5	5.40141	4.57867	1.31902	6.5	270.33290	358.60439	1.32653
1.56	2.28777	2.69777	1.22198	2.6	5.94265	4.62691	1.32653	6.6	300.41044	395.89037	1.32653
1.57	2.30749	2.70562	1.22395	2.7	6.54000	4.67514	1.33402	6.7	335.71735	436.71735	1.32653
1.58	2.32739	2.71353	1.22587	2.8	7.20000	4.72338	1.34159	6.8	376.41988	480.71735	1.32653
1.59	2.34752	2.72150	1.22777	2.9	7.92000	4.77161	1.34927	6.9	423.12299	528.31907	1.32653
1.60	2.36787	2.72952	1.22962	3.0	8.70000	4.81985	1.35695	7.0	476.41988	580.00000	1.32653
1.61	2.38843	2.73761	1.23144	3.1	9.54000	4.86809	1.36463	7.1	536.91735	635.71735	1.32653
1.62	2.40922	2.74577	1.23324	3.2	10.44000	4.91633	1.37231	7.2	605.41988	695.41988	1.32653
1.63	2.43027	2.75399	1.23500	3.3	11.40000	4.96457	1.38000	7.3	682.41988	760.00000	1.32653
1.64	2.45156	2.76227	1.23673	3.4	12.42000	5.01281	1.38769	7.4	768.41988	829.41988	1.32653
1.65	2.47299	2.77062	1.23842	3.5	13.50000	5.06105	1.39538	7.5	864.41988	903.41988	1.32653
1.66	2.49466	2.77907	1.24009	3.6	14.64000	5.10929	1.40307	7.6	971.41988	982.41988	1.32653
1.67	2.51657	2.78762	1.24173	3.7	15.84000	5.15753	1.41076	7.7	1089.41988	1067.41988	1.32653
1.68	2.53872	2.79627	1.24333	3.8	17.10000	5.20577	1.41845	7.8	1219.41988	1158.41988	1.32653
1.69	2.56111	2.80502	1.24491	3.9	18.42000	5.25401	1.42614	7.9	1351.41988	1255.41988	1.32653
1.70	2.58369	2.81387	1.24647	4.0	20.80000	5.30225	1.43383	8.0	1495.41988	1358.41988	1.32653
1.71	2.60649	2.82282	1.24797	4.1	23.24000	5.35049	1.44152	8.1	1651.41988	1467.41988	1.32653
1.72	2.62965	2.83187	1.24949	4.2	25.74000	5.39873	1.44921	8.2	1819.41988	1582.41988	1.32653
1.73	2.65320	2.84102	1.25093	4.3	28.30000	5.44697	1.45690	8.3	1999.41988	1704.41988	1.32653
1.74	2.67719	2.85027	1.25237	4.4	30.92000	5.49521	1.46459	8.4	2191.41988	1833.41988	1.32653
1.75	2.70166	2.85962	1.25379	4.5	33.60000	5.54345	1.47228	8.5	2395.41988	1969.41988	1.32653
1.76	2.72663	2.86907	1.25519	4.6	36.34000	5.59169	1.48000	8.6	2611.41988	2113.41988	1.32653
1.77	2.75212	2.87862	1.25657	4.7	39.14000	5.63993	1.48771	8.7	2839.41988	2265.41988	1.32653
1.78	2.77815	2.88827	1.25793	4.8	42.00000	5.68817	1.49542	8.8	3079.41988	2425.41988	1.32653
1.79	2.80474	2.89802	1.25927	4.9	44.92000	5.73641	1.50313	8.9	3331.41988	2593.41988	1.32653
1.80	2.83189	2.90787	1.26059	5.0	47.90000	5.78465	1.51084	9.0	3595.41988	2769.41988	1.32653
1.81	2.85962	2.91782	1.26193	5.1	50.94000	5.83289	1.51855	9.1	3871.41988	2953.41988	1.32653
1.82	2.88794	2.92787	1.26316	5.2	54.04000	5.88113	1.52626	9.2	4159.41988	3145.41988	1.32653
1.83	2.91687	2.93802	1.26436	5.3	57.20000	5.92937	1.53397	9.3	4459.41988	3345.41988	1.32653
1.84	2.94642	2.94827	1.26556	5.4	60.42000	5.97761	1.54168	9.4	4771.41988	3553.41988	1.32653
1.85	2.97660	2.95862	1.26673	5.5	63.70000	6.02585	1.54939	9.5	5095.41988	3769.41988	1.32653
1.86	2.99742	2.96907	1.26789	5.6	67.04000	6.07409	1.55710	9.6	5431.41988	3993.41988	1.32653
1.87	3.01887	2.97962	1.26903	5.7	70.44000	6.12233	1.56481	9.7	5779.41988	4225.41988	1.32653
1.88	3.04097	2.99027	1.27017	5.8	73.90000	6.17057	1.57252	9.8	6139.41988	4465.41988	1.32653
1.89	3.06372	3.00102	1.27131	5.9	77.42000	6.21881	1.58023	9.9	6511.41988	4713.41988	1.32653
1.90	3.08712	3.01187	1.27245	6.0	81.00000	6.26705	1.58794	10.0	6895.41988	4969.41988	1.32653
1.91	3.11117	3.02282	1.27359								
1.92	3.13587	3.03387	1.27473								
1.93	3.16122	3.04502	1.27587								
1.94	3.18722	3.05627	1.27701								
1.95	3.21387	3.06762	1.27815								
1.96	3.24117	3.07907	1.27929								
1.97	3.26912	3.09062	1.28043								
1.98	3.29772	3.10227	1.28157								
1.99	3.32697	3.11402	1.28271								
2.00	3.35687	3.12587	1.28385								

TABLE 11B. Lanchester-Clifford-Schl fli Functions $F_\alpha(x)$, $H_{1-\alpha}(x)$, and

$T_\alpha(x)$ for $\alpha = 4/7$ and x from 1.50 to 10.0.

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